



Lose the Wrappers – A New Approach for Estimating Project Support Functions

***A Battle Between Common Sense and
Math/Statistics***

Preliminary Results

8/12/14

***Shawn Hayes, Mark Jacobs and Matt Pitlyk
Victory Solutions MIPSS Team***



Objective & Approach



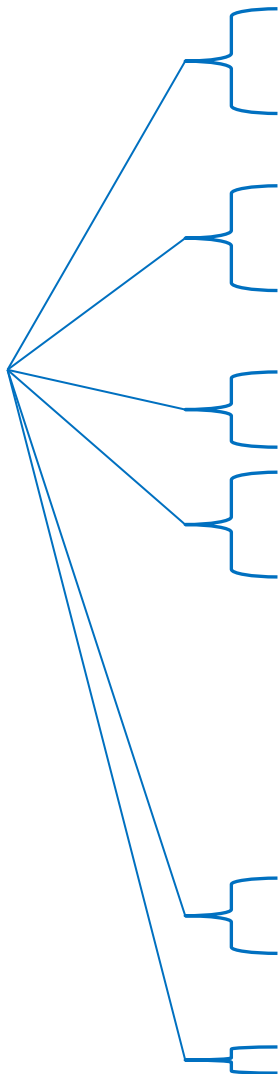
- **Objective: *Develop an improved estimating methodology to capture Management, Systems Engineering, Mission Assurance, and Integration & Test costs***
 - Explore alternatives to the “wrap factor” approach
 - **Cover robotic science spacecraft projects (unmanned)**
- Effort began with proof-of-concept rapid prototype development using an approach similar to what is used for the NASA Space Operations Cost Model (SOCM)
- 2nd Modeling effort explored three alternatives:
 - Standard regression approach
 - Constructive, SOCM-like approach (relies on expert judgment)
 - Statistical approach using Principal Component Analysis (PCA)



NASA WBS Elements Included



NASA WBS items included for this modeling effort



NASA WBS (augmented)	
WBS #	Item
1.0	Program Management
2.0	Systems Engineering
3.0	Mission Assurance
4.0	Science
5.0	Payload
5.1	P/L PM
5.2	P/L SE
5.3	P/L MA
5.4	Instrument #1
5.xx	Instrument n
5.70	P/L Software
5.80	P/L IAT
5.90	P/L GSE
6.0	Spacecraft
6.1	S/C PM
6.2	S/C SE
6.3	S/C MA
6.4	Structure & Mech. (SC 1)
6.5	Thermal (SC 1)
6.6	Power (SC 1)
6.7	C&DH (SC 1)
6.8	Communications (SC 1)
6.9	ACS (SC 1)
6.10	Propulsion (SC 1)
6.11	Harness (SC 1)
6.12-6.43	Repeat 6.4-6.11 for each Flt Element
6.70	S/C Software
6.80	S/C IAT
6.90	S/C GSE
7.0	MOS
8.0	Launch Services
9.0	GDS
10.0	System Level IAT
11.0	E/PO



Study Timeline/Schedule



Feb/Mar 2014

Rapid
Prototype
“Proof-of-
Concept”

Developed with
initial normalized
data set of 20
projects

Utilized findings
from past
studies, which
included
interviews with
PM/SE/MA/I&T
experts from
more than a
dozen projects

Apr/May 2014

Reviewed approach
with cost modeling
experts

Identified candidate cost
drivers (model inputs)

Continued data
collection &
normalization to
expand data set

Expanded to > 40
projects

May-Jul 2014

Explored multiple
modeling
approaches and
compared
findings/results

Includes Standard Regression,
Constructive/SOCM-based,
and Advanced Statistical
options

Continually refined
each approach
incorporating
lessons-learned
from each attempt

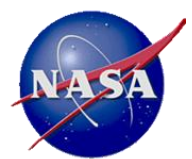
Lessons-learned include
observations of input sensitivity
and project differences

Constructive
Model Option
(SOCM-based)

Uses data-derived
input weightings
and lower-level
cost ranges

Statistical
Model Option
(PCA-based)

Uses advanced
statistical analysis
to develop CERs



Rapid Prototype Inputs



- Individual input weightings are assigned for each WBS element (PM/SE/MA/I&T) in each phase (Design/Fab/I&T/Launch Ops)

		1	2	3	4	5	
Inputs Used for Rapid Prototype	Program- matics	1a NASA PROGRAM	EV, Other	Explorer	Discovery	Flagship	
		1b MISSION RISK CLASS	Class D	Class C	Class B	Class A	
		1c MISSION TARGET/TYPE	Earth Orbiting or Lunar	Mercury, Venus, Mars	Small Bodies	Outer Planets	Planetary Lander or Sample Return
	Lead Type	2a LEAD ORGANIZATION TYPE	Univ	Govt	APL/JPL/SwRI	Industry	Mix/Int'l
		2b FLIGHT SYSTEM ORGANIZATION TYPE	Univ	Govt	APL/JPL/SwRI	Industry	Mix/Int'l
		2c PAYLOAD ORGANIZATION TYPE	Univ	Govt	APL/JPL/SwRI	Industry	Mix/Int'l
	Lead Exp	3a LEAD ORG EXPERIENCE	Extensive		Nominal		Minimal
		3b FLIGHT SYSTEM LEAD ORG EXPERIENCE	Extensive		Nominal		Minimal
		3c PAYLOAD LEAD ORG EXPERIENCE	Extensive		Nominal		Minimal
		4 INHOUSE SCOPE	All Flight Elements Inhouse	Most Flight Elements Inhouse	System contractor for Flight HW	Multiple (2+) major system contractors	Multiple major developers including Gov+Ind
		5 INTERNATIONAL PARTICIPATION	No Int'l HW or Col's	No Int'l HW; Some Int'l Cols	Minimal Int'l HW; Some Int'l Cols	\$2-10M of Int'l HW + Int'l Col's	Sign Int'l HW + Int'l Col's
	FltSys Size/Cplx	6a FLIGHT SYSTEM MASS	<200kg	200-400kg	400-600kg	600-1000kg	>1000kg
		6b FLIGHT SYSTEM POWER	<250W	250-500W	500-700W	700-1000W	>1000W
		6c FLIGHT SYSTEM HERITAGE&TRL	No TRL<7	1-2 TRL<7 items	3-4 TRL<7 items	Several TRL<7 items	Significant ATD for key elements
	Pyld Size/Cplx	7a PAYLOAD MASS	<50kg	50-100kg	100-150kg	150-300kg	>300kg
		7b PAYLOAD POWER	<50W	50-100W	100-300W	300-500W	>500W
		7c PAYLOAD HERITAGE&TRL	No TRL<7	1-2 TRL<7 items	3-4 TRL<7 items	Several TRL<7 items	Significant ATD for key elements

Add'l Input
Candidates

Management: # of Major External I/Fs, S/C Contractor Cost, S/C In-House Cost, Degree of Off-site Oversight, Scope of Identified Risks

Sys Engrng: # of Reqts, Simulation/Test Scope, Contingencies/Margins, Redundancy, Prototyping, Funded Schedule Margin, Unfunded Schedule Slack

Mission Assurance: Parts Quality, Redundancy, Sparing

I&T: # of Flight Elements, Prototyping, Facility Reqts, GSE, Spares, I&T Schedule Margin/Slack

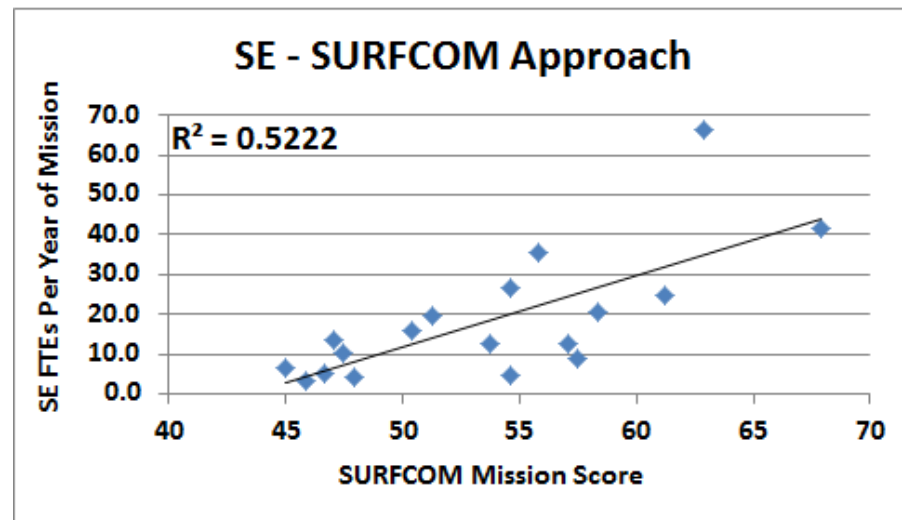
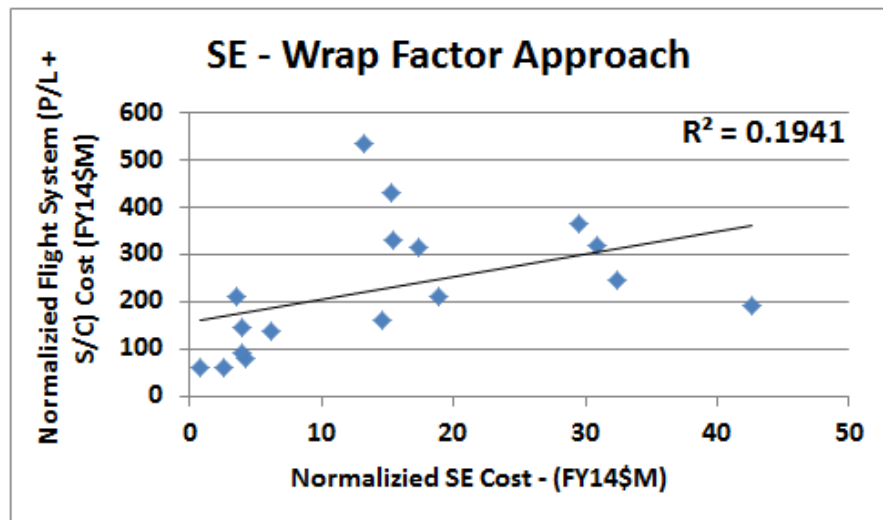
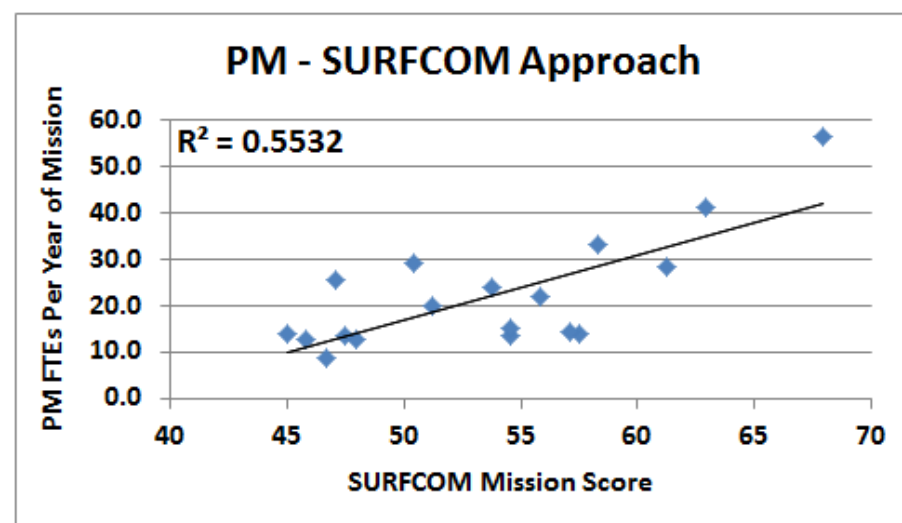
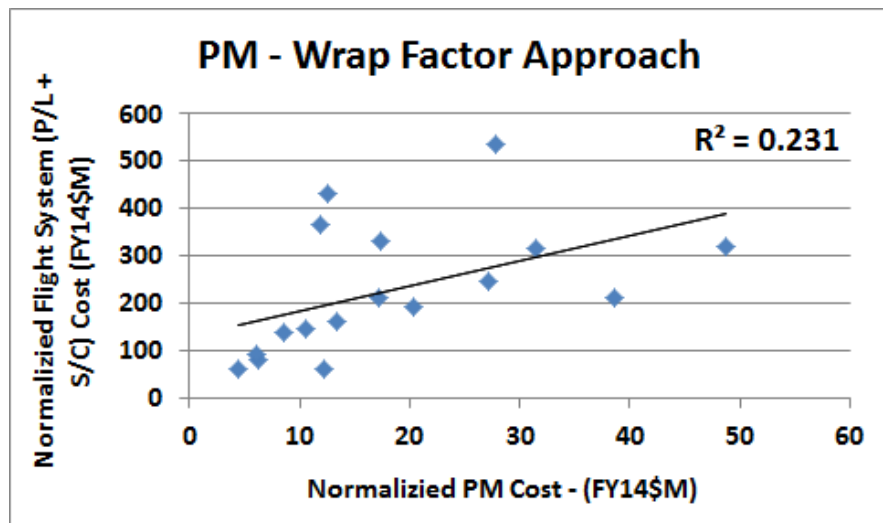




Rapid Prototype Comparison to Wrap Factors, 1 of 2



Engineering
Cost
Office



SURFCOM = Support Function Cost Model



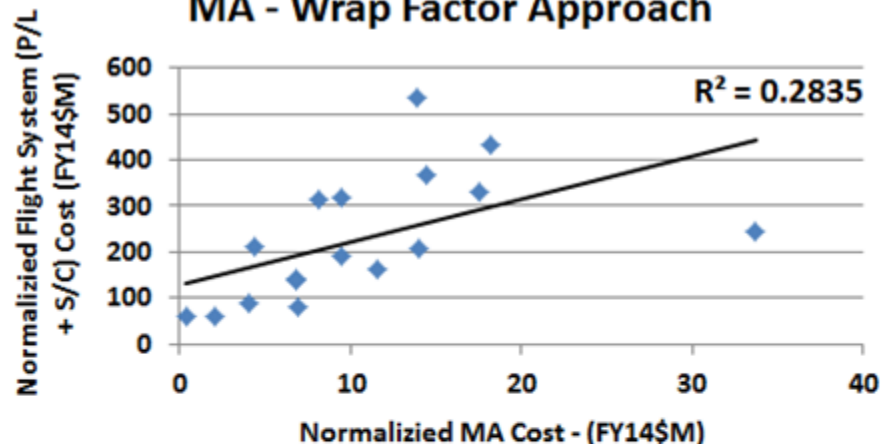


Rapid Prototype Comparison to Wrap Factors, 2 of 2

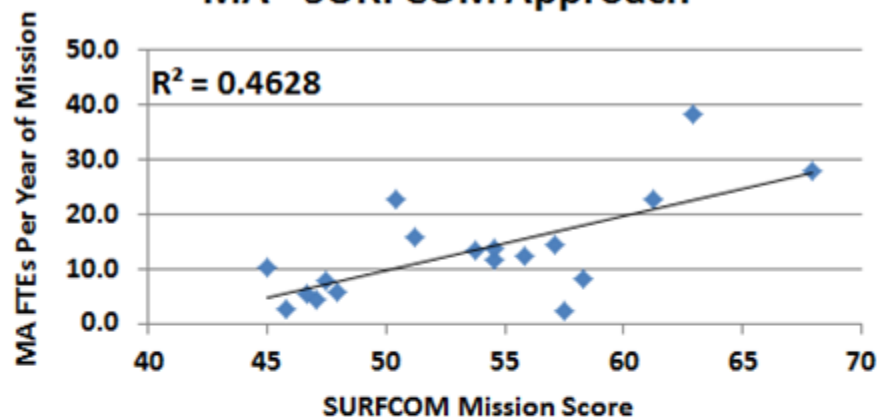


Engineering
Cost
Office

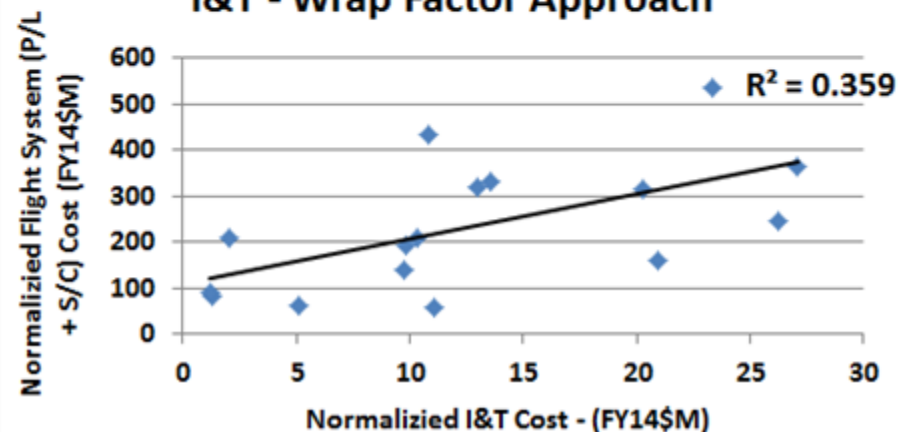
MA - Wrap Factor Approach



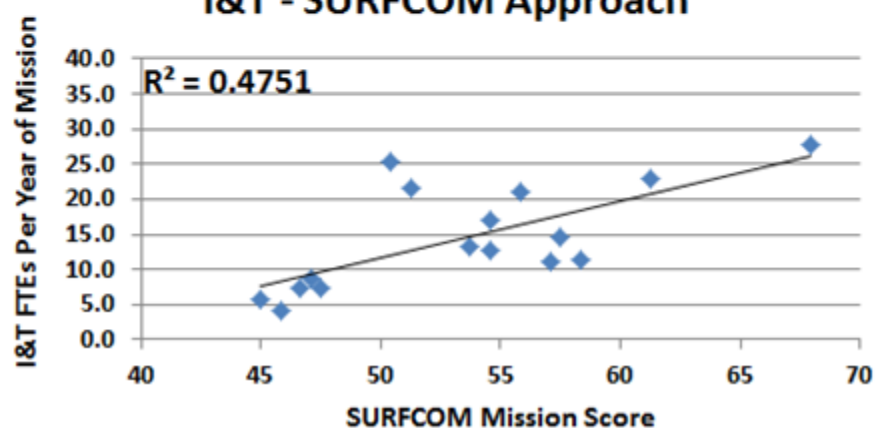
MA - SURFCOM Approach



I&T - Wrap Factor Approach



I&T - SURFCOM Approach



SURFCOM = Support Function Cost Model



Although the prototype approach showed improvement over using wrap factors, it lacked the traceability to the data that comes with a more math/statistics-based approach.



Although separate paths were explored, there was significant “cheating”, where progress down one path was gained from a successful/failed attempt down a different path

“Common-Sense” Path

Explored optimization of a constructive SOCM-type approach

Math/Statistics Path

Explored standard regression and other advanced statistical approaches



Expanded Model Data Set



- Before going further with any modeling attempts, additional data was analyzed and normalized to provide an expanded data set
- The data set used for the initial prototype effort was more than doubled
- The process used for normalizing the data was applied to the latest launch CADRe (described in a separate presentation)

Rapid Prototype Projects

Added Projects for Expanded Set

MISSION	Launch Date	Lead Org PM	Lead Org Flt Sys	NASA Program
1 TDRSS K-L	1/23/14	GSFC	Boeing	Space Comm
2 MAVEN	11/18/13	GSFC	LMA	Planetary
3 LADEE	9/6/13	GSFC	ARC	Planetary
4 IRIS	6/27/13	GSFC	LMMS	Astrophysics/SMEX
5 Van Allen Probes	8/30/12	GSFC	APL	Heliophysics/LWS
6 NuSTAR	6/13/12	JPL	OSC	Astrophysics/Explorer
7 MSL	11/26/11	JPL	JPL/LMA	Planetary/Mars Expl
8 GRAIL	9/10/11	JPL	LMA	Planetary/Discovery
9 Juno	8/5/11	JPL	LMA	Planetary/New Frontiers
10 Glory	3/4/11	GSFC	OSC/Swales	Earth Sciences
11 GOES (-P)	3/4/10	GSFC/NOAA	Boeing/SGT	Earth Sciences
12 SDO	2/11/10	GSFC	GSFC	Heliophysics
13 WISE	12/14/09	JPL	BATC	Astrophysics/Explorer
14 LCROSS	6/18/09	ARC	NG	Planetary/Discovery
15 LRO	6/18/09	GSFC	GSFC	Planetary
16 KEPLER	3/6/09	JPL	BATC	Astrophysics/Discovery
17 OCO	2/24/09	JPL	OSC	Earth Science
18 IBEX	10/19/08	SwRI	OSC	Astrophysics/Explorer
19 DAWN	9/27/07	JPL	OSC/JPL	Planetary/Discovery
20 Phoenix	8/4/07	JPL	LMA	Planetary
21 AIM	4/25/07	LASP	OSC	Heliophysics
22 THEMIS	2/17/07	UCB	Swales	Astrophysics/Explorer
23 STEREO	10/26/06	GSFC	APL	Heliophysics
24 CLOUDSAT	4/28/06	GSFC	BATC	Earth Sciences
25 NEW HORIZONS	1/19/06	APL	APL	Planetary/New Frontiers
26 MRO	8/12/05	JPL	LMA	Planetary/Mars Expl
27 DEEP IMPACT	1/12/05	JPL	BATC	Planetary/Discovery
28 Swift	11/20/04	GSFC	Spectrum Astro	Astrophysics/Explorer
29 MESSENGER	8/3/04	APL	APL	Planetary/Discovery
30 Spitzer	8/25/03	JPL	LMA	Astrophysics
31 MER	6/10/03	JPL	JPL	Planetary/Mars Expl
32 GALEX	4/28/03	JPL	OSC	Astrophysics/Explorer
33 RHESSI	2/5/02	UCB	Spectrum Astro	Heliophysics
34 TIMED	12/7/01	APL	APL	Earth Sciences
35 GENESIS	8/8/01	JPL	LMA	Planetary/Discovery
36 Mars Odyssey	7/7/01	JPL	LMA	Planetary/Mars Expl
37 WMAP	6/30/01	GSFC	GSFC	Astrophysics/Explorer
38 WIRE	3/5/99	GSFC	GSFC	Astrophysics/Explorer
39 TRACE	4/2/98	GSFC	GSFC	Astrophysics/Explorer
40 Cassini	10/15/97	JPL	JPL	Planetary/Outer Planets
41 Mars Global Surveyor	11/7/96	JPL	LMA	Planetary/Mars Expl
42 NEAR	2/17/96	APL	APL	Planetary/Discovery



Data Levels



Level 1

Total PM/SE/MA/I&T (All Phases)

Level 2

PM Total
(All Phases)

SE Total
(All Phases)

MA Total
(All Phases)

I&T Total
(All Phases)

Level 3 (Dsgn.)

PM – Dsgn.
Phase

SE – Dsgn.
Phase

MA – Dsgn.
Phase

I&T – Dsgn.
Phase

Level 3 (Fab.)

PM – Fab.
Phase

SE – Fab.
Phase

MA – Fab.
Phase

I&T – Fab.
Phase

Level 3 (I&T)

PM – I&T
Phase

SE – I&T
Phase

MA – I&T
Phase

I&T – I&T
Phase

Level 3 (LOCO)

PM – LOCO
Phase

SE – LOCO
Phase

MA – LOCO
Phase

I&T – LOCO
Phase

Design (Dsgn.) Phase is defined as the period between start of **Phase B** and **CDR**

Fabrication (Fab.) Phase is defined as the period between **CDR** and **SIR**

Integration & Test (I&T) Phase is defined as the period between **SIR** and **Ship** (to launch site)

Launch Ops. & Check Out (LOCO) Phase is defined as the period between **Ship** and **End of On-Orbit C/O**



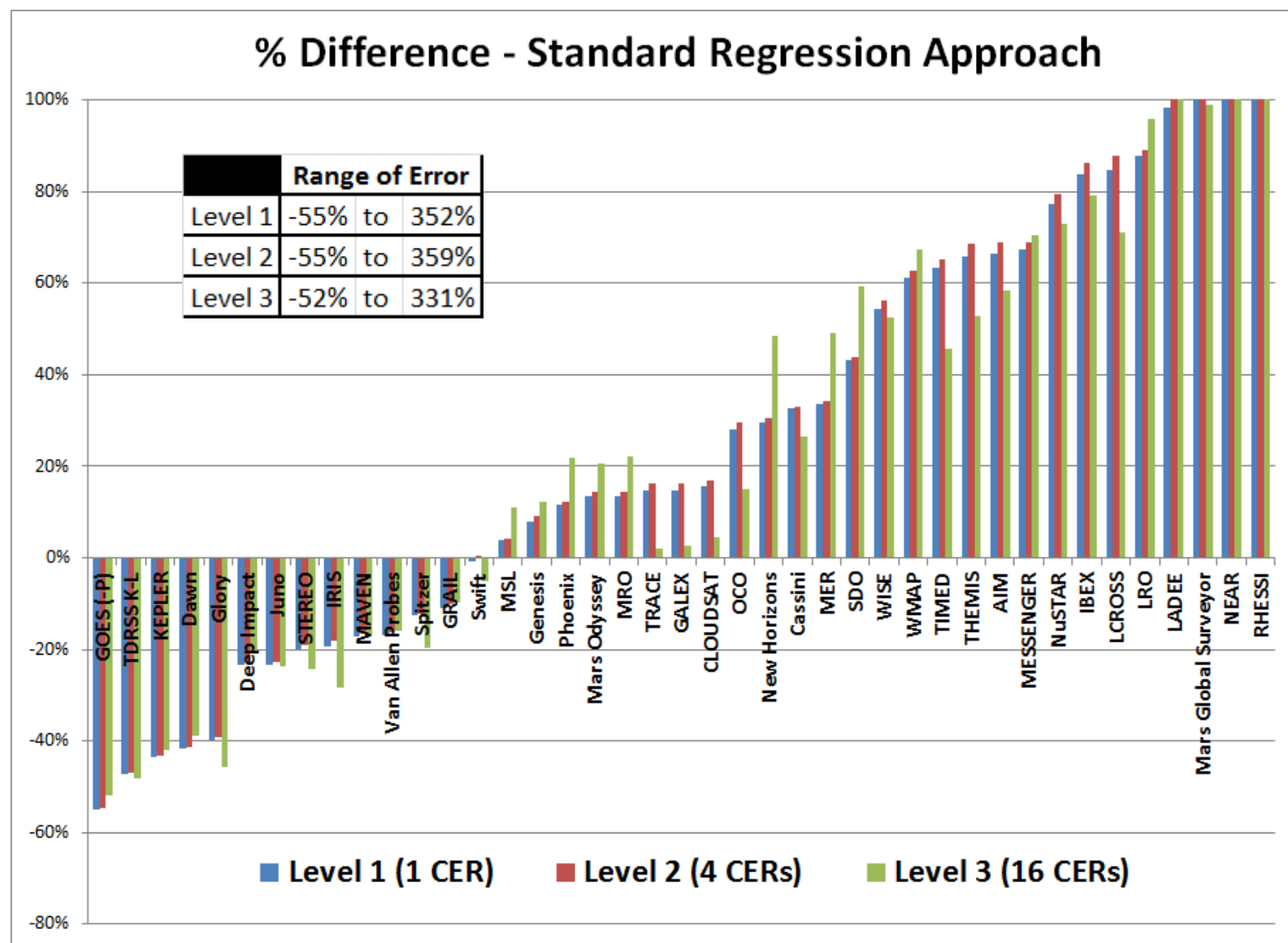
Model Development Standard Regression Approach



- A standard regression analysis was performed using Excel to give a baseline for the analysis.
 - Best fits were mostly linear
 - Outliers were present but were not removed
 - R^2 values ranged from approximately 0.2 to 0.8
- The relationships obtained from this analysis were used to estimate the observations in the data set at all 3 levels.
 - The percent difference was approximately the same across all the levels at -60% to 350%



Standard Regression Approach Results





Model Development Constructive Approach



1) Developed ranges for each item across each development phase using normalized data results

	DB LOW \$/K/mo DES	DB LOW \$/K/mo FAB	DB LOW \$/K/mo I&T	DB LOW \$/K/mo LOCO	DB HIGH \$/K/mo DES	DB HIGH \$/K/mo FAB	DB HIGH \$/K/mo I&T	DB HIGH \$/K/mo LOCO
PM	59.1	36.7	65.4	3.3	2,005	1,650	1,150	1,122
SE	25.0	13.3	13.2	0.2	3,358	2,525	2,248	2,768
MA	25.3	6.4	2.1	0.2	1,573	1,765	1,163	731
I&T	0.3	2.9	10.7	0.1	1,857	1,763	3,762	6,170

2) Assigned input weightings – this process used a combination of common-sense (using past studies and experience) and results from statistical analysis (= *cheating*)

Weightings												kg	W	W/kg					kg	W	W/kg						
	DIRECTED or AO	MISSION RISK CLASS	MISSION DESTINA TION	FLIGHT SYSTEM TYPE	MULTIPL E FLIGHT SYSTEMS ?	LEAD ORGANIZ ATION	FLIGHT SYSTEM ORGANIZ ATION	PAYLOAD ORG.	LEAD ORGANIZ ATION EXPERIEN CE	FLIGHT SYSTEM LEAD ORG. EXPERIEN CE	PAYLOAD LEAD ORG. EXPERIEN CE	FLIGHT SYSTEM MASS	FLIGHT SYSTEM POWER	FLIGHT SYSTEM POWER/ MASS RATIO	NUMBER OF SPACER AFT	FLIGHT SYSTEM HERITAG E & TRL	PARTS RATING	# OF KEY SPACER AFT CONTRA CTORS	PAYLOAD MASS	PAYLOAD POWER	PAYLOAD POWER/ MASS RATIO	# OF PAYLOAD ELEMENT S	# OF KEY PAYLOAD CONTRA CTORS	IN- HOUSE SCOPE	INTERNA TIONAL PARTICIP ATION (HW)		
Input #	1	0.5	0.35	0.3	0.1	0	0.05	0.05		0.2	0.2	0.2	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.2	0.1		
	2	1.2	0.5	0.6	0.3	1	0.1	0.1		0.4	0.4	0.4	1	1	1	1	1	1	1	1	1	1	1	0.4	1		
	3		0.75	0.6	0.4		0.6	0.6		0.6	0.6	0.6												0.6			
	4		1.8	0.7	0.6		1.2	1.3		0.8	0.8	0.8												0.8			
	5			0.9	0.8		1.8	1.2		1	1	1												1			
	6			1.747	1.5																						
	7			2																							

3) Input weightings include weightings for each input option and for each input to a specific WBS/phase

Correlations		Key Performance Indicators (KPIs)																								
		Directed or AO	Mission Risk Class	Mission Destination	Flight System Type	Multiple Flight Systems?	Lead Organization	Flight System Organization	Payload Org.	Lead Organization Experience	Flight System Lead Org. Experience	Payload Lead Org. Experience	Flight System Mass	Flight System Power	Flight System Power/Mass Ratio	Number of Spacecraft	Flight System Heritage & TRL	Parts Rating	# of Key Spacecraft Contractors	Payload Mass	Payload Power	Payload Power/Mass Ratio	# of Payload Elements	# of Key Payload Contractors	In-House Scope	International Participation (HW)
PM	DES	0	1.05	0.2625	0	0.525	0	0.525	0.7875	0.7875	0.525	1.3125	0.7875	1.3125	0	0	1.05	0	1.05	1.3125	0	0	0	0	0	0
	FAB	0	0.525	0.2625	0	0.525	0	0.7875	0.525	0.7875	0.525	0.7875	0.7875	1.05	0.525	0	0	1.05	0.525	0.525	0.525	0	0	0	0	
	I&T	0	0	0.2625	0	0.525	0	0.7875	0	0	0	0.7875	0	0	0.525	0	0	0.7875	0.525	0	0	0	0	0	0	
	LOCO	0	0	0.525	0.525	0.525	0.525	0.525	0.525	0	0	0.525	1.05	0.7875	0.525	0	0	0.525	1.05	0	0	0.525	0	0	0	
SE	DES	0.4	0.8	0.2	0	0.4	0	0.6	0.6	0.6	0.4	1	0.4	0.6	0	0	0	0.6	0	0	0.6	0.4	0	0	0	
	FAB	0.4	0.6	0.6	0.8	0.4	0.64	0.8	0.4	0.4	0	0.4	0.8	0.4	0	0	0	0.8	0.6	0	0	0.4	0	0	0	
	I&T	0	0.4	0.4	0.8	0.4	0.64	0.6	0	0.4	0	0	0.8	0	0	0	0.8	0.6	0	0	0	0	0	0		
	LOCO	0	0.4	0.4	1	0.4	0.8	0.6	0	0.4	0	0	1.2	0.4	0	0	0	0.8	0.6	0	0.4	0.8	0	0	0	
MA	DES	0	1.25	0.5	0	0.5	0	0.5	0.5	0	0	1	1	0.75	0	0	0	1.25	0	0.5	0.75	0	0	0	0.5	
	FAB	0	0.75	1	0	0.5	0	0.5	0	0	0	0.75	1.5	0.5	0	0	0.5	1.5	0.5	0	0.5	0	0.75	0.5	0	
	I&T	0	0	0.75	0	0.5	0	0.5	0	0	0	0	0.5	0.5	0	0	0.5	1	0	0	0	0	0	0.5	0	
	LOCO	0	0.5	0.5	0	0.5	0	0.5	0	0	0	0	0.5	0.75	0.5	0	0.5	0.75	0.5	0	0	0	0	0	0	
I&T	DES	0	0.75	0.25	0	0.5	0	0	0.5	0.5	0.5	1.25	0.5	0.75	0	0	0	0.75	0	0.75	1.25	0	0	0	0	
	FAB	0	0	0.5	0.5	0.5	0.75	0	0.5	0	0.5	1.25	0.5	0.5	0	0	0	1	0.5	0.75	1	0.5	0.5	0	0	
	I&T	0	0	0.75	1	0.5	1	0.5	0	0	0	1.5	0	0	0	0	0	0.75	0.75	0	0.5	0.75	0.75	0	0	
	LOCO	0	0	0.5	0.75	0.5	0.75	0	0	0	0	1.5	0	0	0	0	0	0.5	0.75	0	1	1	0.75	0	0	

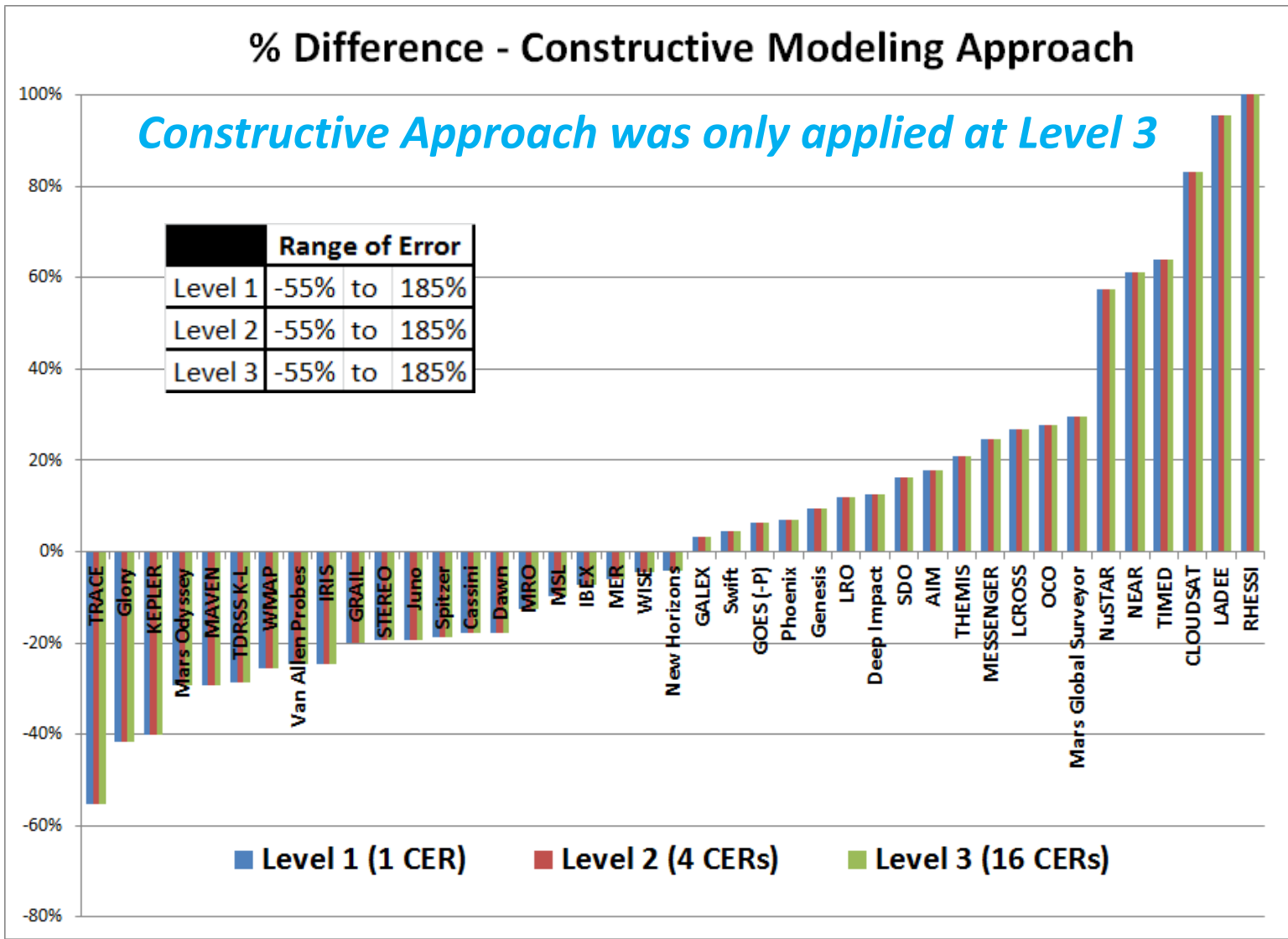




Constructive Approach Results



Engineering
Cost
Office



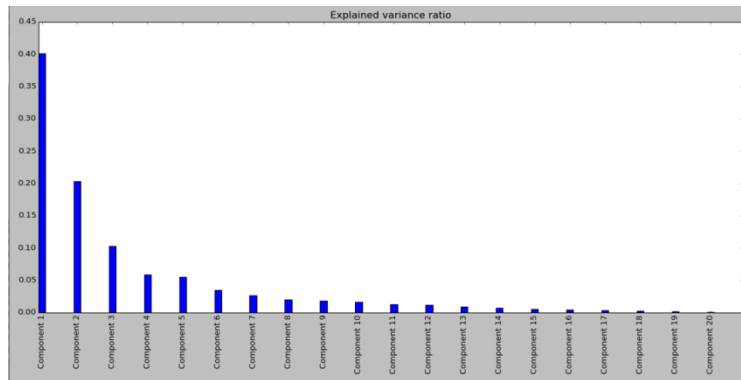


Model Development Principle Component Analysis Approach



Engineering
Cost
Office

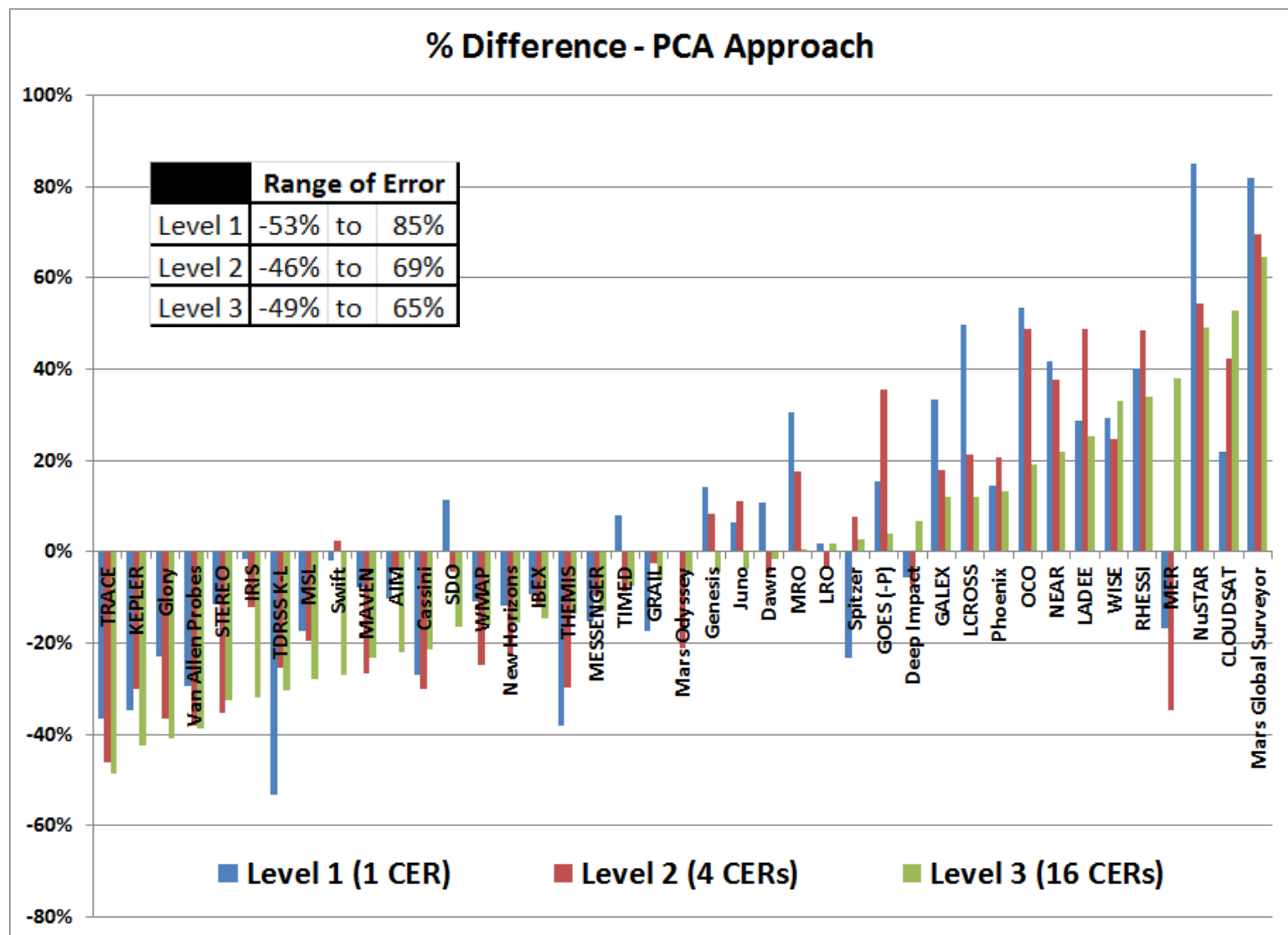
- 1) A correlation matrix was generated to get a sense of the of the dependency between variables.
 - Several of the variables appeared to be correlated, making PCA an attractive method to apply to the data set.



- 2) The principal components were determined using an algorithm developed in Python.
 - The first 6 principal components which account for 85% of variance in the data set were selected and used to determine which of the 20 variables were most likely related to cost.
- 3) For each of the 21 data sets examined, 4 subsets of the 20 variables were run through a multiple regression routine to determine the new cost estimating relationships.



Principle Component Analysis Approach Results





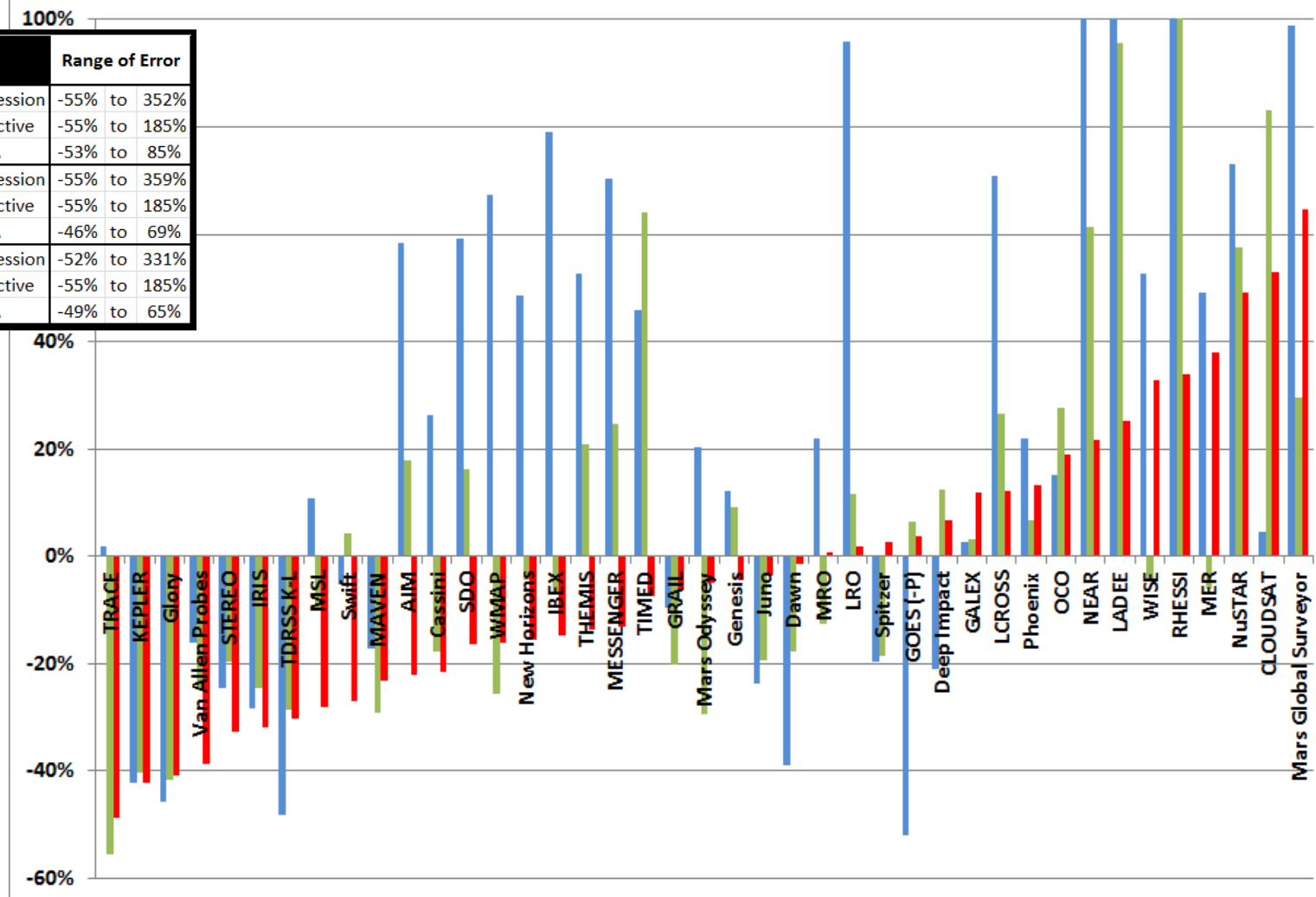
Modeling Method Performance Comparisons



Engineering
Cost
Office

■ Std. Regression Approach - Level 3 ■ Constructive Approach - Level 3 ■ PCA Approach - Level 3

		Range of Error	
Level 1	Std. Regression	-55%	to 352%
	Constructive	-55%	to 185%
	PCA	-53%	to 85%
Level 2	Std. Regression	-55%	to 359%
	Constructive	-55%	to 185%
	PCA	-46%	to 69%
Level 3	Std. Regression	-52%	to 331%
	Constructive	-55%	to 185%
	PCA	-49%	to 65%



TBD





Findings and Next Steps



- Findings

- Each of the modeling approaches developed provides an improvement over the wrap factor approach. The PCA approach provides the best results at levels 1, 2 and 3
 - The Level3 constructive approach range of errors is between -50% - 185%
 - The Level 3 PCA approach range of errors is between -50% - 65%

- Next Steps

- Consider eliminating extreme outliers
- Try to improve the normalized data set by:
 - Increasing the number of normalized observations
 - Improve support function lower level allocations in the normalized data
- Explore non-linear statistical techniques
- Subject Matter Expert review of inputs and relationships



BACKUP



**Engineering
Cost
Office**



BACKUP



**Engineering
Cost
Office**

CONSTRUCTIVE APPROACH DETAILS



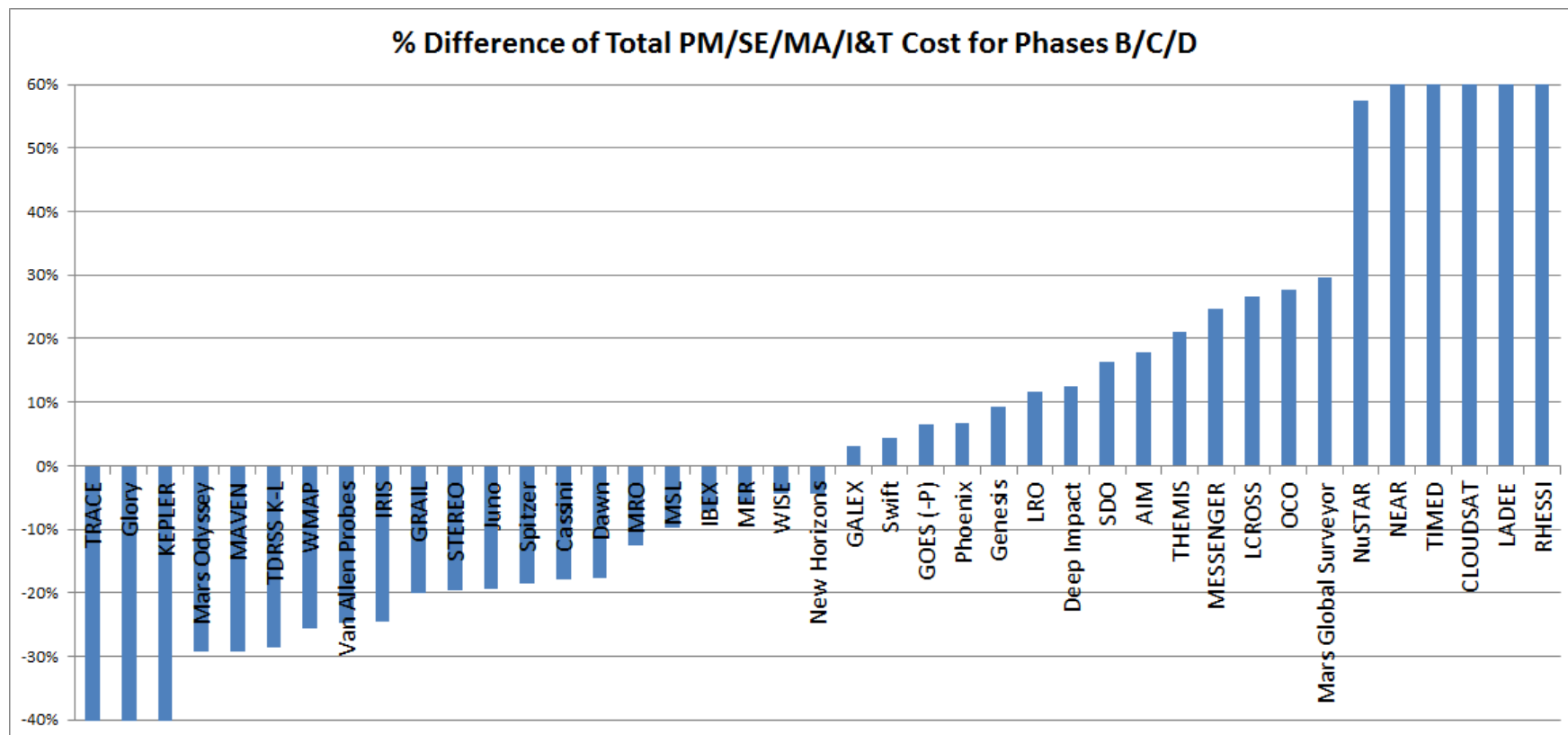


Constructive Approach Results

Total PM/SE/MA/I&T

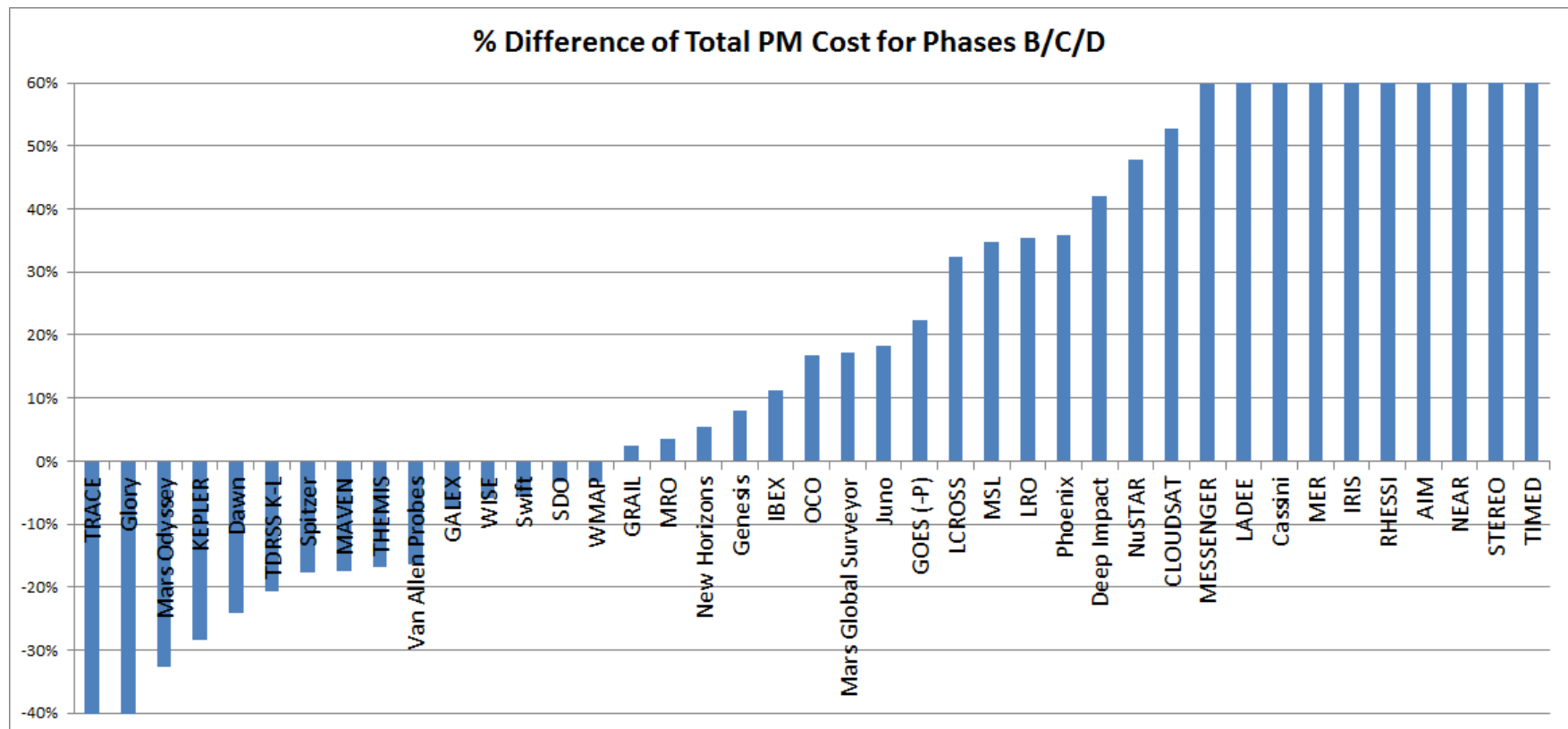


Engineering
Cost
Office





Constructive Approach Results Total PM



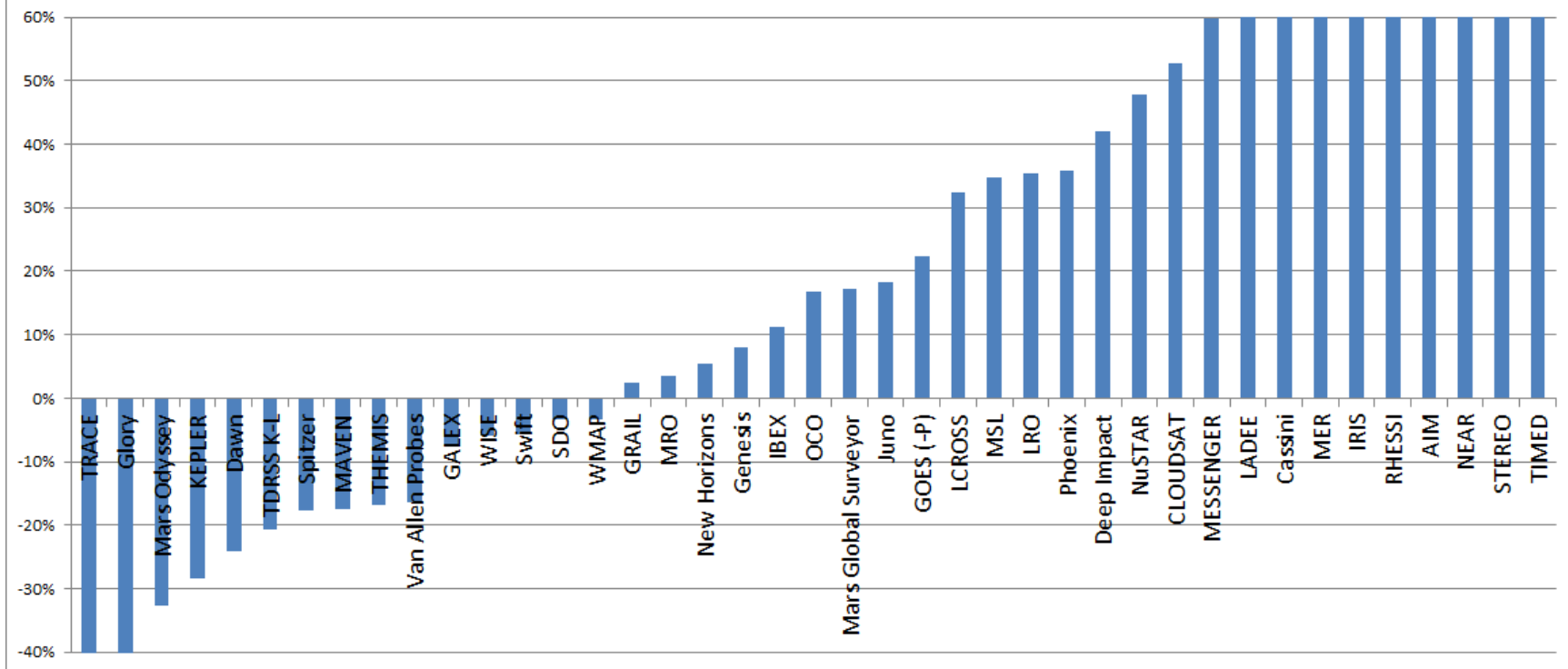


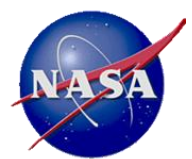
Constructive Approach Results Total SE



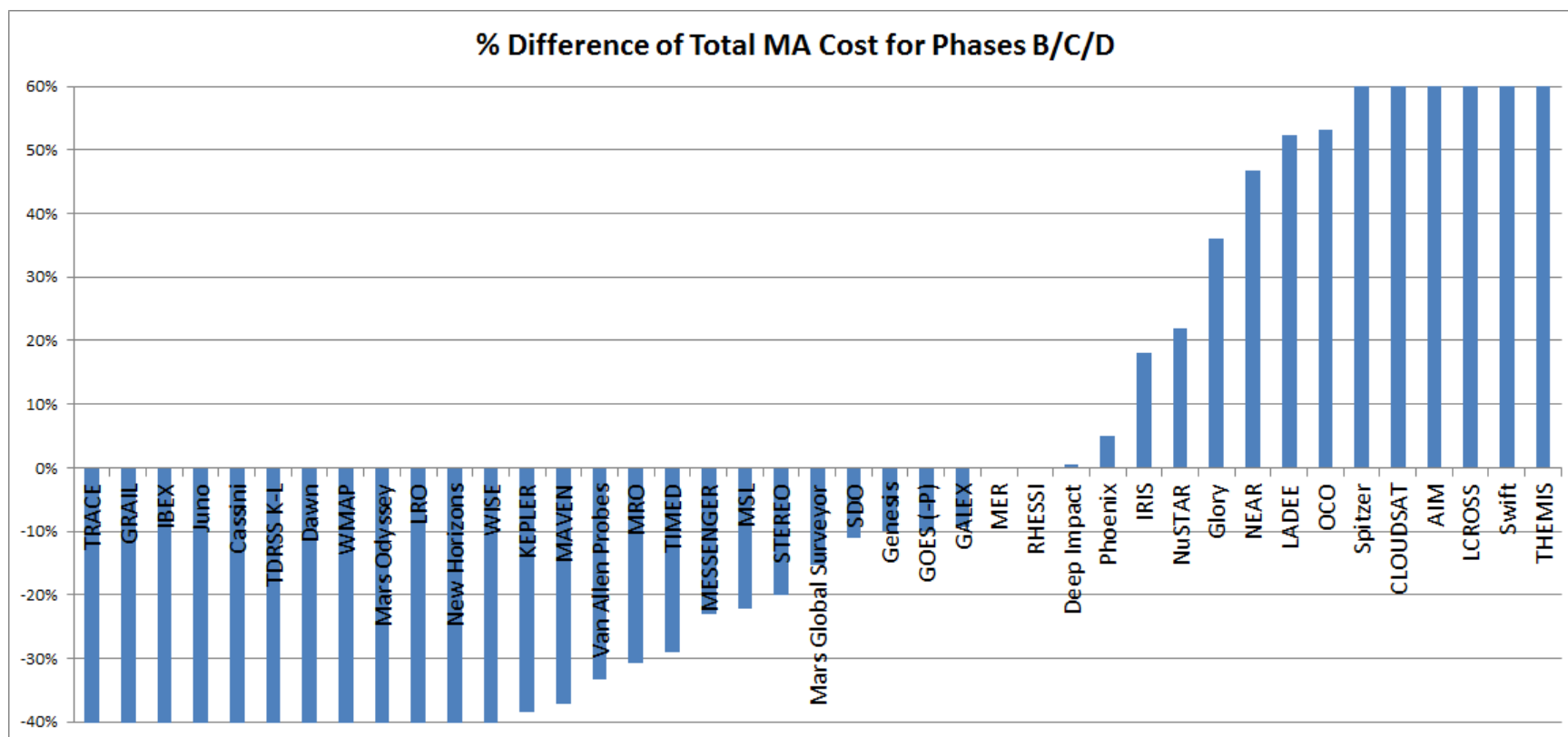
Engineering
Cost
Office

% Difference of Total SE Cost for Phases B/C/D



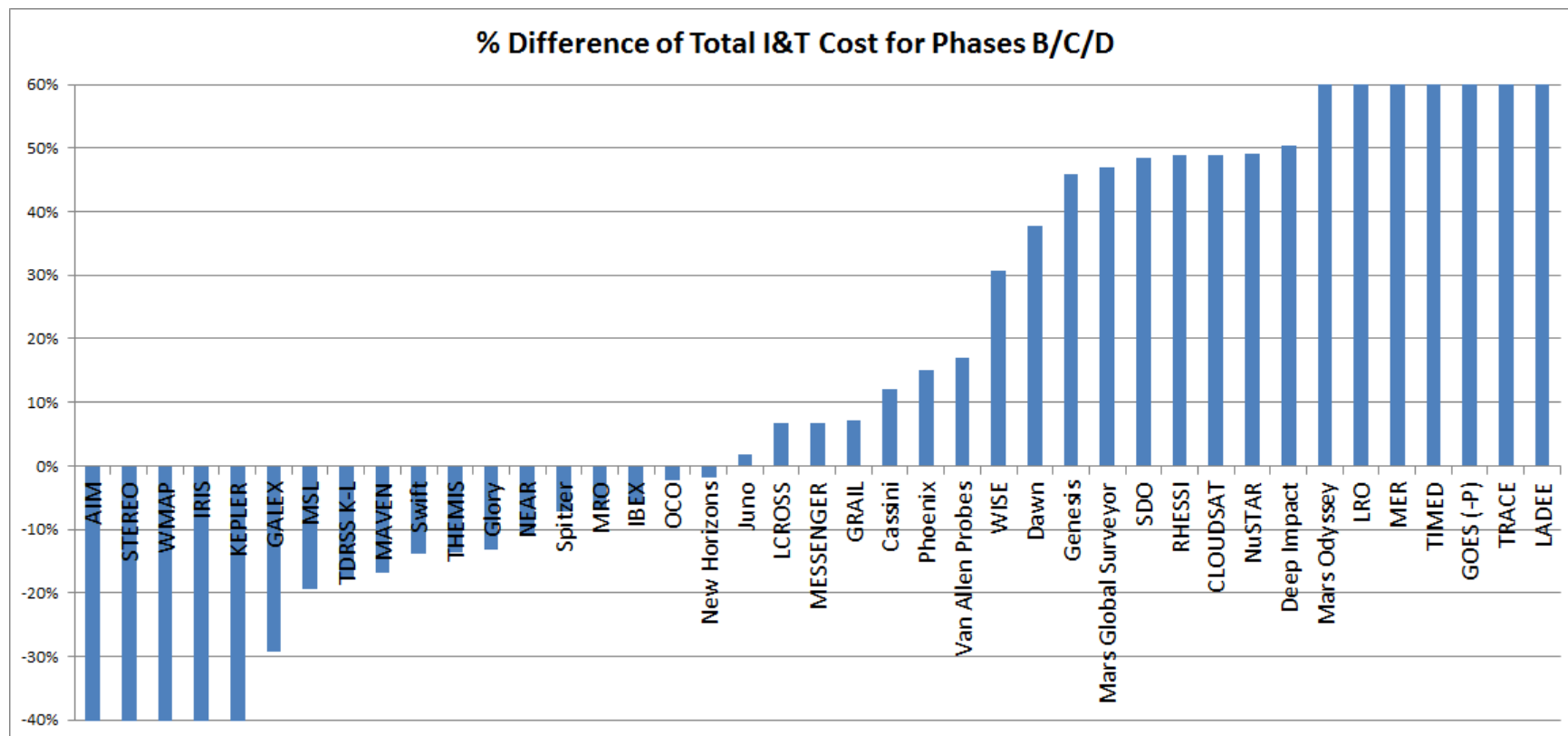


Constructive Approach Results Total MA



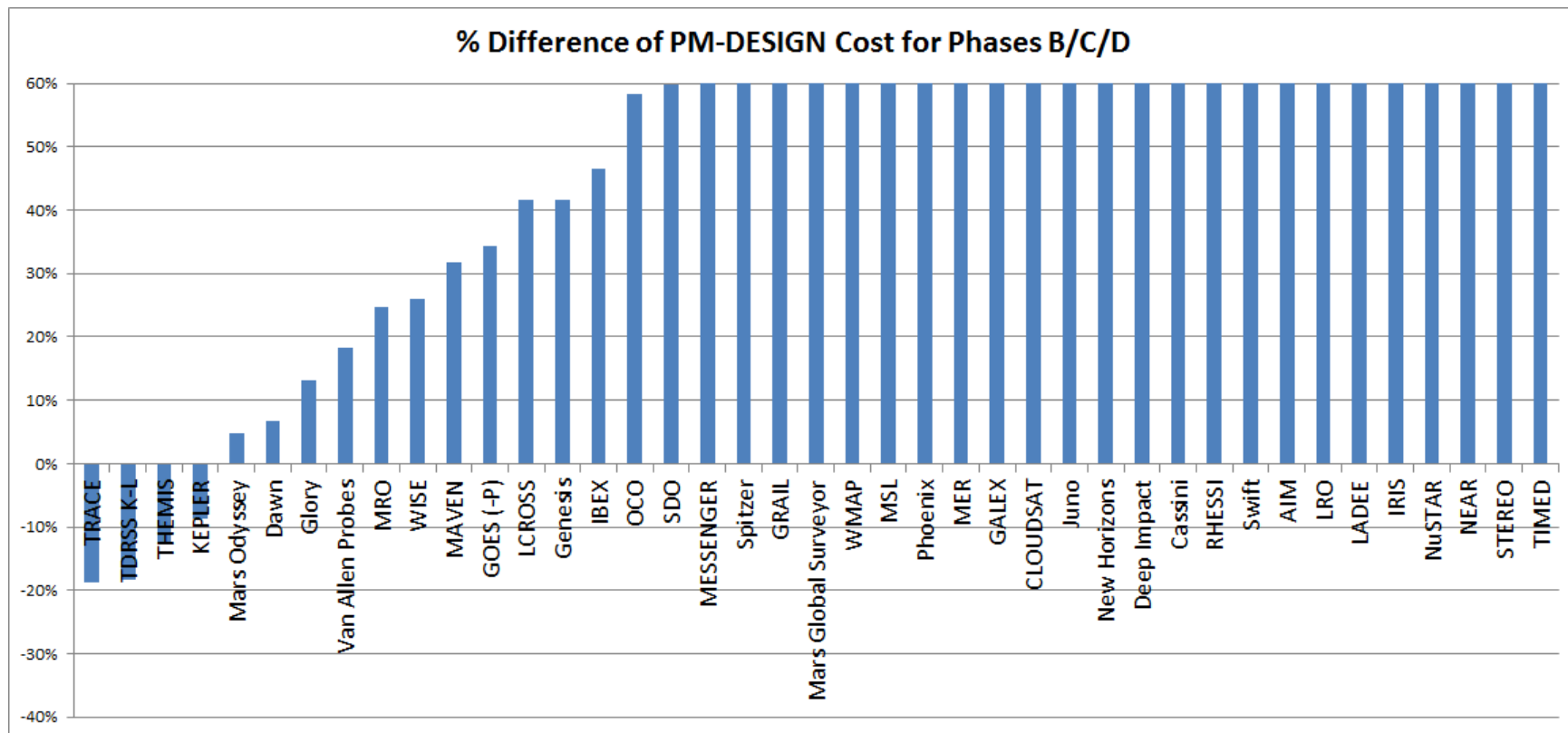


Constructive Approach Results Total I&T



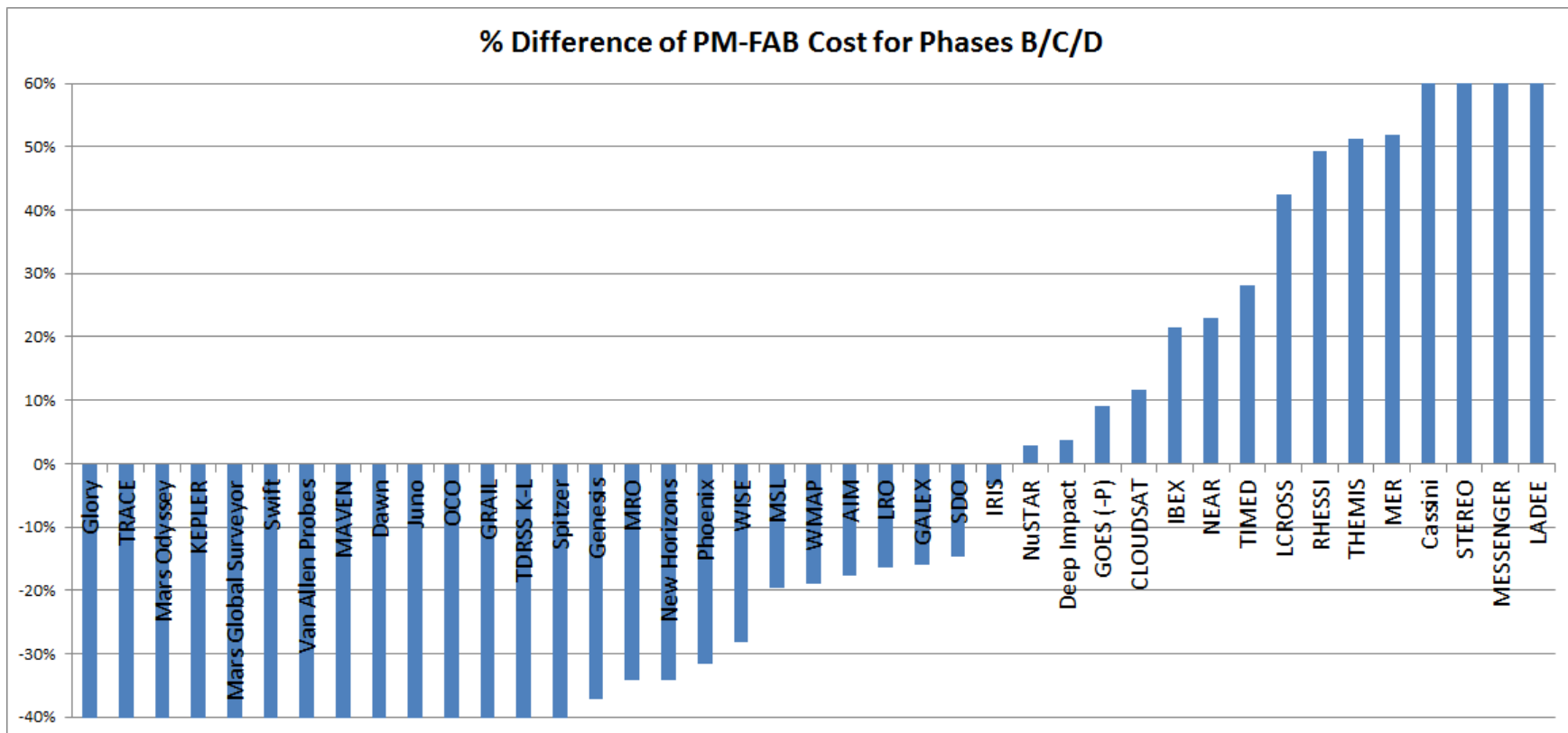


Constructive Approach Results PM - DESIGN





Constructive Approach Results PM - FAB

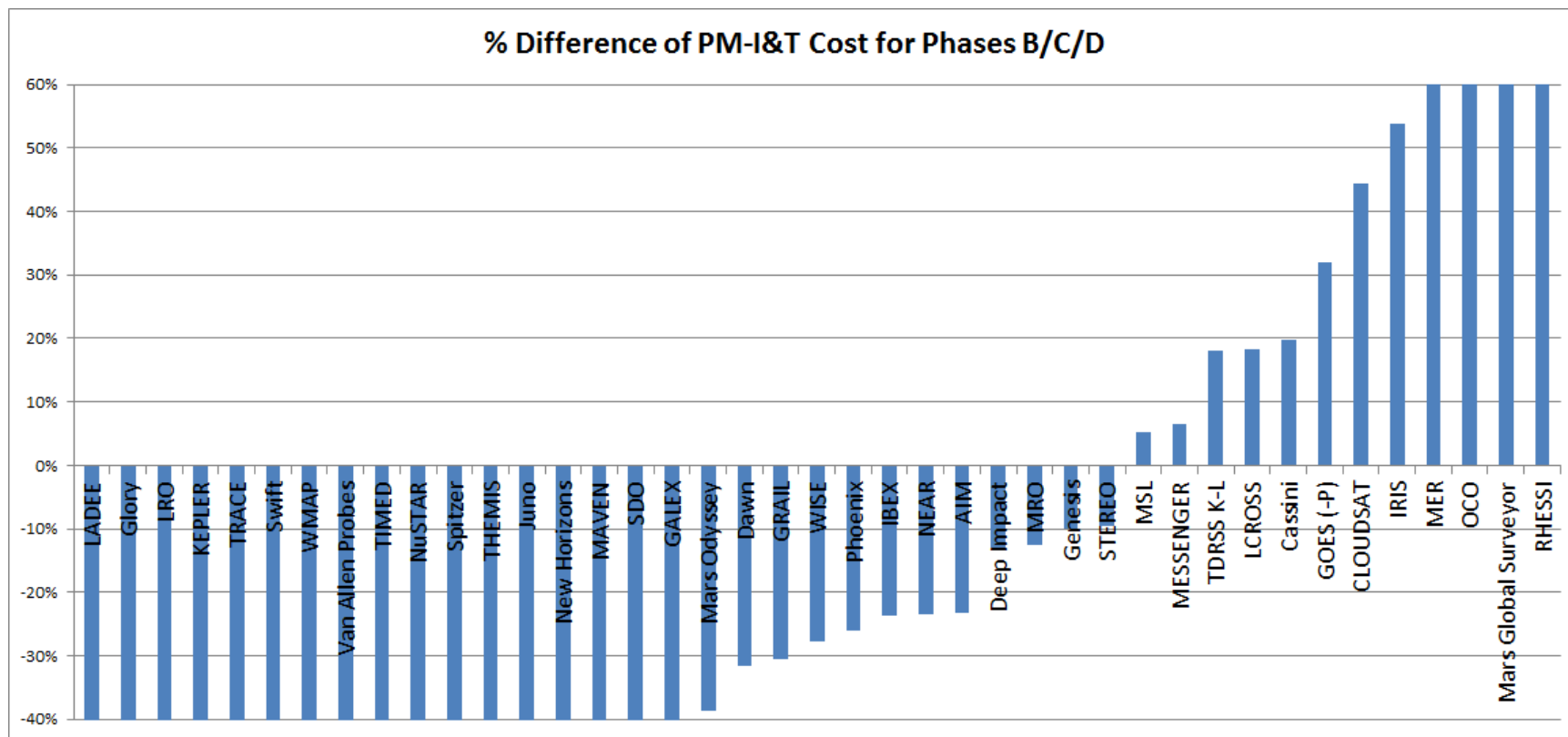




Constructive Approach Results PM – I&T

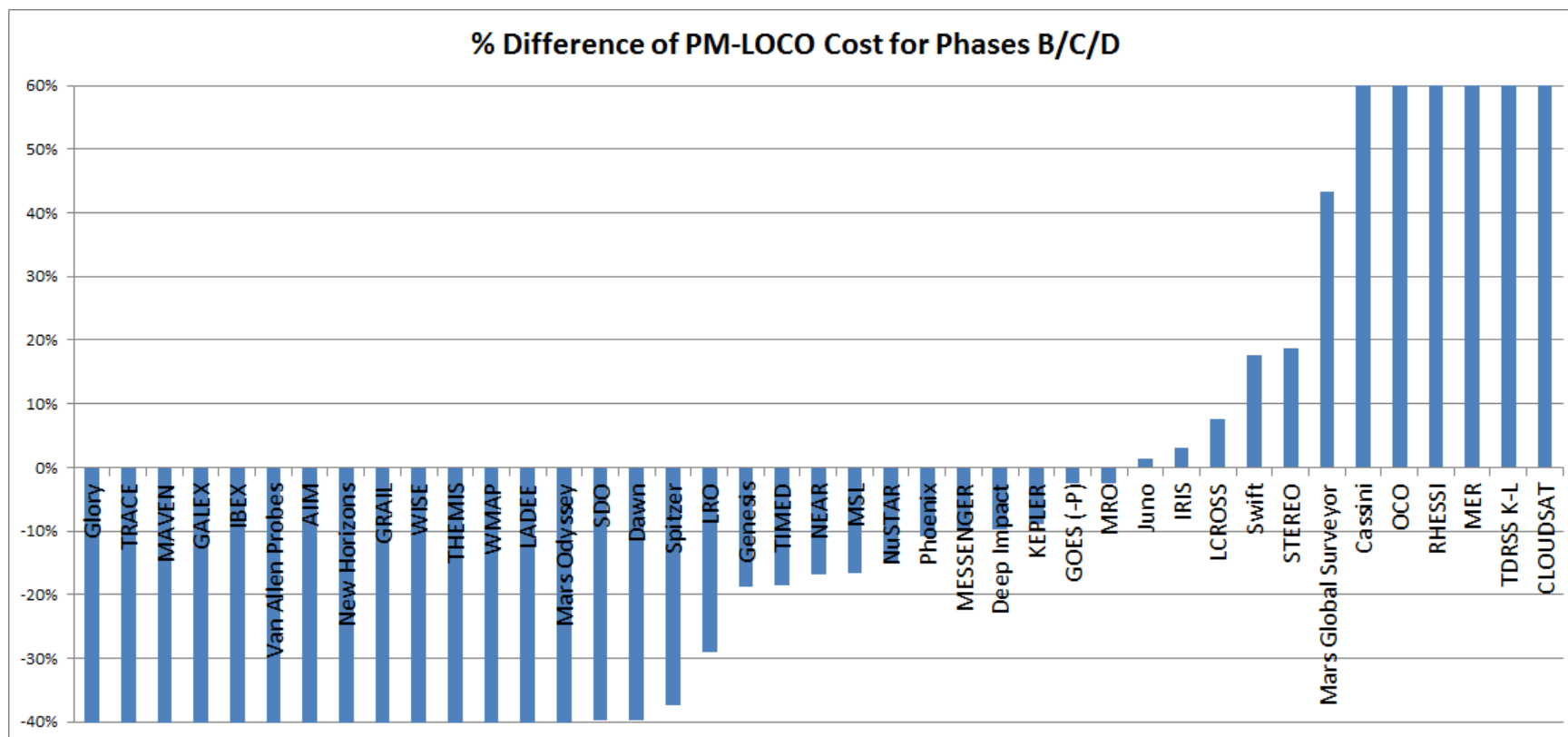


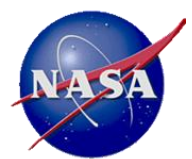
Engineering
Cost
Office





Constructive Approach Results PM - LOCO

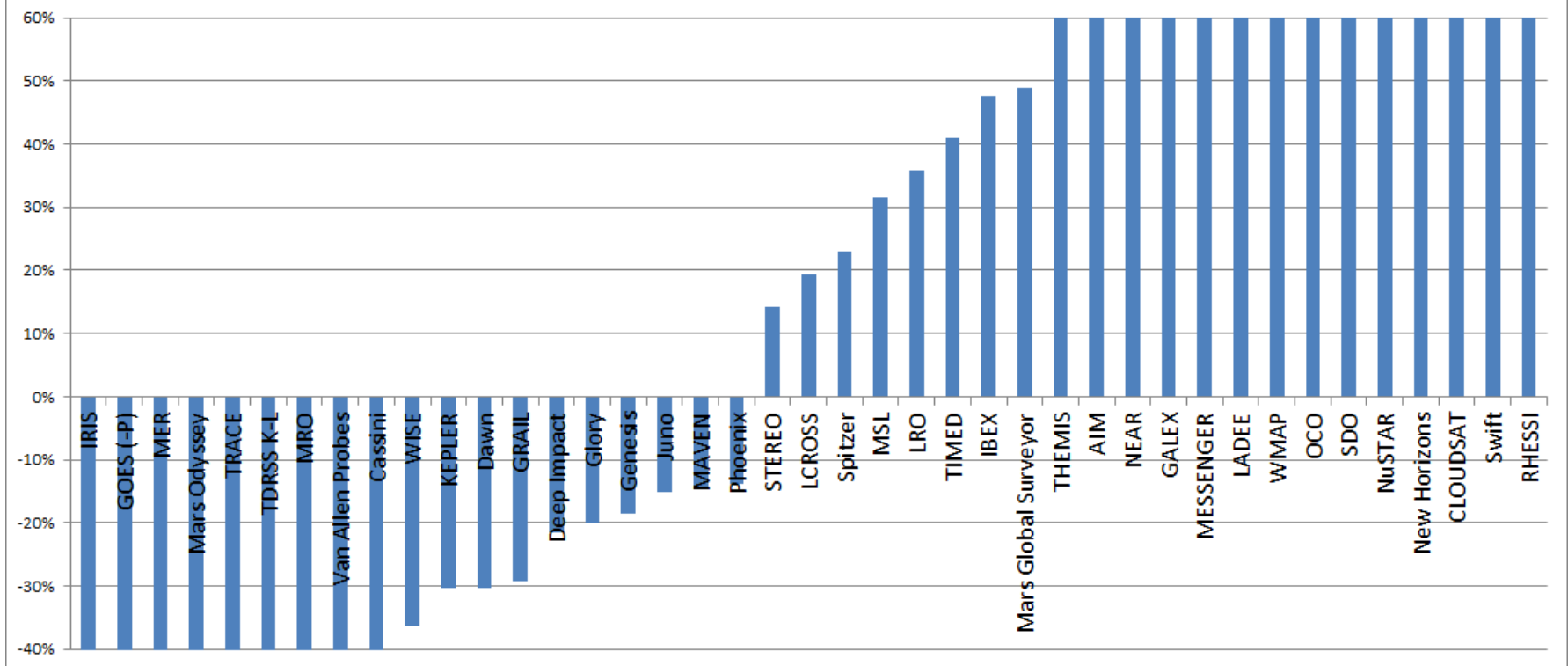




Constructive Approach Results SE - DESIGN



% Difference of SE-DESIGN Cost for Phases B/C/D

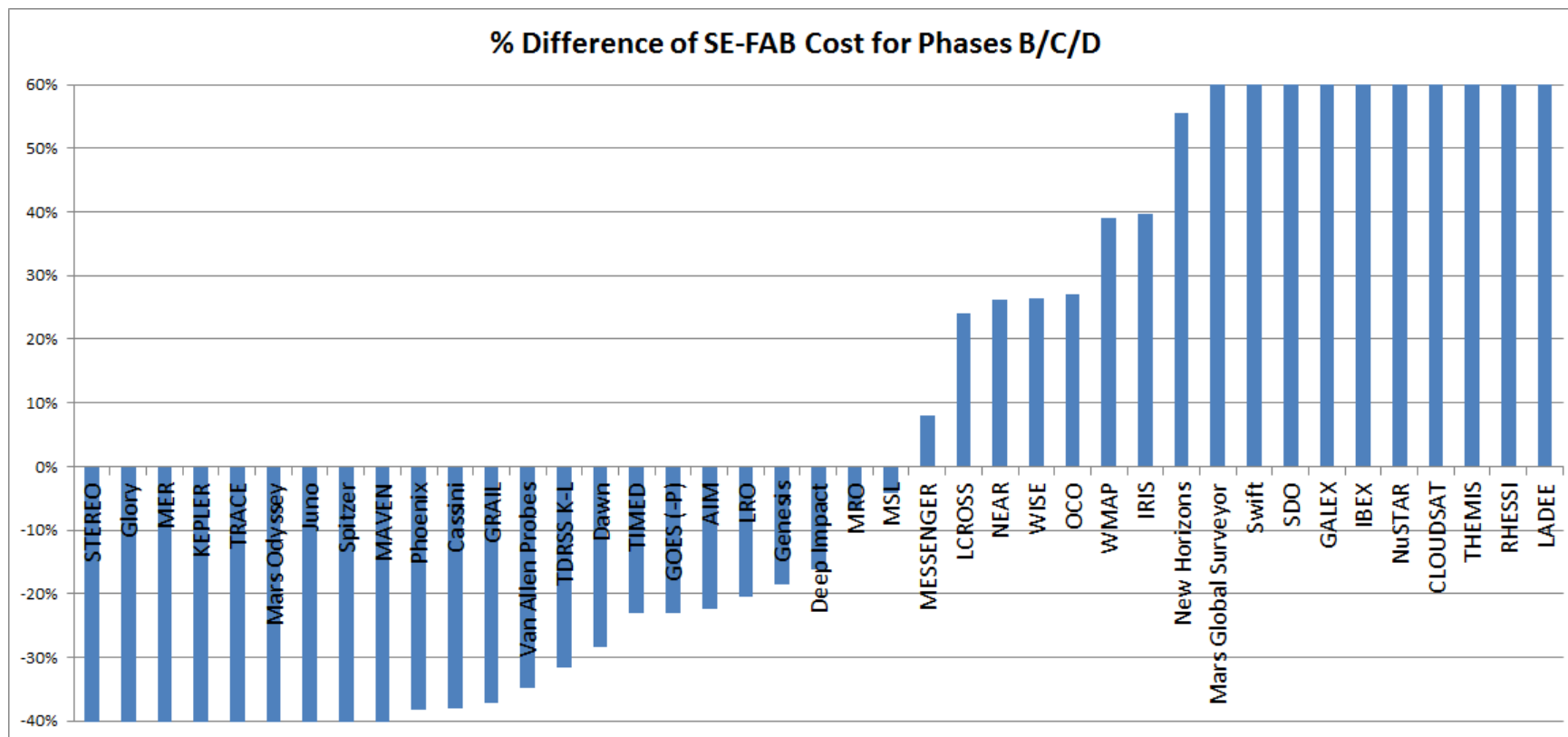




Constructive Approach Results SE - FAB



Engineering
Cost
Office

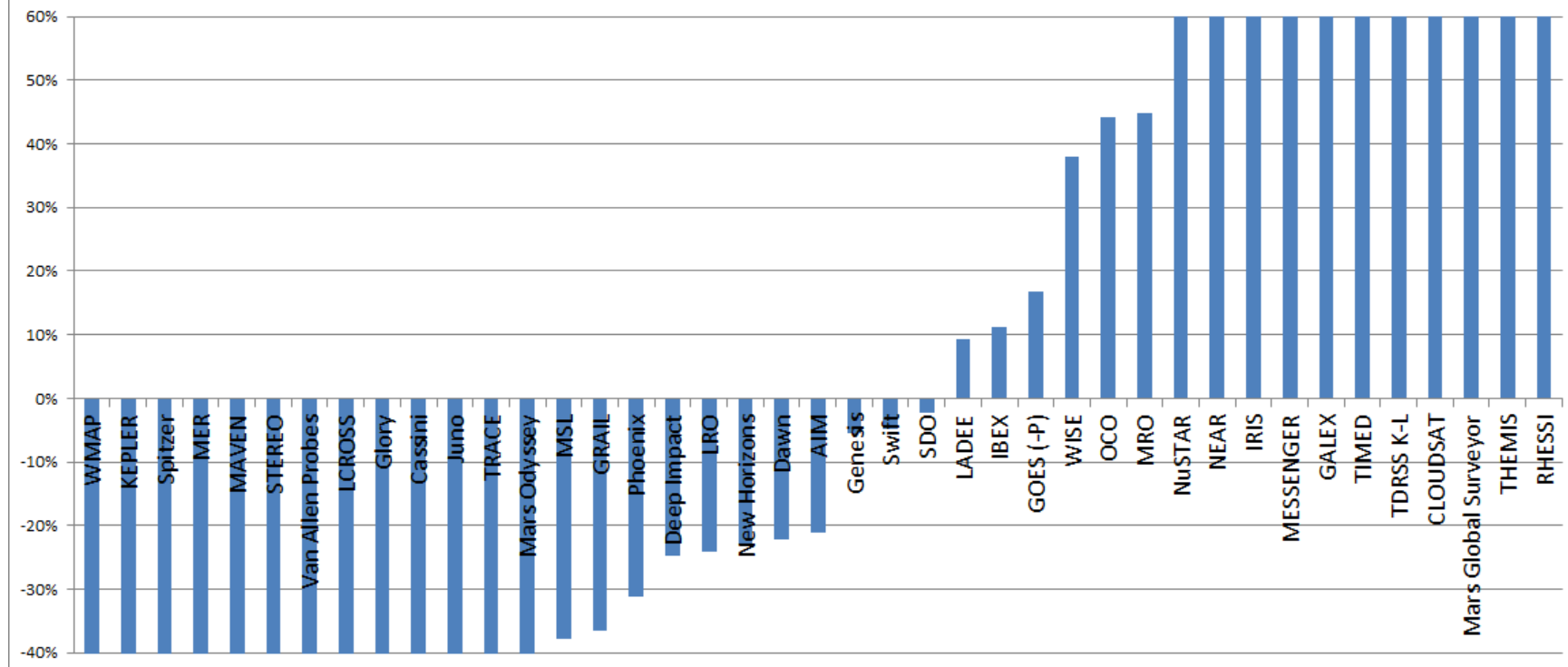




Constructive Approach Results SE – I&T



% Difference of SE-I&T Cost for Phases B/C/D

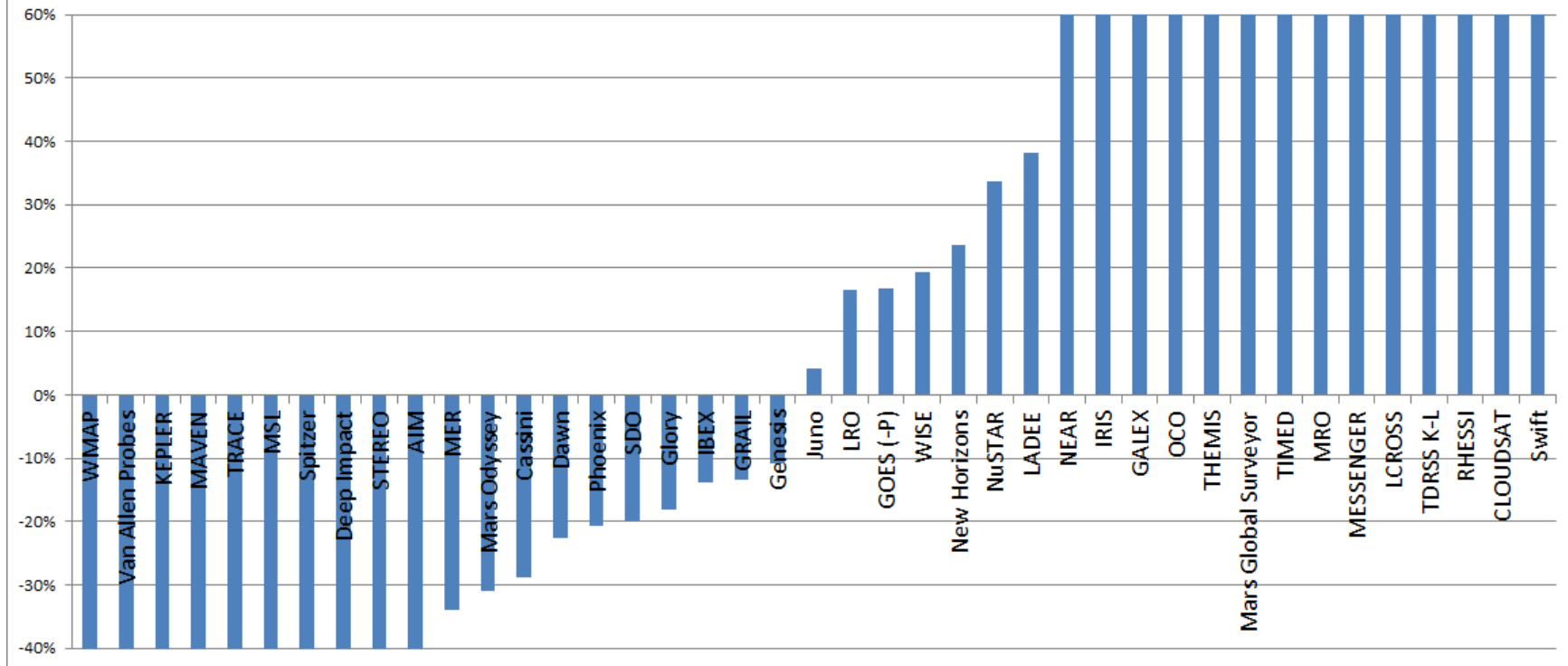




Constructive Approach Results SE - LOCO

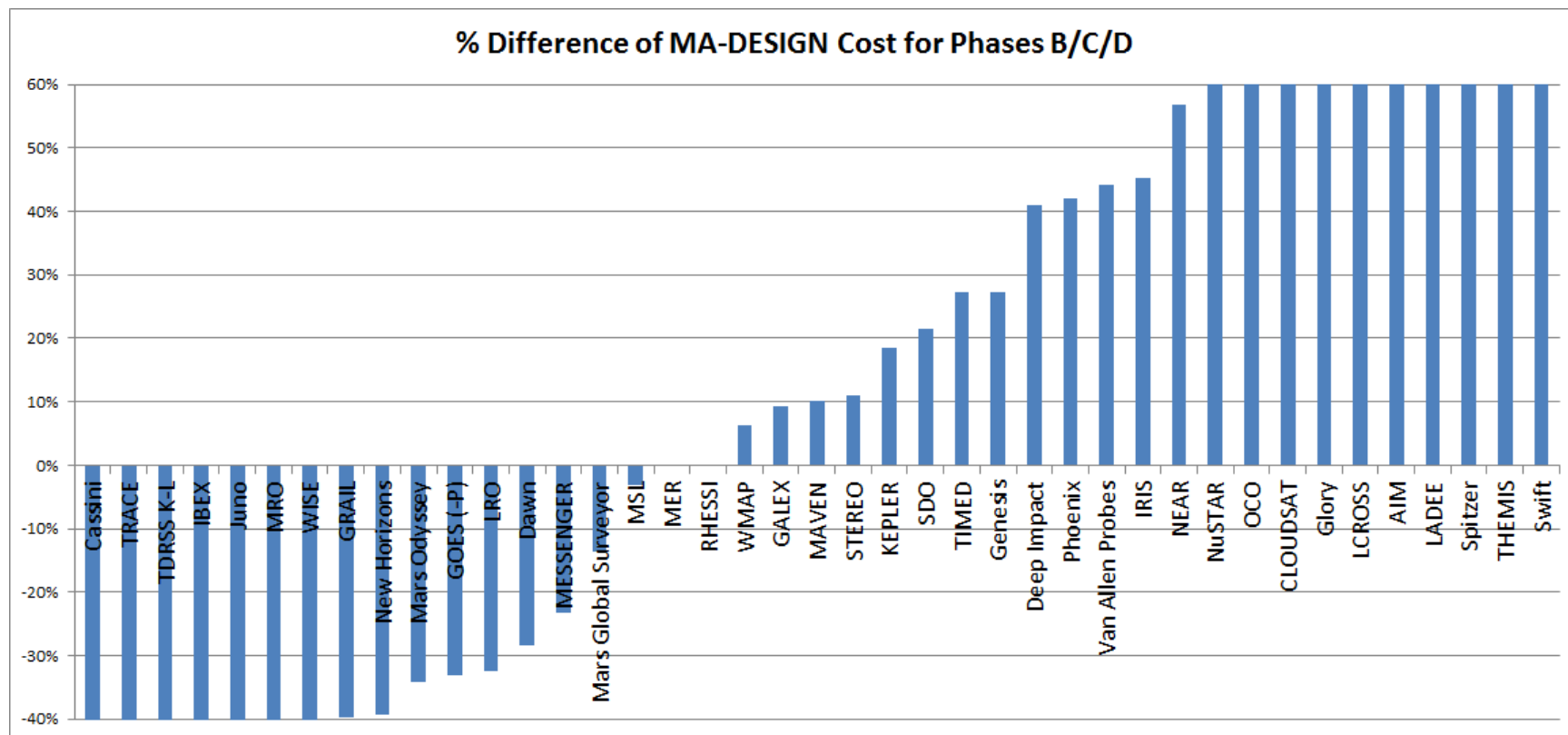


% Difference of SE-LOCO Cost for Phases B/C/D





Constructive Approach Results MA - DESIGN

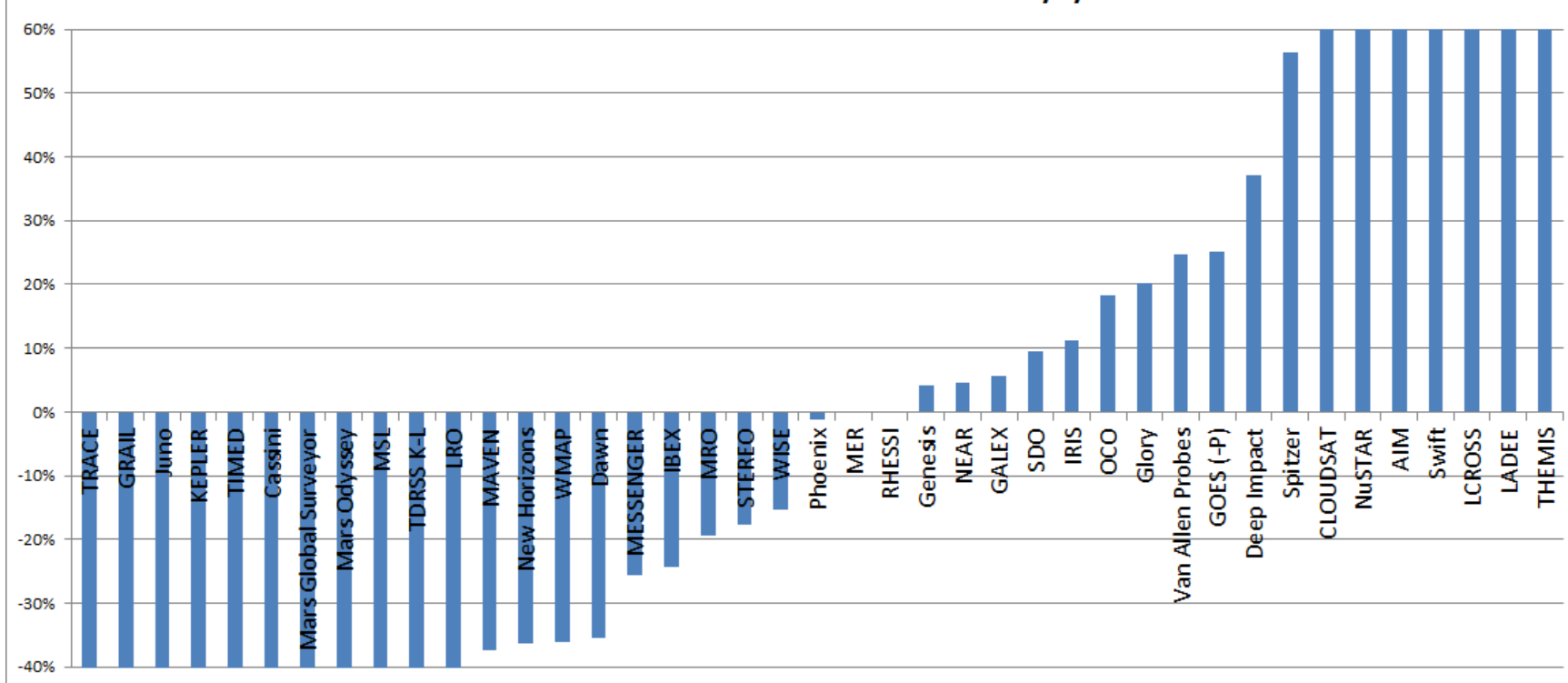


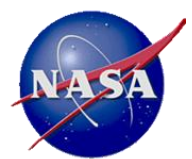


Constructive Approach Results MA - FAB

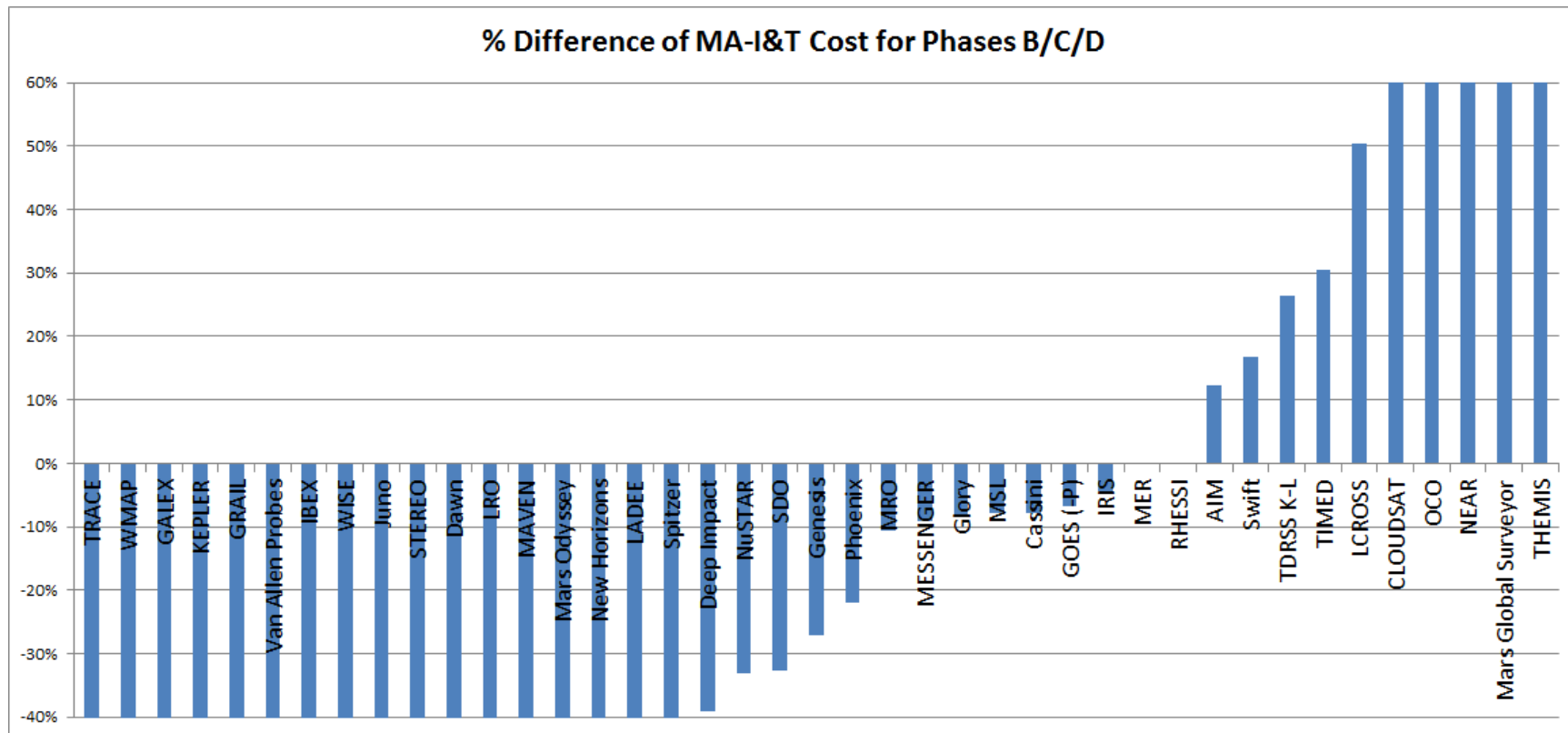


% Difference of MA-FAB Cost for Phases B/C/D



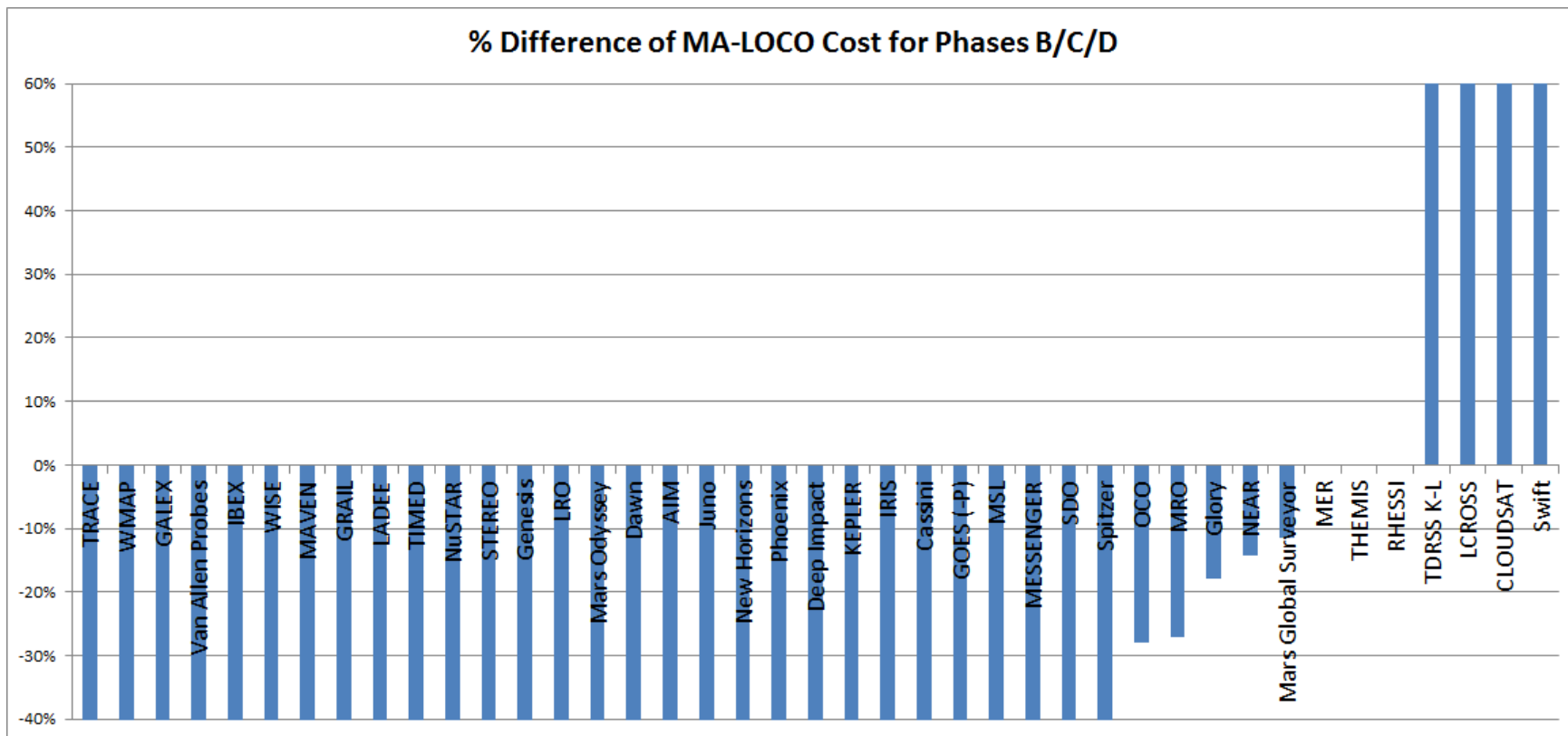


Constructive Approach Results MA – I&T





Constructive Approach Results MA - LOCO

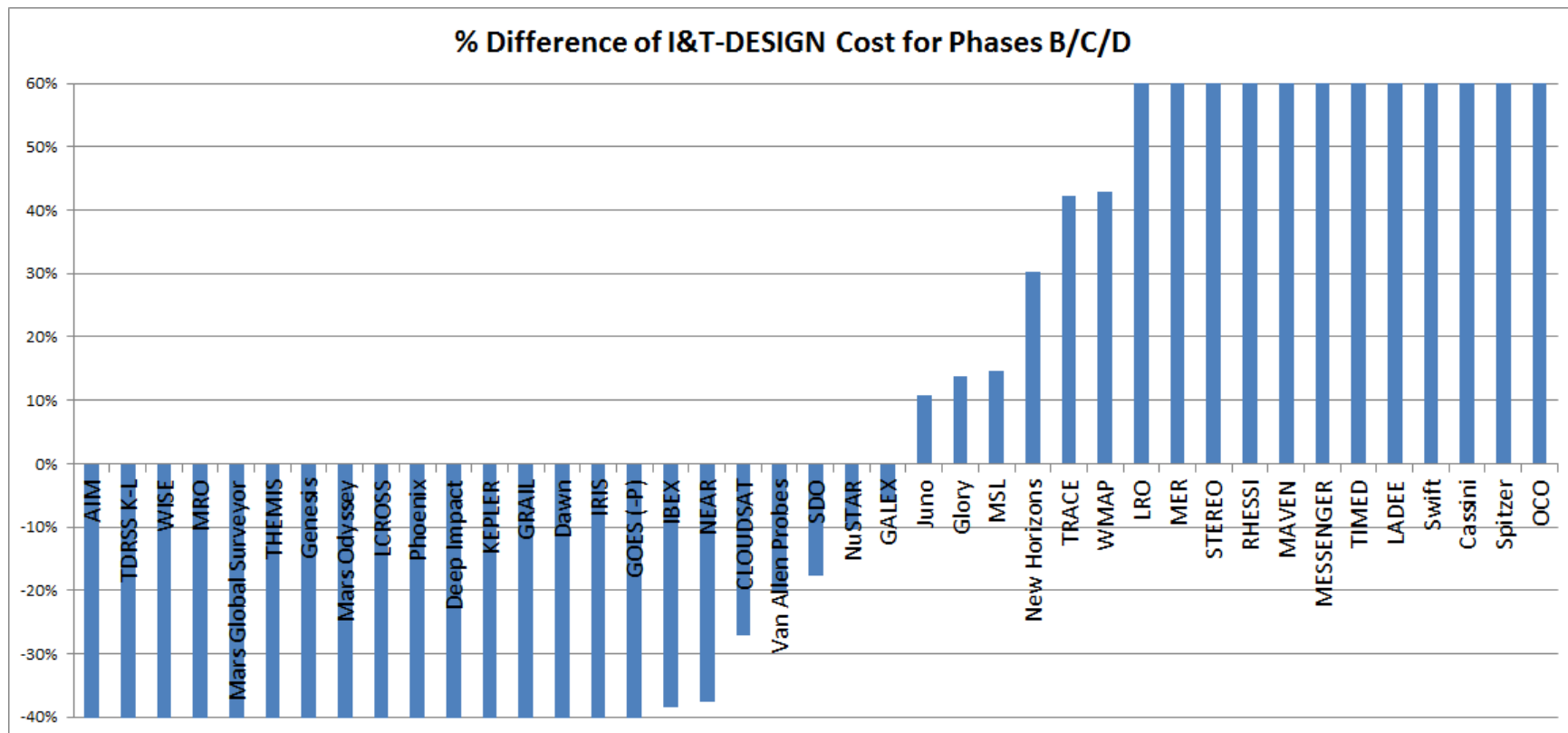




Constructive Approach Results I&T - DESIGN

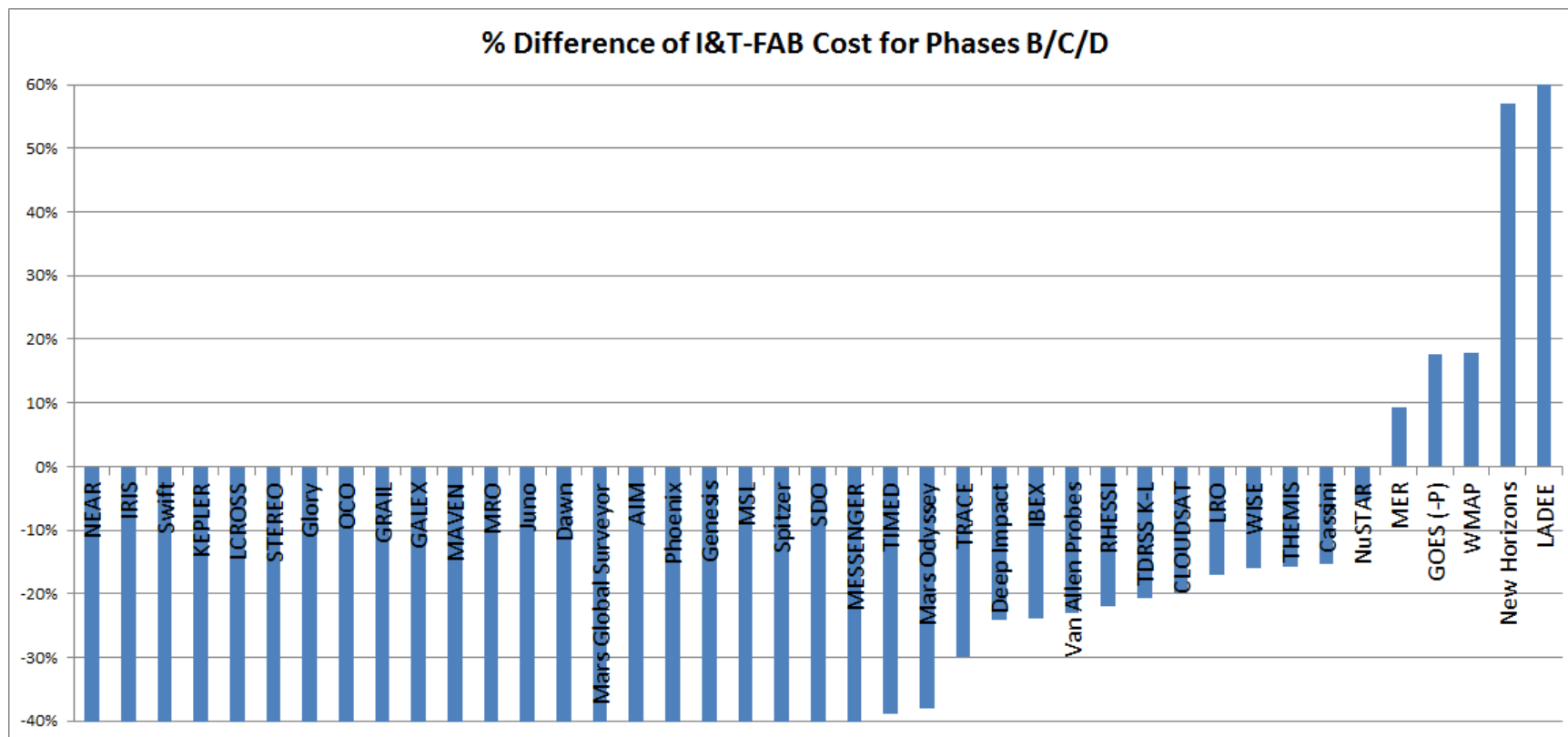


Engineering
Cost
Office



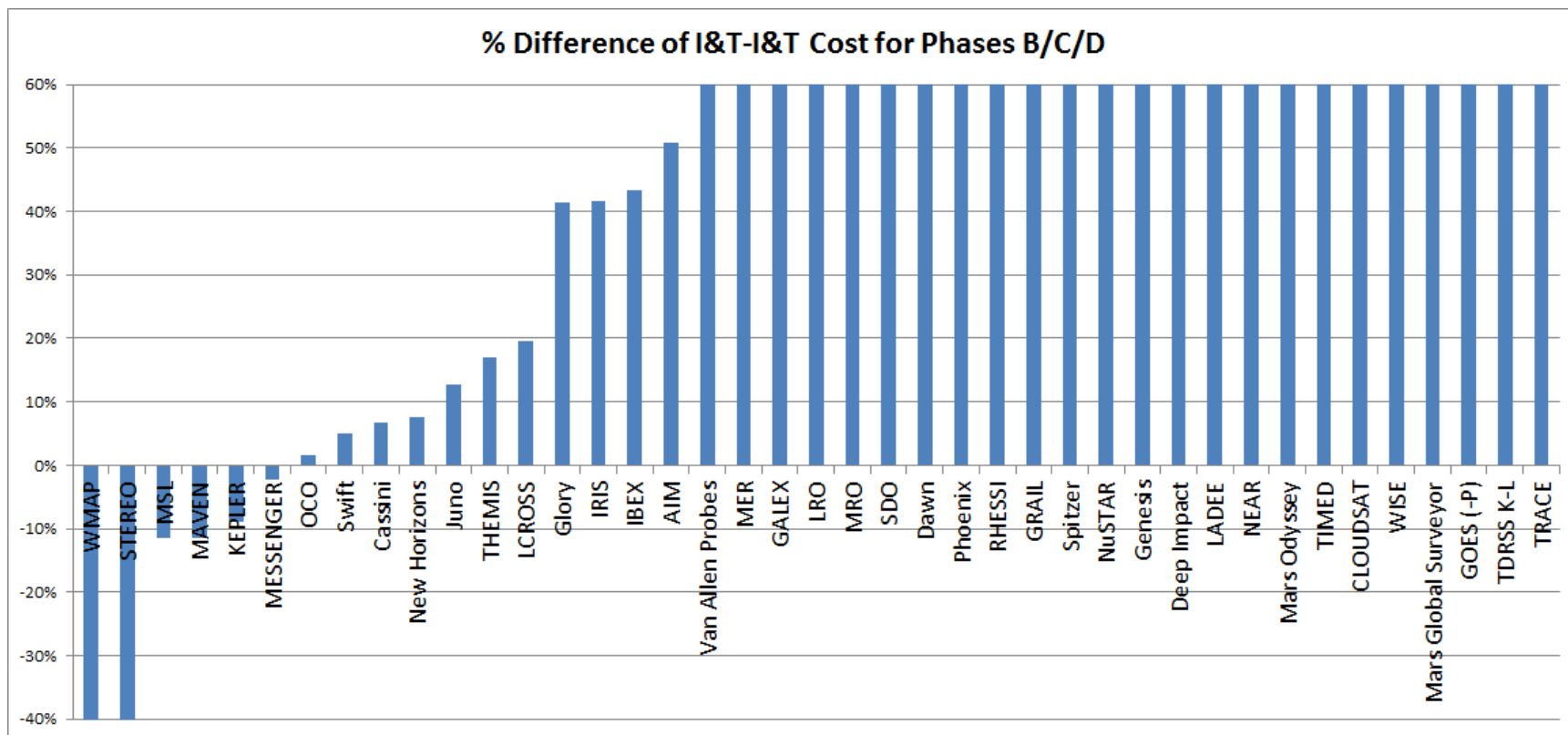


Constructive Approach Results I&T - FAB



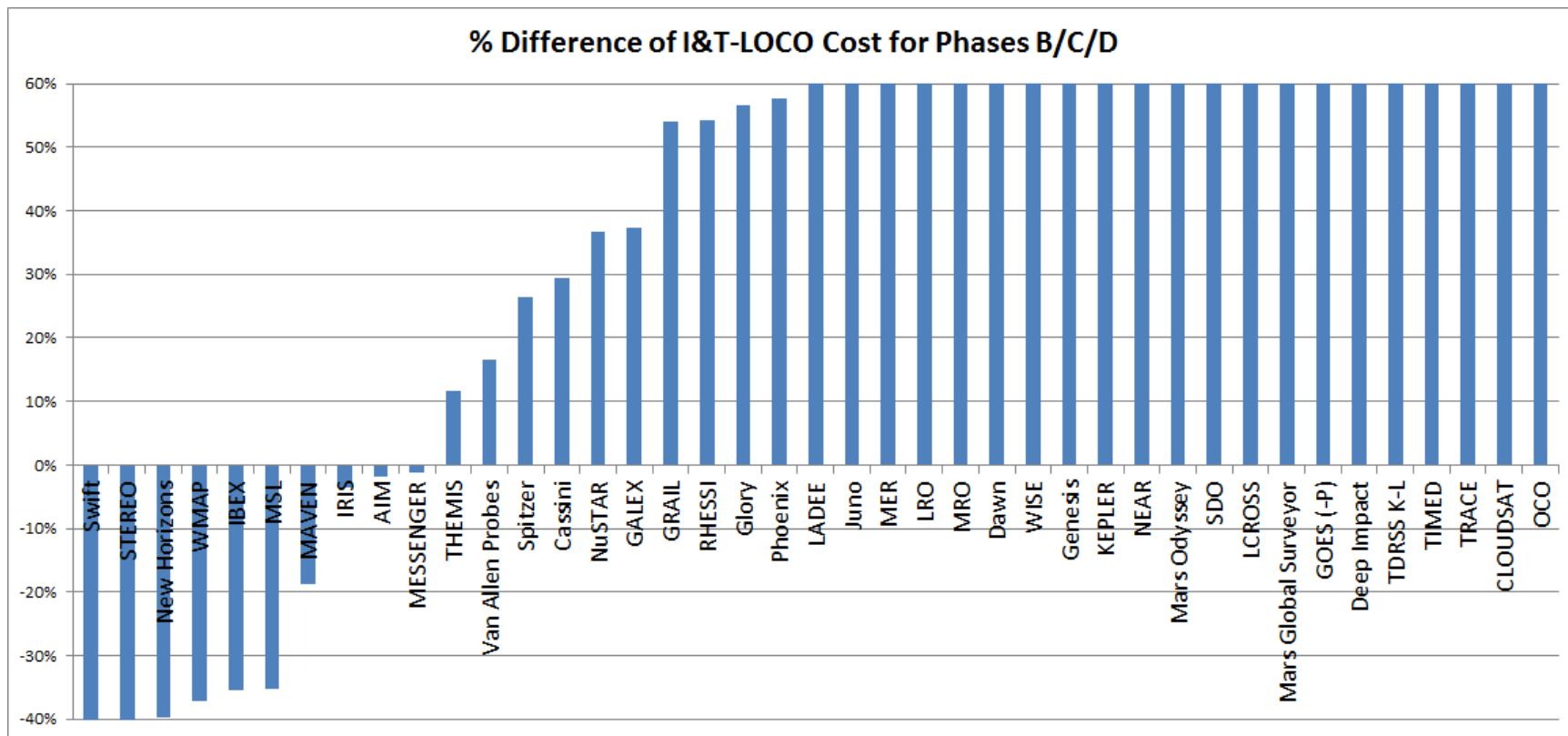


Constructive Approach Results I&T – I&T





Constructive Approach Results I&T - LOCO





Constructive Approach Results Data Ranges and Input Weightings



Engineering
Cost
Office

	DB	DB	DB	DB	DB	DB	DB	DB	DB
	LOW	LOW	LOW	LOW	HIGH	HIGH	HIGH	HIGH	HIGH
	\$K/mo	\$K/mo	\$K/mo	\$K/mo	\$K/mo	\$K/mo	\$K/mo	\$K/mo	\$K/mo
	DES	FAB	I&T	LOCO	DES	FAB	I&T	LOCO	
PM	59.1	36.7	65.4	3.3	2,005	1,650	1,150	1,122	
SE	25.0	13.3	13.2	0.2	3,358	2,525	2,248	2,768	
MA	25.3	6.4	2.1	0.2	1,573	1,765	1,163	731	
I&T	0.3	2.9	10.7	0.1	1,857	1,763	3,762	6,170	

Weightings																										
	DIRECTED or AO	MISSION RISK CLASS	MISSION DESTINA TION	FLIGHT SYSTEM TYPE	MULTIPL E FLIGHT SYSTEMS ?	LEAD ORGANIZ ATION	FLIGHT SYSTEM ORGANIZ ATION	PAYLOAD ORG.	LEAD ORGANIZ ATION EXPERIEN CE	FLIGHT SYSTEM LEAD ORG. EXPERIEN CE	PAYLOAD LEAD ORG. EXPERIEN CE	kg	W	W/kg	NUMBER OF SPACECR AFT	FLIGHT SYSTEM HERITAG E & TRL	PARTS RATING	# OF KEY SPACECR AFT CONTRA CTORS	kg	W	W/kg	PAYLOAD POWER/ MASS RATIO	# OF PAYLOAD ELEMENT S	# OF KEY PAYLOAD CONTRA CTORS	IN- HOUSE SCOPE	INTERNA TIONAL PARTICIP ATION (HW)
Input #																										
1	0.5	0.35	0.3	0.1	0	0.05	0.05		0.2	0.2	0.2	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1
2	1.2	0.5	0.6	0.3	1	0.1	0.1		0.4	0.4	0.4	1	1	1	1	1	1	1	1	1	1	1	1	1	1	1
3		0.75	0.6	0.4		0.6	0.6		0.6	0.6	0.6															0.6
4		1.8	0.7	0.6		1.2	1.3		0.8	0.8	0.8															0.8
5			0.9	0.8		1.8	1.2		1	1	1															1
6			1.747	1.5																						
7			2																							

Correlations																										
	DIRECTED or AO	MISSION RISK CLASS	MISSION DESTINA TION	FLIGHT SYSTEM TYPE	MULTIPL E FLIGHT SYSTEMS ?	LEAD ORGANIZ ATION	FLIGHT SYSTEM ORGANIZ ATION	PAYLOAD ORG.	LEAD ORGANIZ ATION EXPERIEN CE	FLIGHT SYSTEM LEAD ORG. EXPERIEN CE	PAYLOAD LEAD ORG. EXPERIEN CE	kg	W	W/kg	NUMBER OF SPACECR AFT	FLIGHT SYSTEM HERITAG E & TRL	PARTS RATING	# OF KEY SPACECR AFT CONTRA CTORS	kg	W	W/kg	PAYLOAD POWER/ MASS RATIO	# OF PAYLOAD ELEMENT S	# OF KEY PAYLOAD CONTRA CTORS	IN- HOUSE SCOPE	INTERNA TIONAL PARTICIP ATION (HW)
PM	DES	0	1.05	0.2625	0	0.525	0	0.525	0.7875	0.7875	0.525	1.3125	0.7875	1.3125	0	0	0	1.05	0	1.05	1.3125	0	0	0	0	0
	FAB	0	0.525	0.2625	0	0.525	0	0.7875	0.525	0.7875	0.525	0.7875	0.7875	1.05	0.525	0	0	1.05	0.525	0.525	0.525	0	0	0	0	0
	I&T	0	0	0.2625	0	0.525	0	0.7875	0	0	0	0	0.7875	0.525	0	0	0.7875	0.525	0	0	0	0	0	0	0	0
	LOCO	0	0	0.525	0.525	0.525	0.525	0.525	0	0	0	0.525	1.05	0.525	0	0	0	0.525	1.05	0	0	0.525	0	0	0	0
SE	DES	0.4	0.8	0.2	0	0.4	0	0.6	0.6	0.6	0.4	1	0.4	0.6	0	0	0	0.6	0	0	0.6	0.4	0	0	0	0
	FAB	0.4	0.6	0.6	0.8	0.4	0.64	0.8	0.4	0.4	0	0.4	0.8	0.4	0	0	0	0.8	0.6	0	0	0.4	0	0	0	0
	I&T	0	0.4	0.4	0.8	0.4	0.64	0.6	0	0.4	0	0	0.8	0	0	0	0	0.8	0.6	0	0	0	0	0	0	0
	LOCO	0	0.4	0.4	1	0.4	0.8	0.6	0	0.4	0	0	1.2	0.4	0	0	0	0.8	0.6	0	0.4	0.8	0	0	0	0
MA	DES	0	1.25	0.5	0	0.5	0	0.5	0.5	0	0	1	1	0.75	0	0	0	1.25	0	0.5	0.75	0	0	0	0.5	0
	FAB	0	0.75	1	0	0.5	0	0.5	0	0	0	0.75	1.5	0.5	0	0	0.5	1.5	0.5	0	0.5	0	0.75	0.5	0.5	0
	I&T	0	0	0.75	0	0.5	0	0.5	0	0	0	0	0.5	0.5	0	0	0.5	1	0	0	0	0	0	0.5	0	0
	LOCO	0	0.5	0.5	0	0.5	0	0.5	0	0	0	0	0.5	0.75	0.5	0	0.5	0.75	0.5	0	0	0	0	0	0	0
I&T	DES	0	0.75	0.25	0	0.5	0	0	0.5	0.5	0.5	1.25	0.5	0.75	0	0	0	0.75	0	0.75	1.25	0	0	0	0	0
	FAB	0	0	0.5	0.5	0.5	0.75	0	0.5	0	0.5	1.25	0.5	0	0	0	0	1	0.5	0.75	1	0.5	0.5	0	0	0
	I&T	0	0	0.75	1	0.5	1	0.5	0	0	0	1.5	0	0	0	0	0	0.75	0.75	0	0.5	0.75	0.75	0	0	0
	LOCO	0	0	0.5	0.75	0.5	0.75	0	0	0	0	1.5	0	0	0	0	0	0.5	0.75	0	1	1	0.75	0	0	0





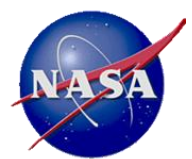
BACKUP



**Engineering
Cost
Office**

STATISTICAL APPROACH DETAILS – Standard Regression Approach



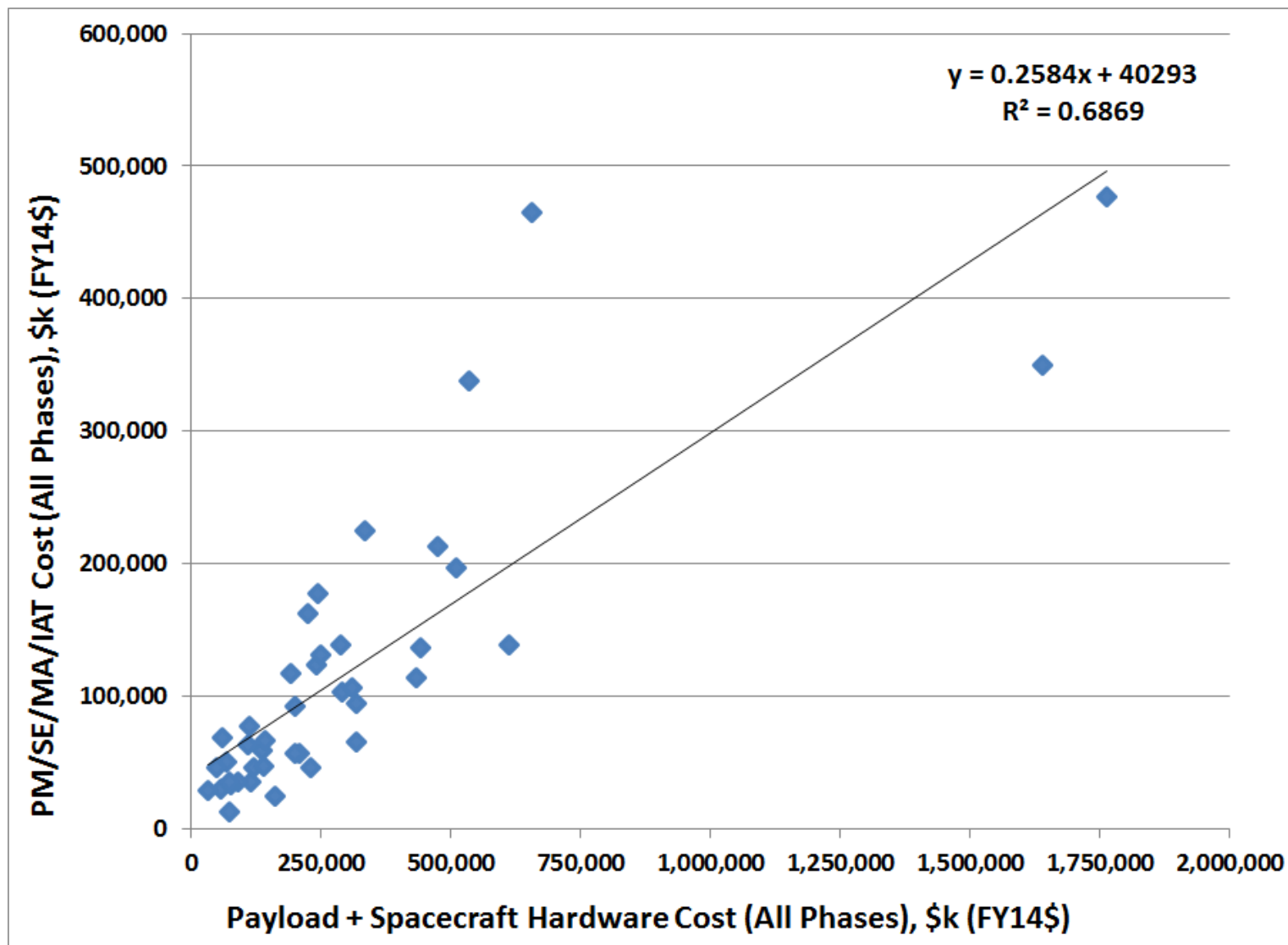


Standard Regression Approach Results

Total PM/SE/MA/I&T for Phases B/C/D



Engineering
Cost
Office



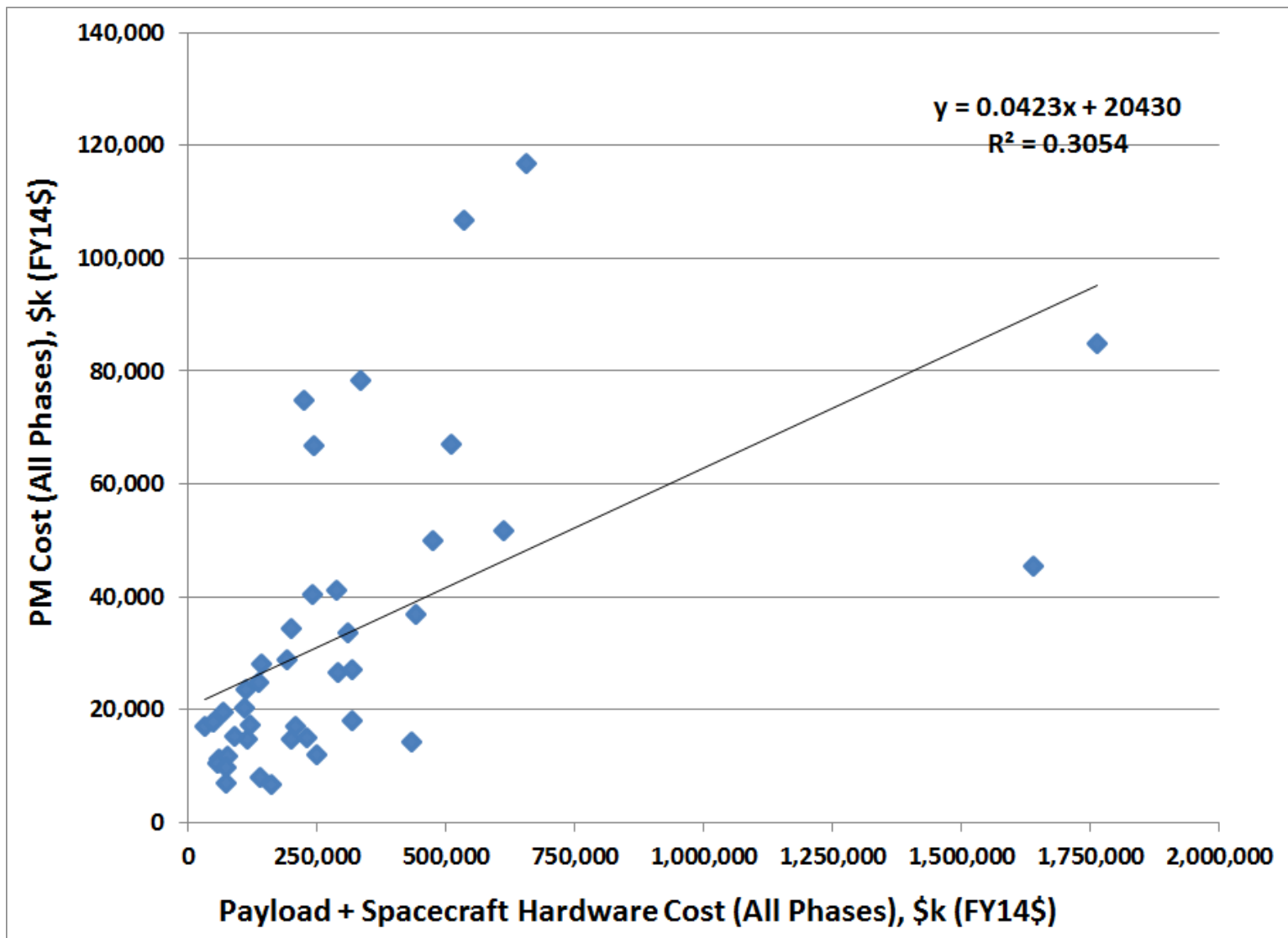


Standard Regression Approach Results

Total PM for Phases B/C/D



Engineering
Cost
Office



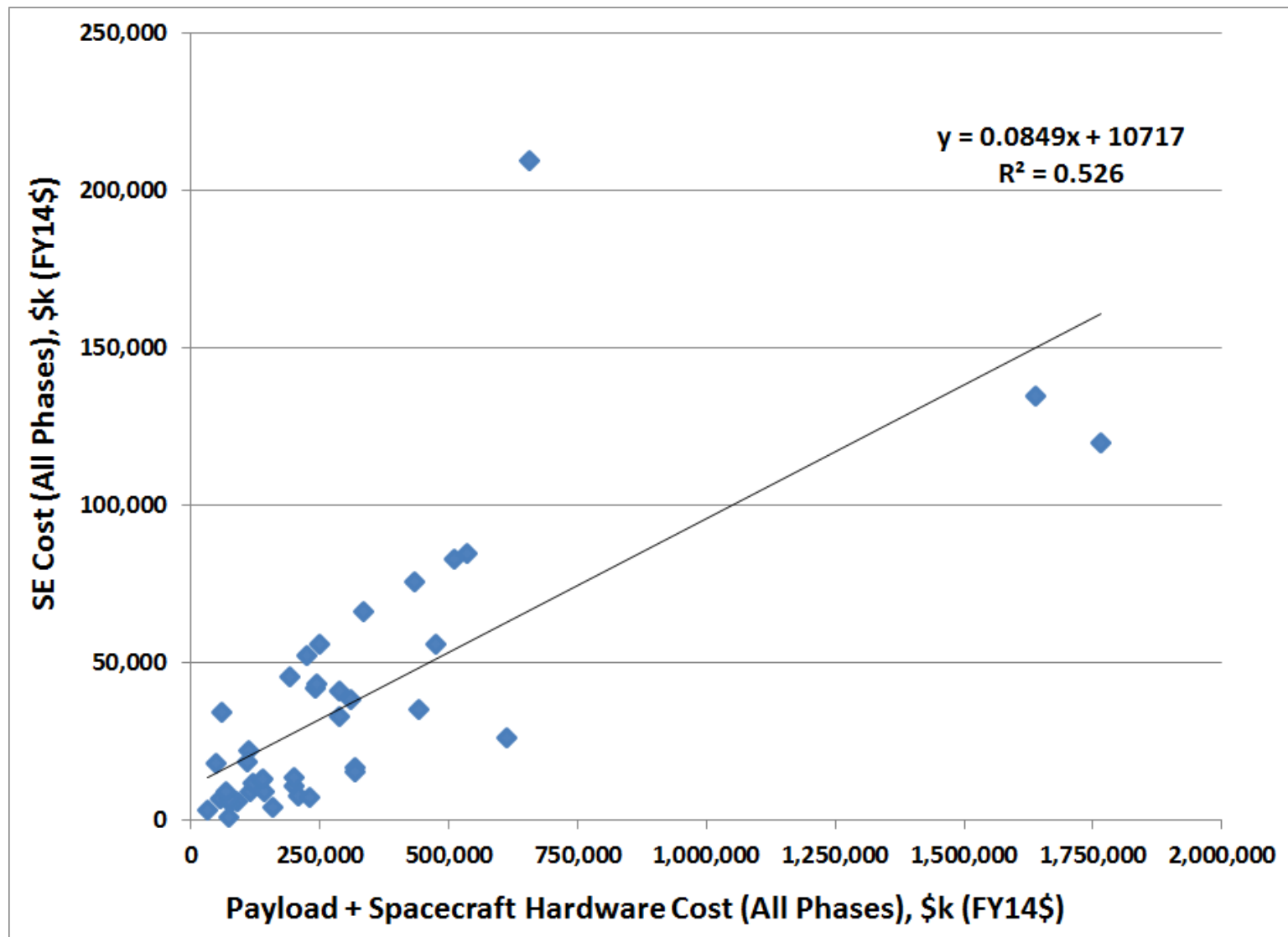


Standard Regression Approach Results

Total SE for Phases B/C/D



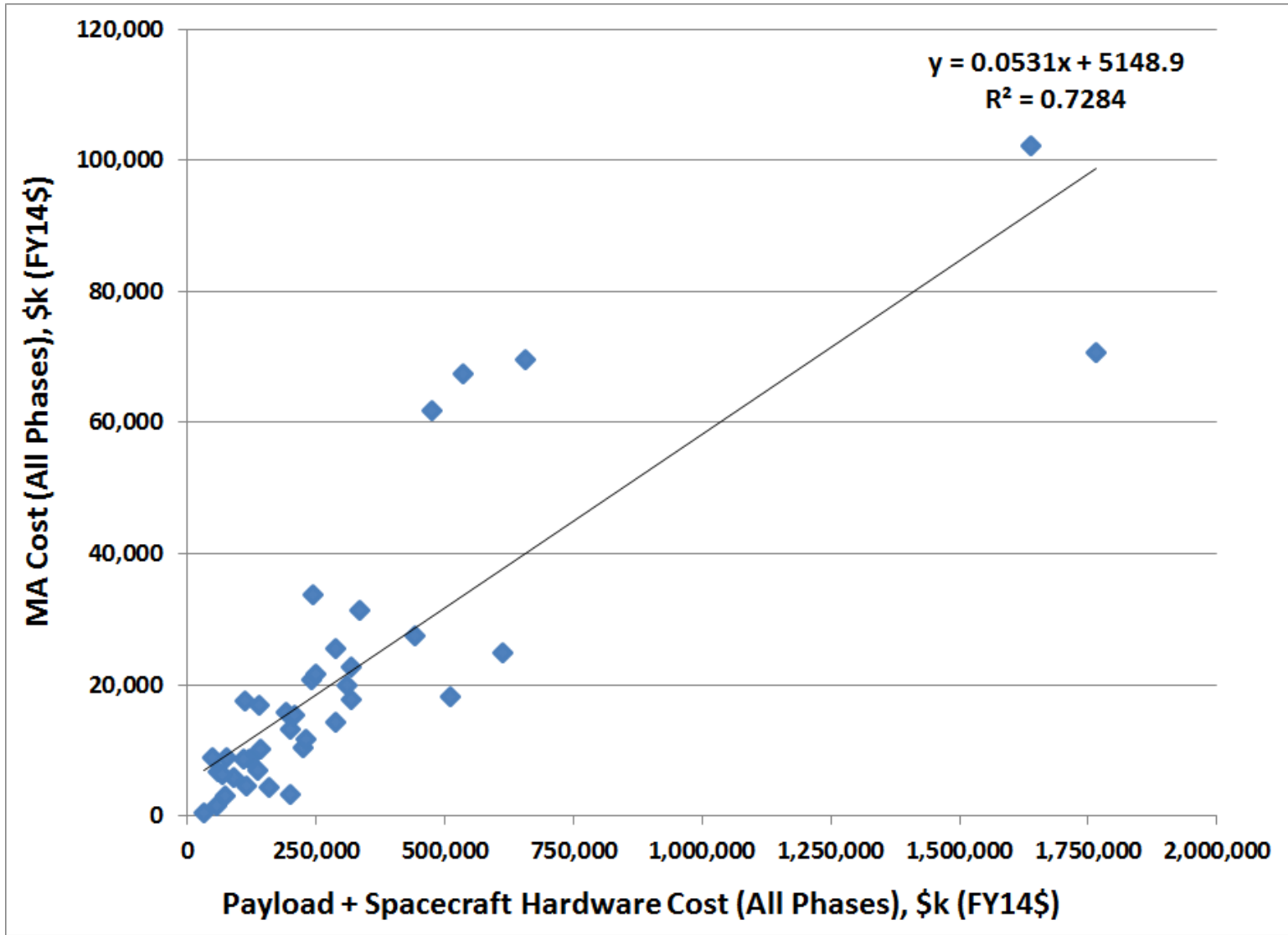
Engineering
Cost
Office





Standard Regression Approach Results

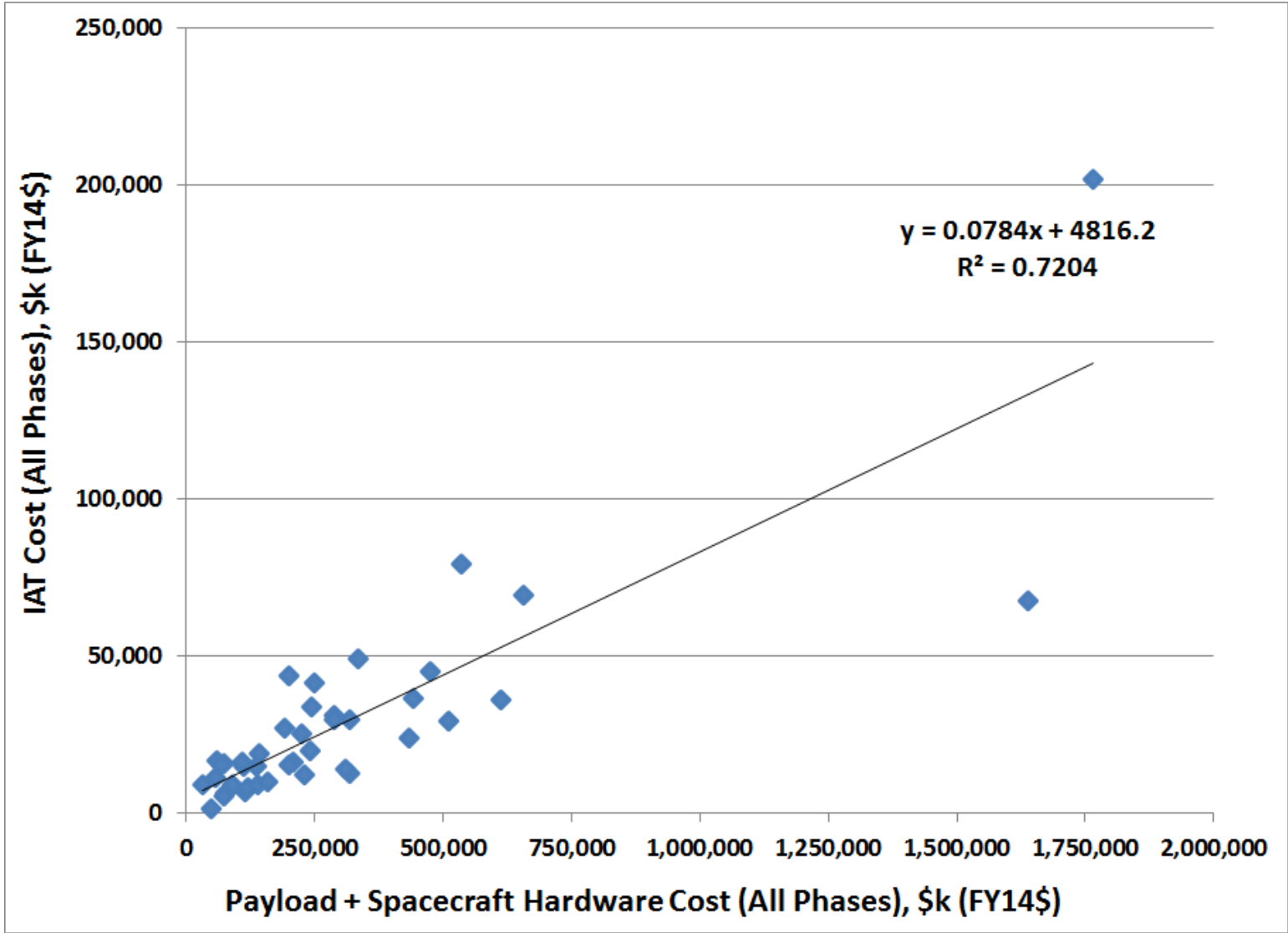
Total MA for Phases B/C/D





Standard Regression Approach Results

Total I&T for Phases B/C/D

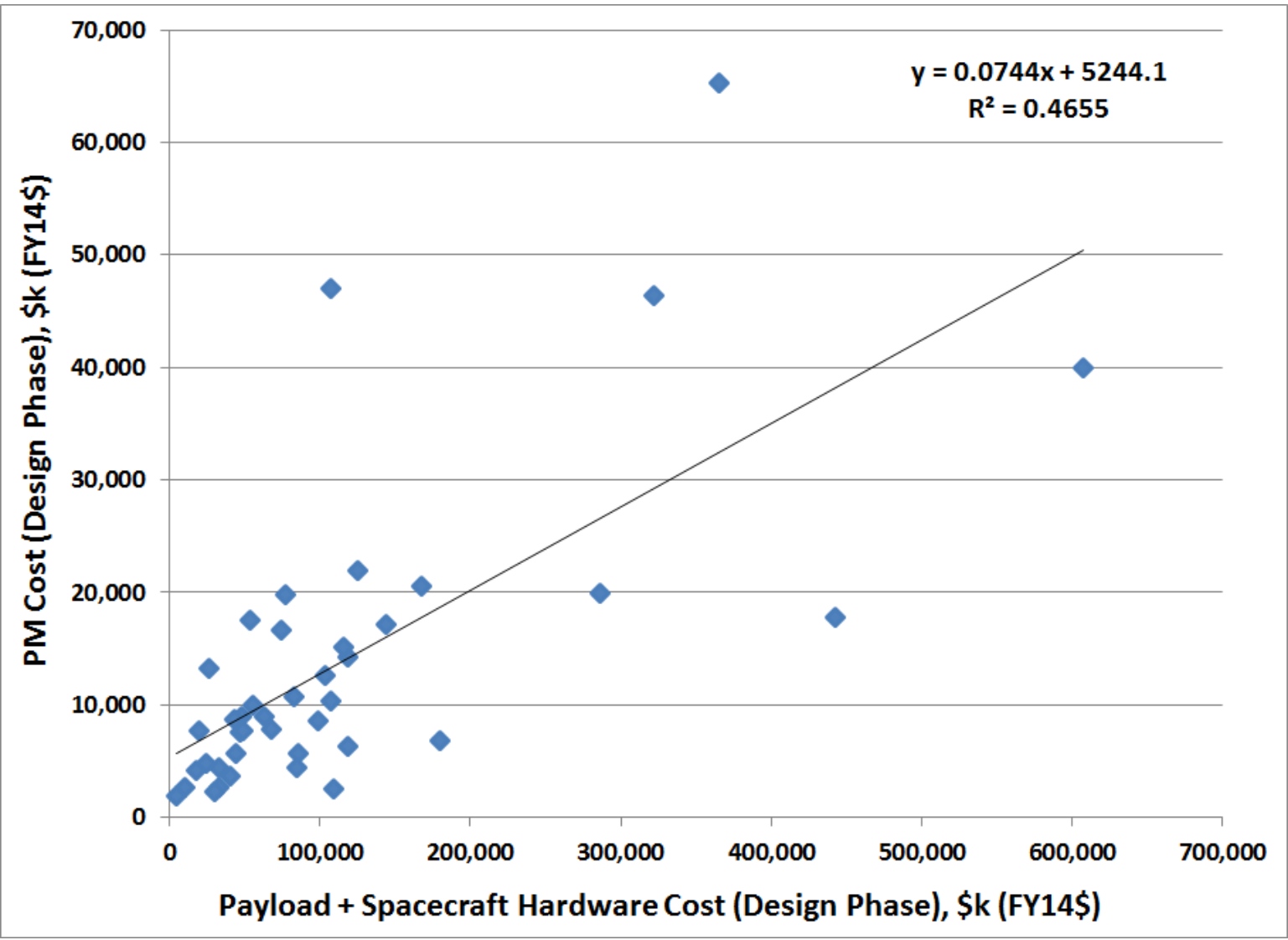




Standard Regression Approach Results – PM for the Design Phase



Engineering
Cost
Office

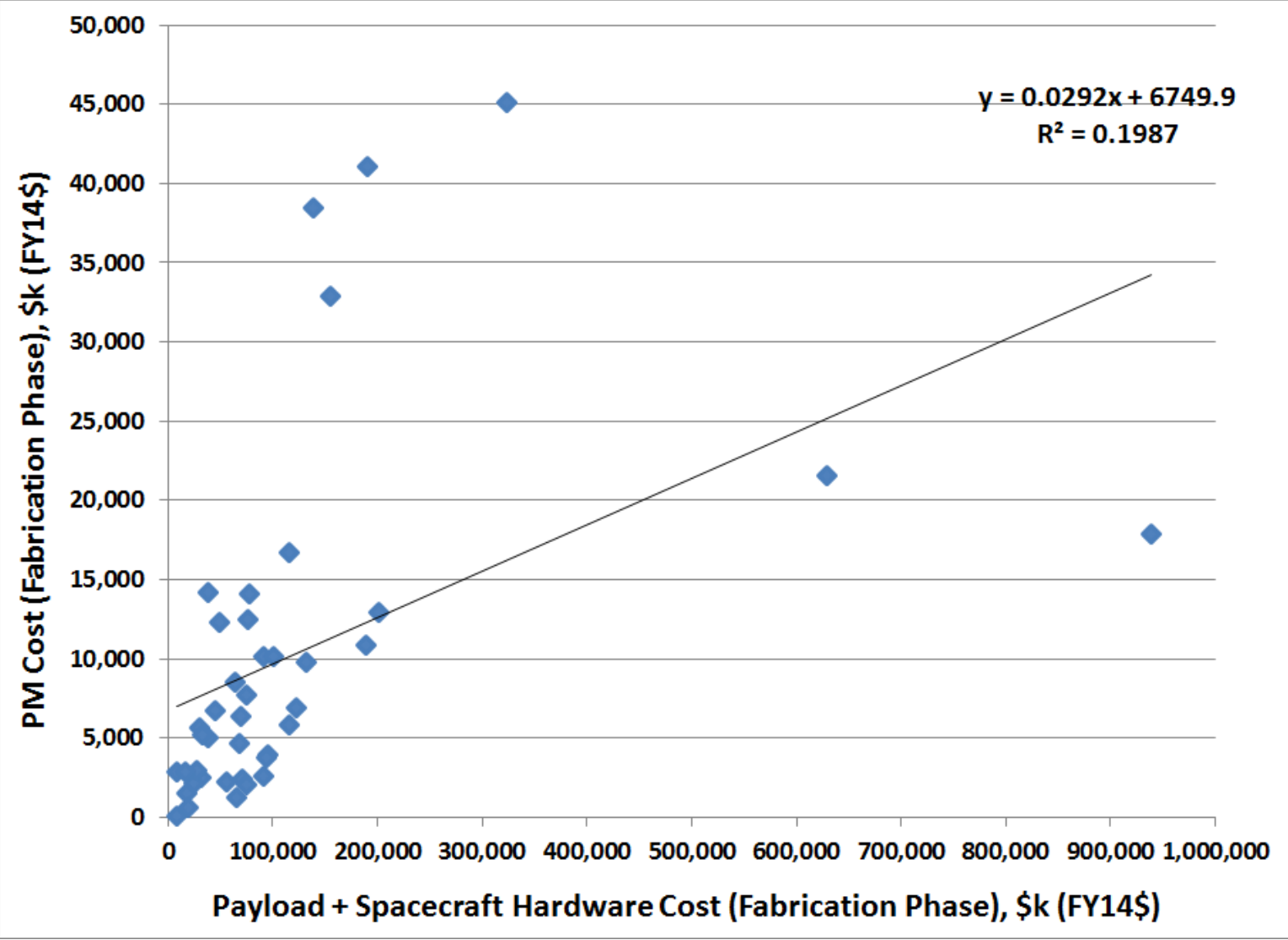




Standard Regression Approach Results – PM for the Fabrication Phase



Engineering
Cost
Office

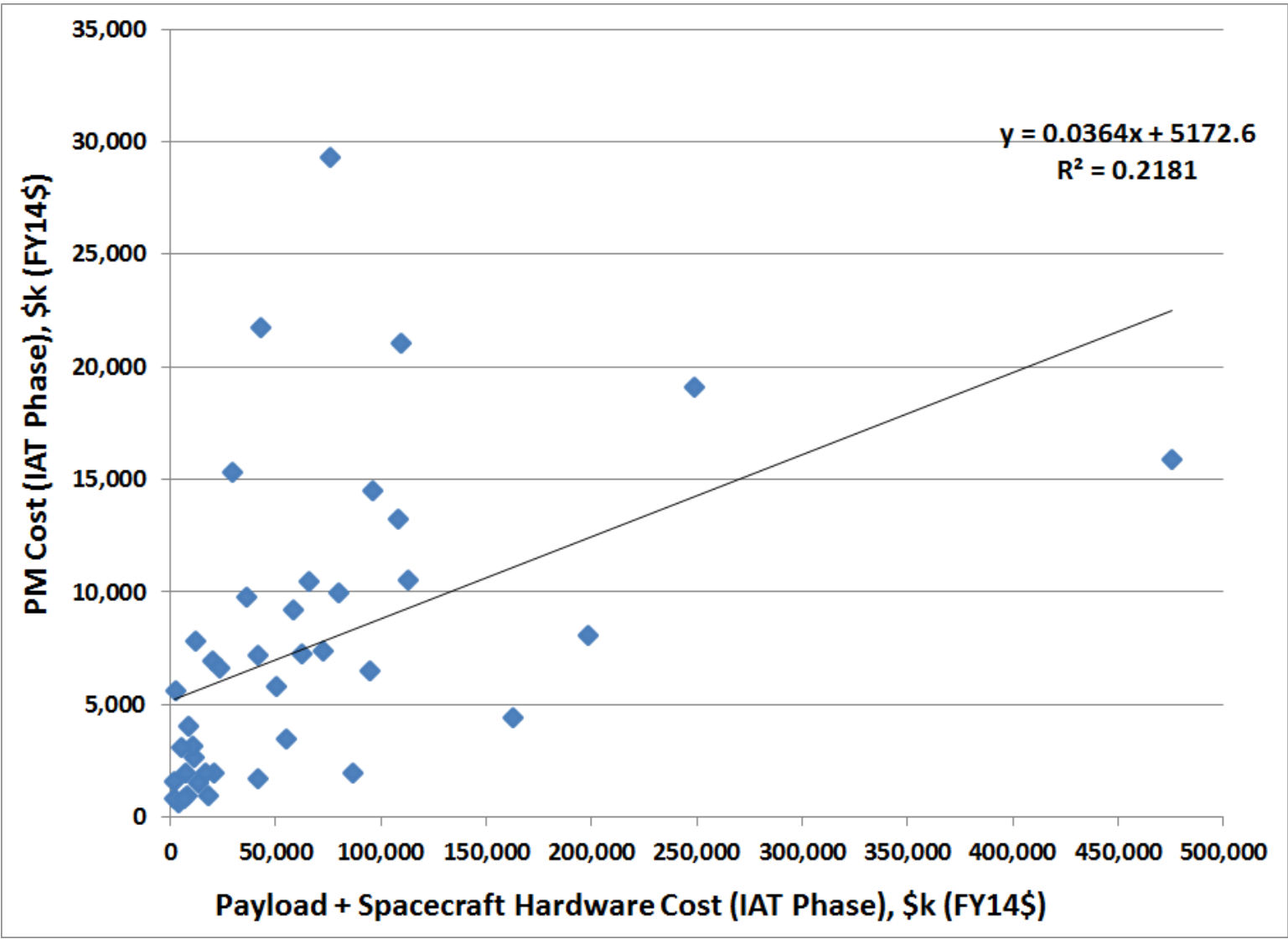


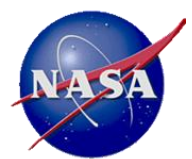


Standard Regression Approach Results – PM for the I&T Phase



Engineering
Cost
Office

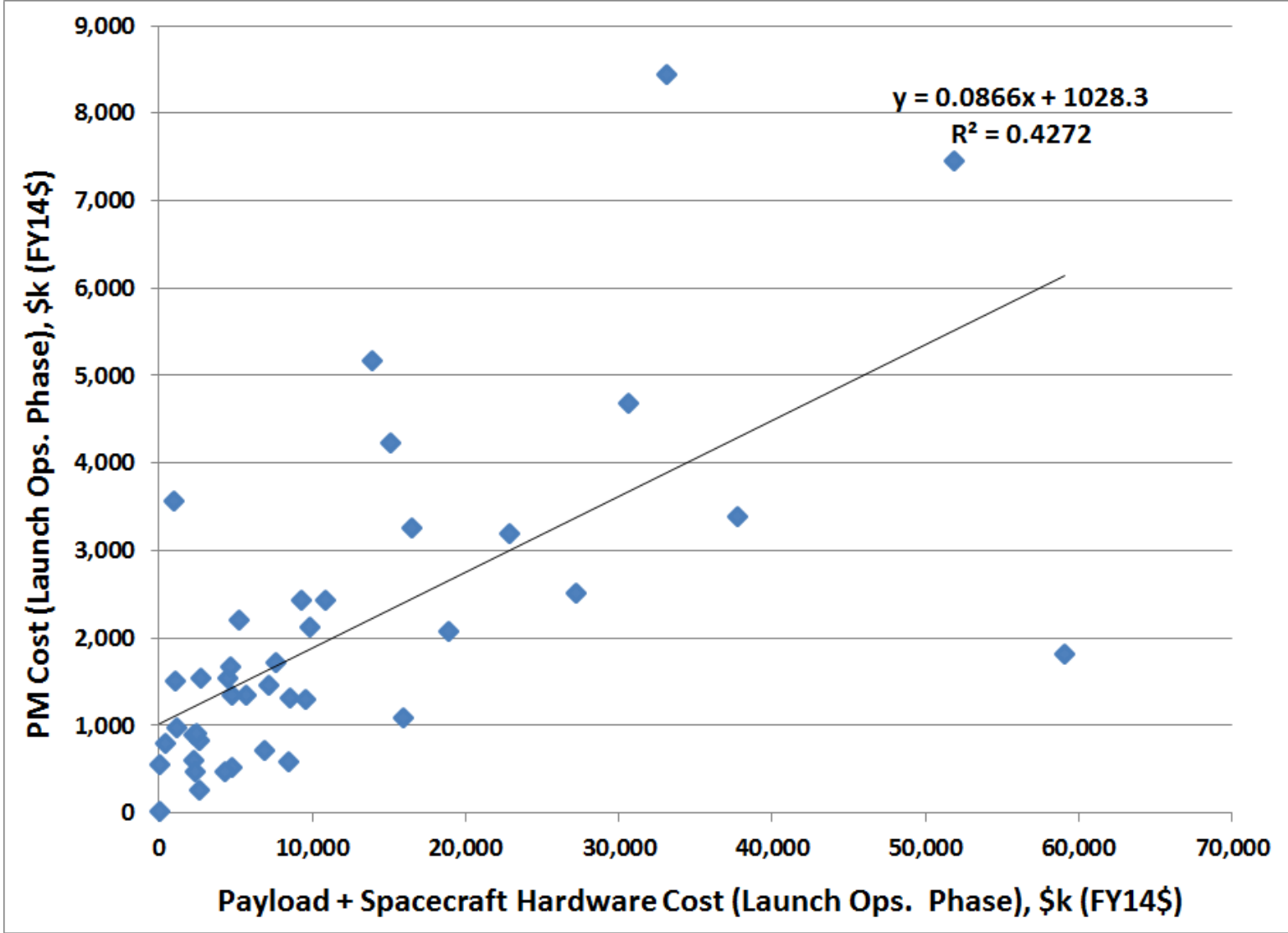




Standard Regression Approach Results – PM for the LOCO Phase



Engineering
Cost
Office

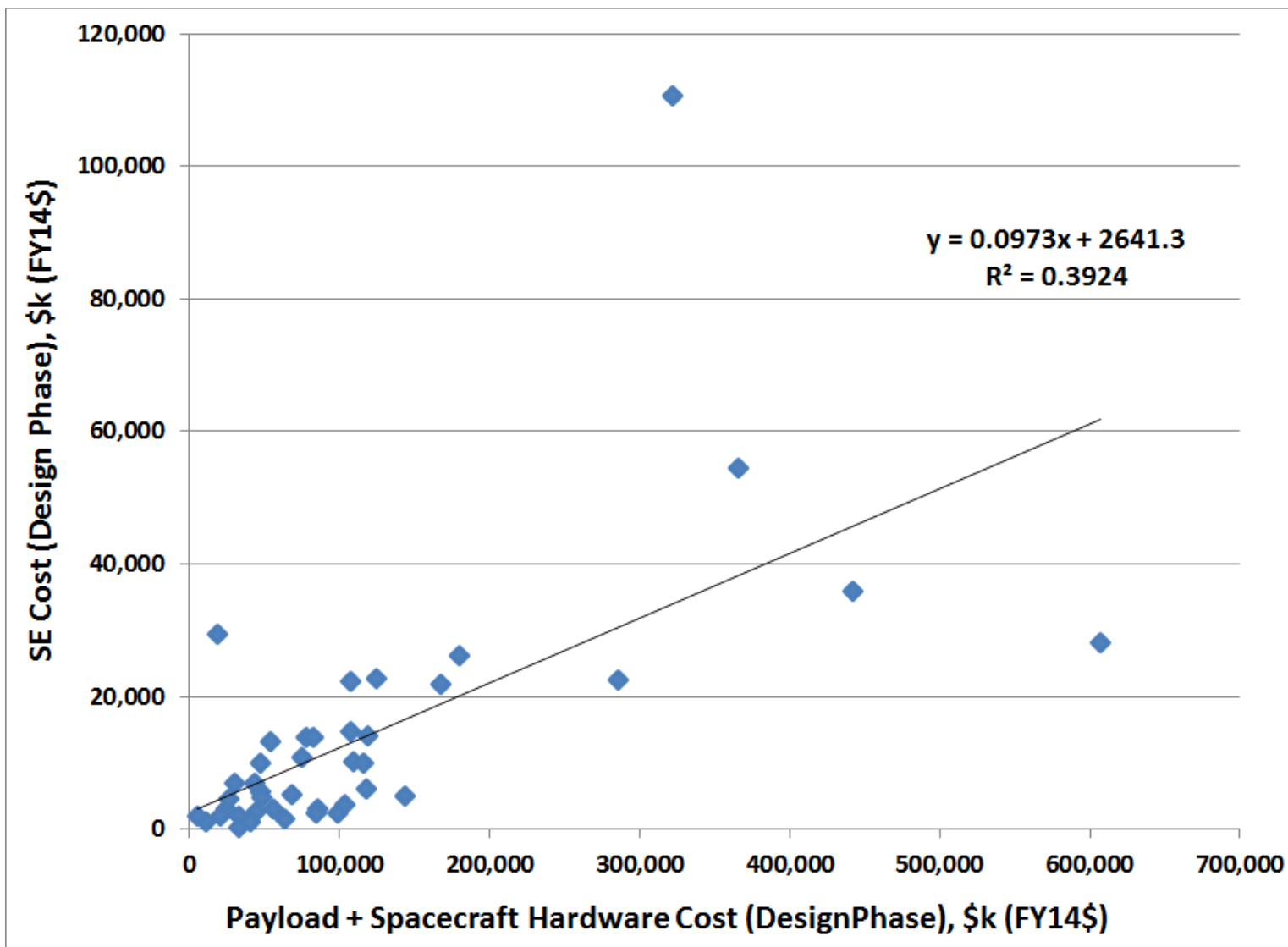




Standard Regression Approach Results – SE for the Design Phase



Engineering
Cost
Office

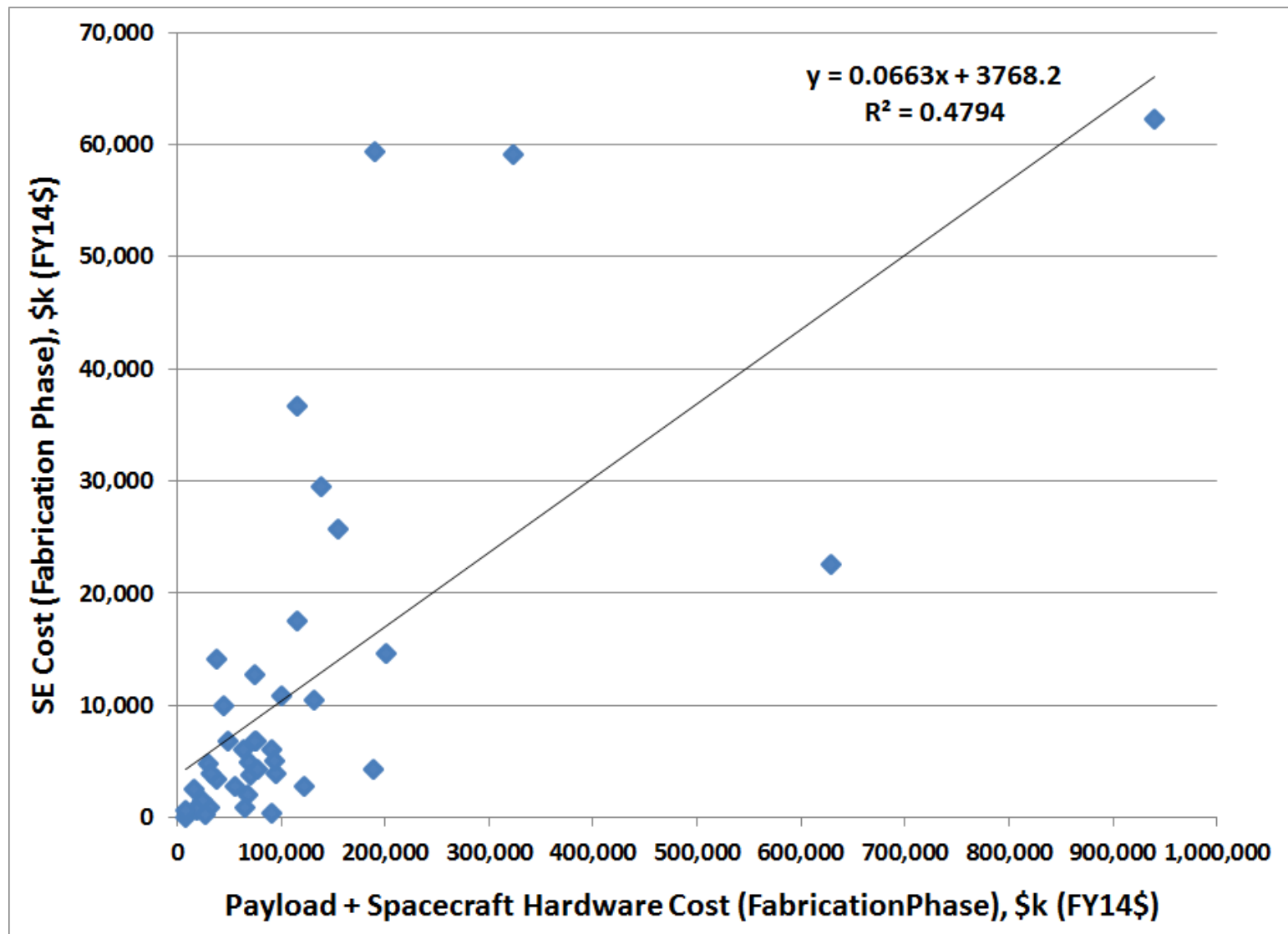




Standard Regression Approach Results – SE for the Fabrication Phase



Engineering
Cost
Office

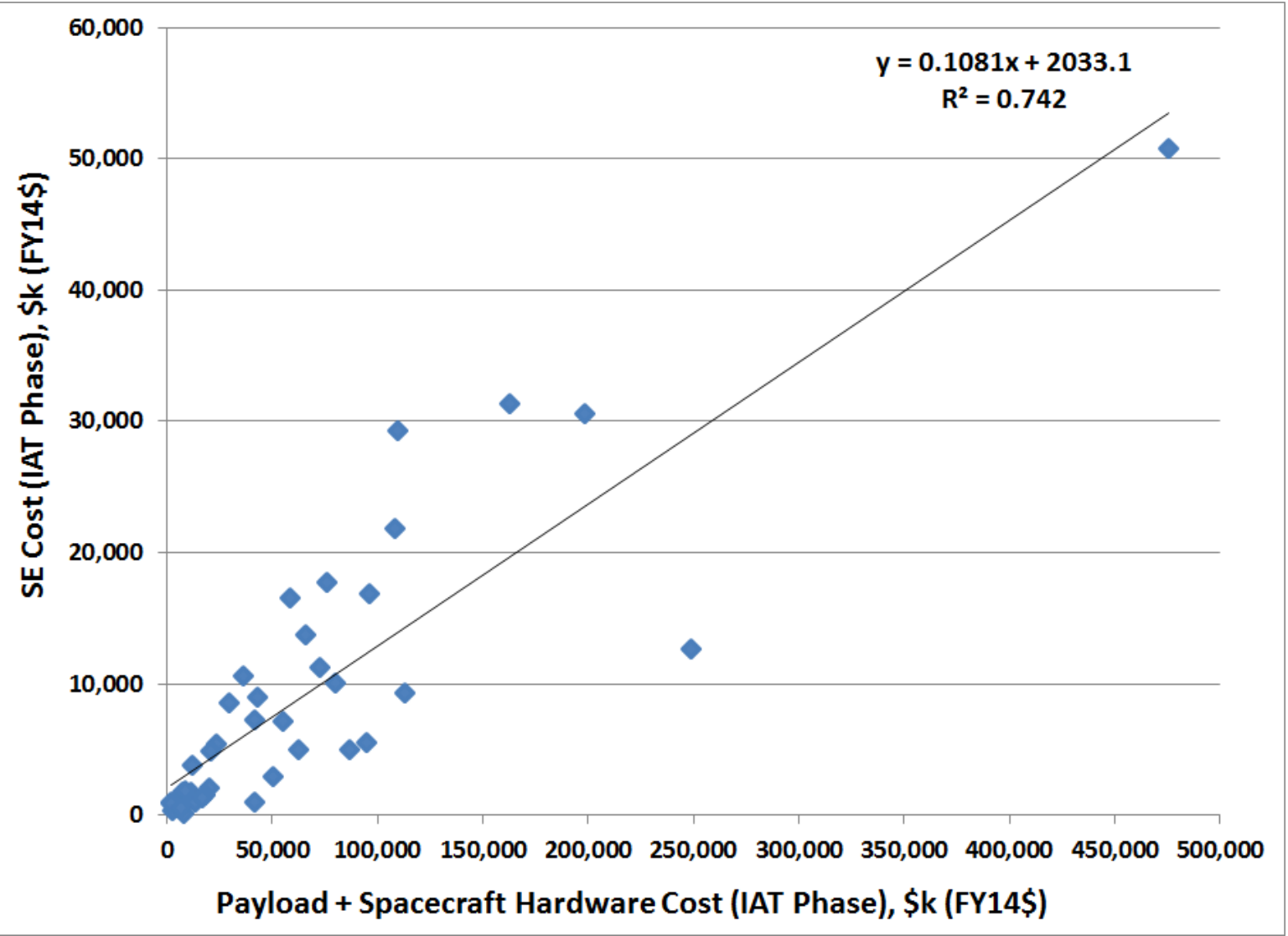




Standard Regression Approach Results – SE for the I&T Phase



Engineering
Cost
Office

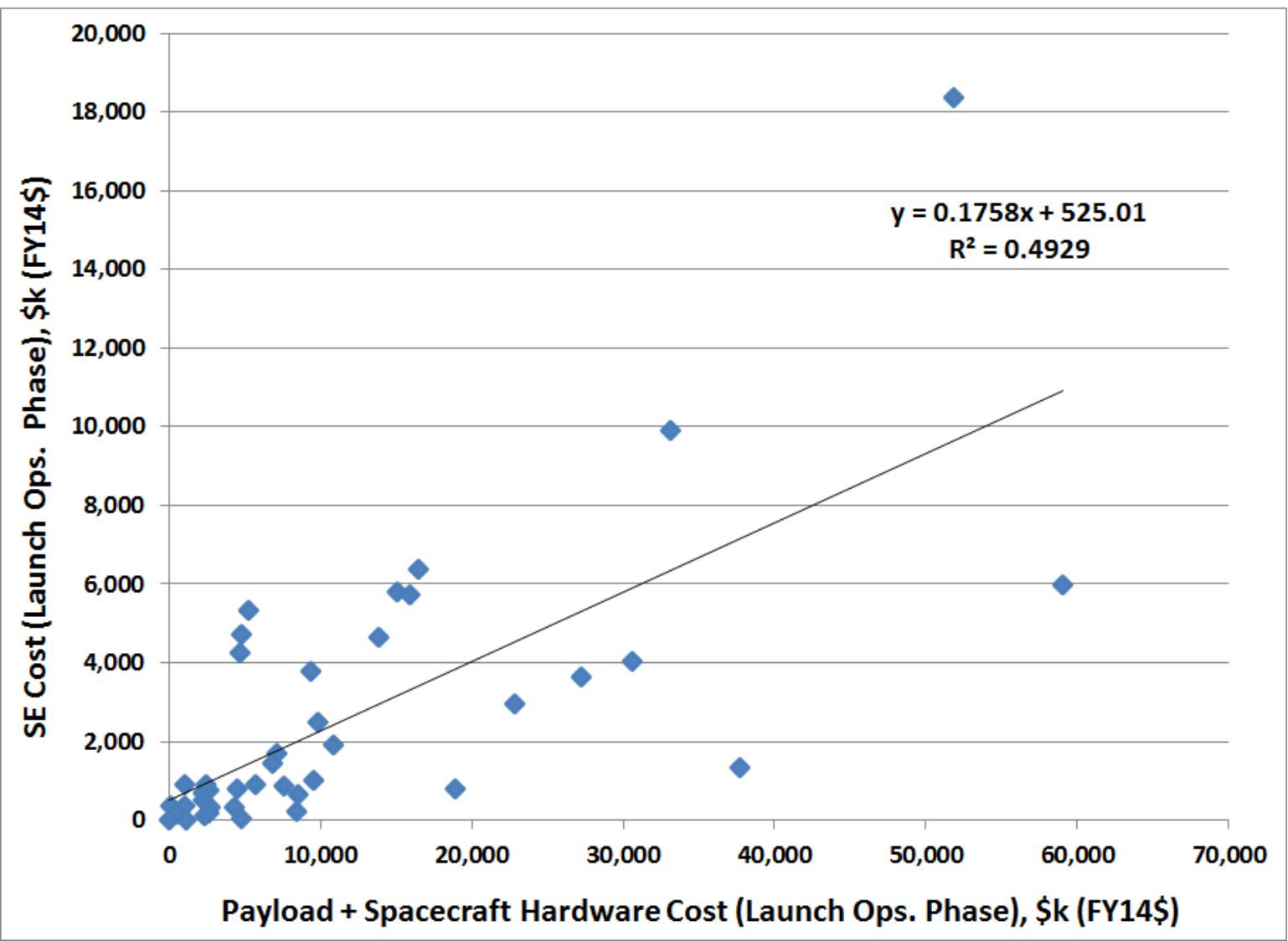




Standard Regression Approach Results – SE for the LOCO Phase



Engineering
Cost
Office

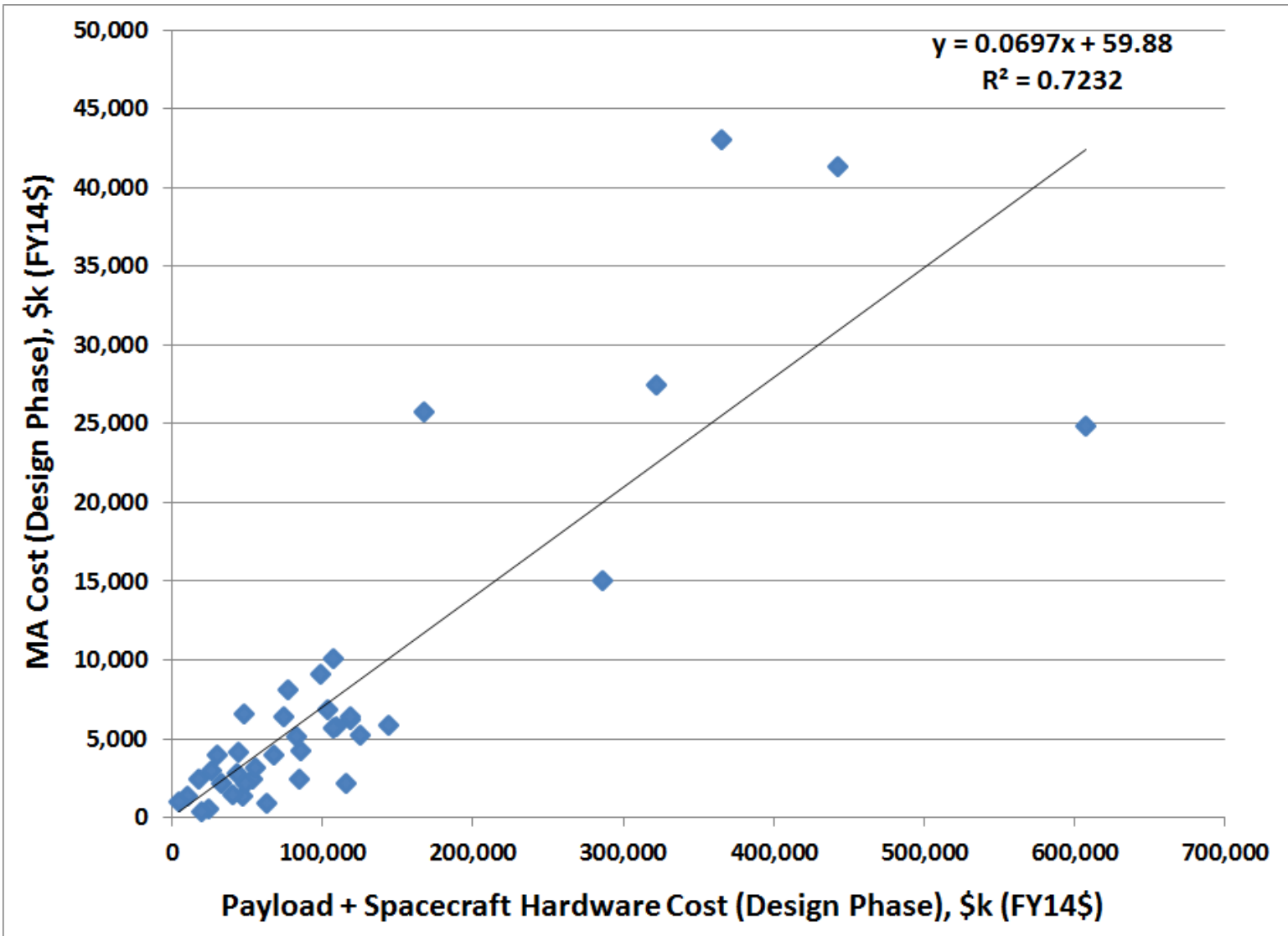




Standard Regression Approach Results – MA for the Design Phase



Engineering
Cost
Office

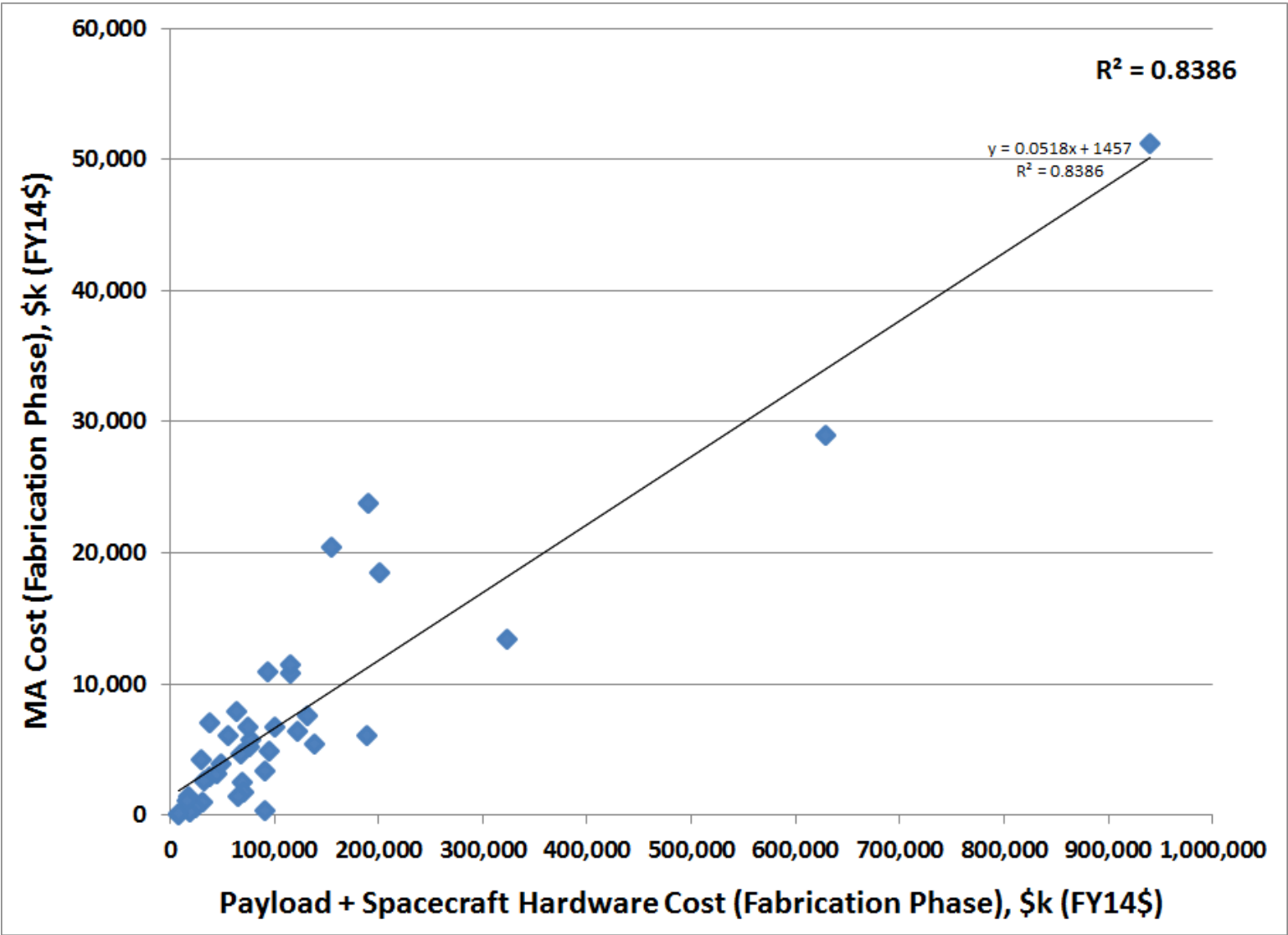




Standard Regression Approach Results – MA for the Fabrication Phase



Engineering
Cost
Office

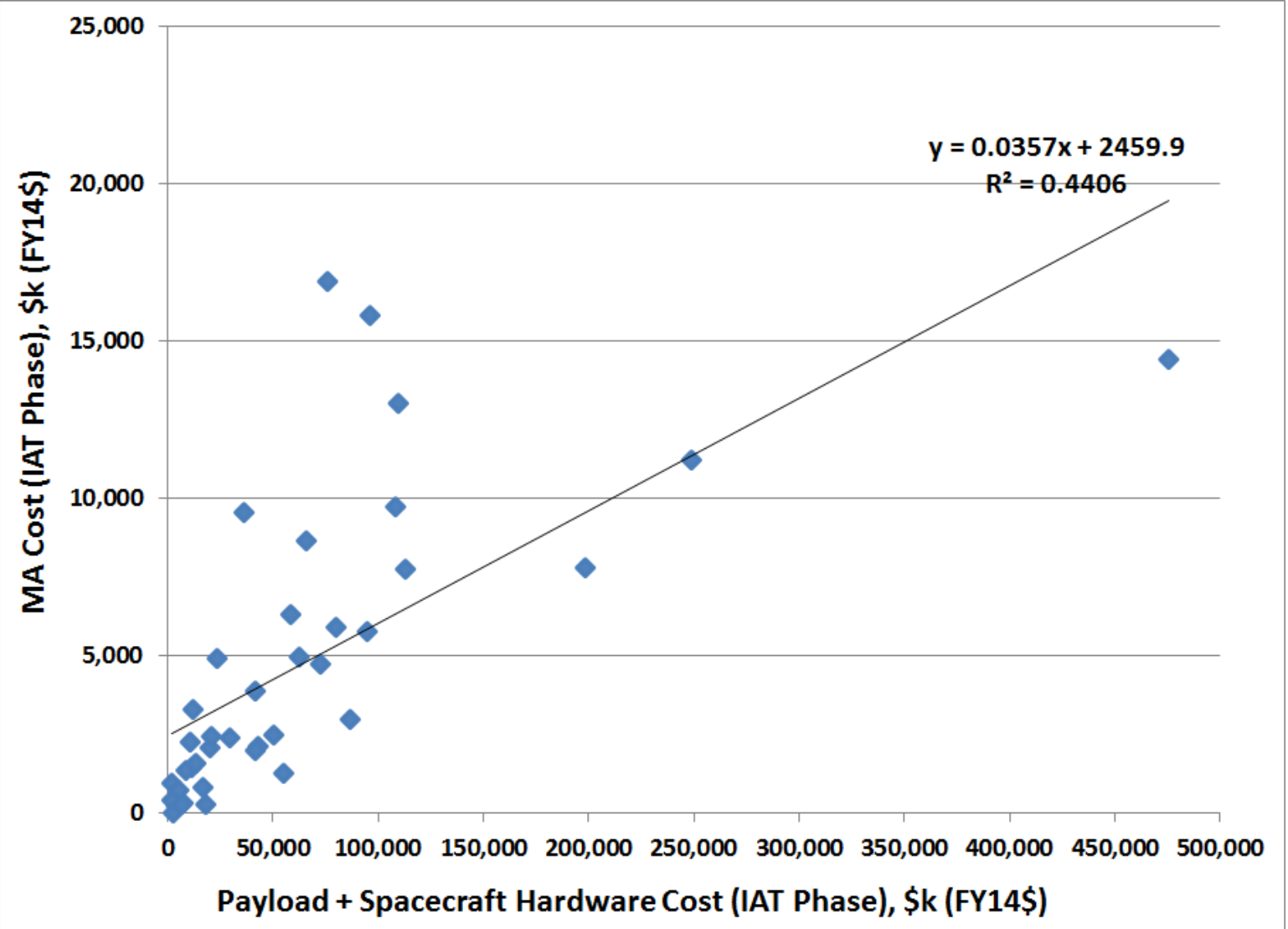


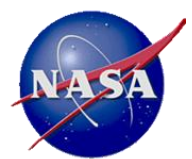


Standard Regression Approach Results – MA for the I&T Phase



Engineering
Cost
Office

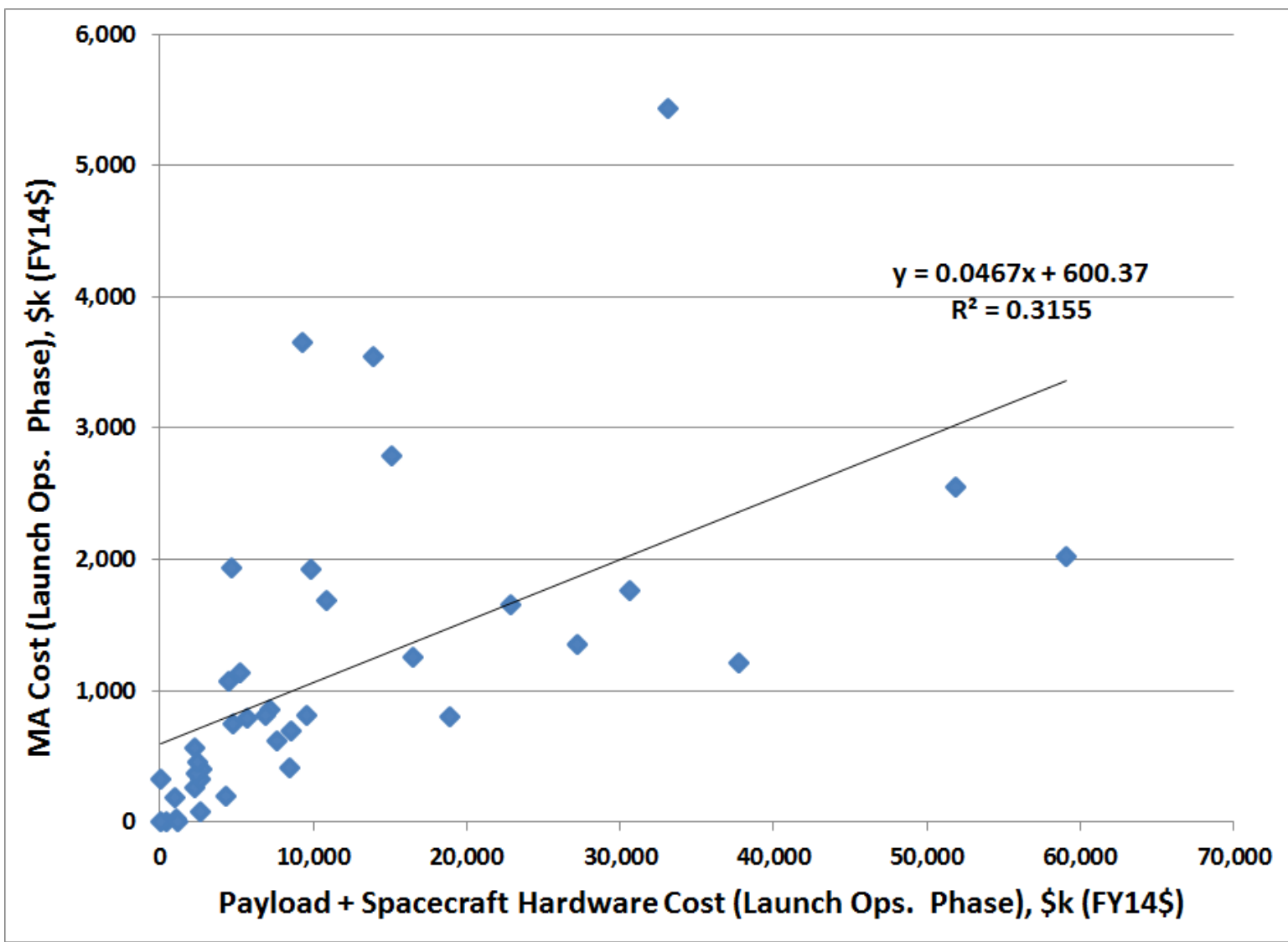




Standard Regression Approach Results – MA for the LOCO Phase



Engineering
Cost
Office

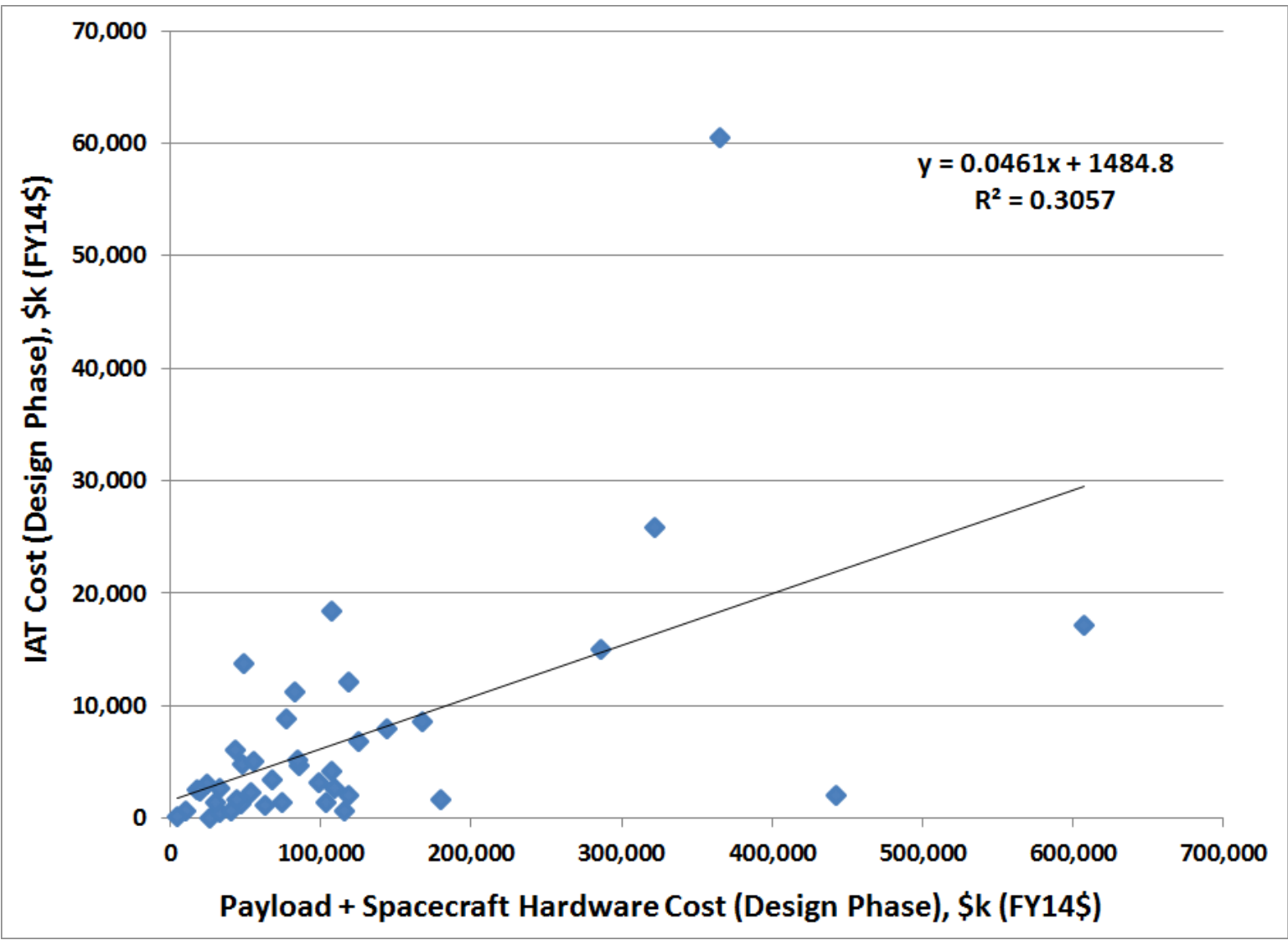




Standard Regression Approach Results – I&T for the Design Phase



Engineering
Cost
Office

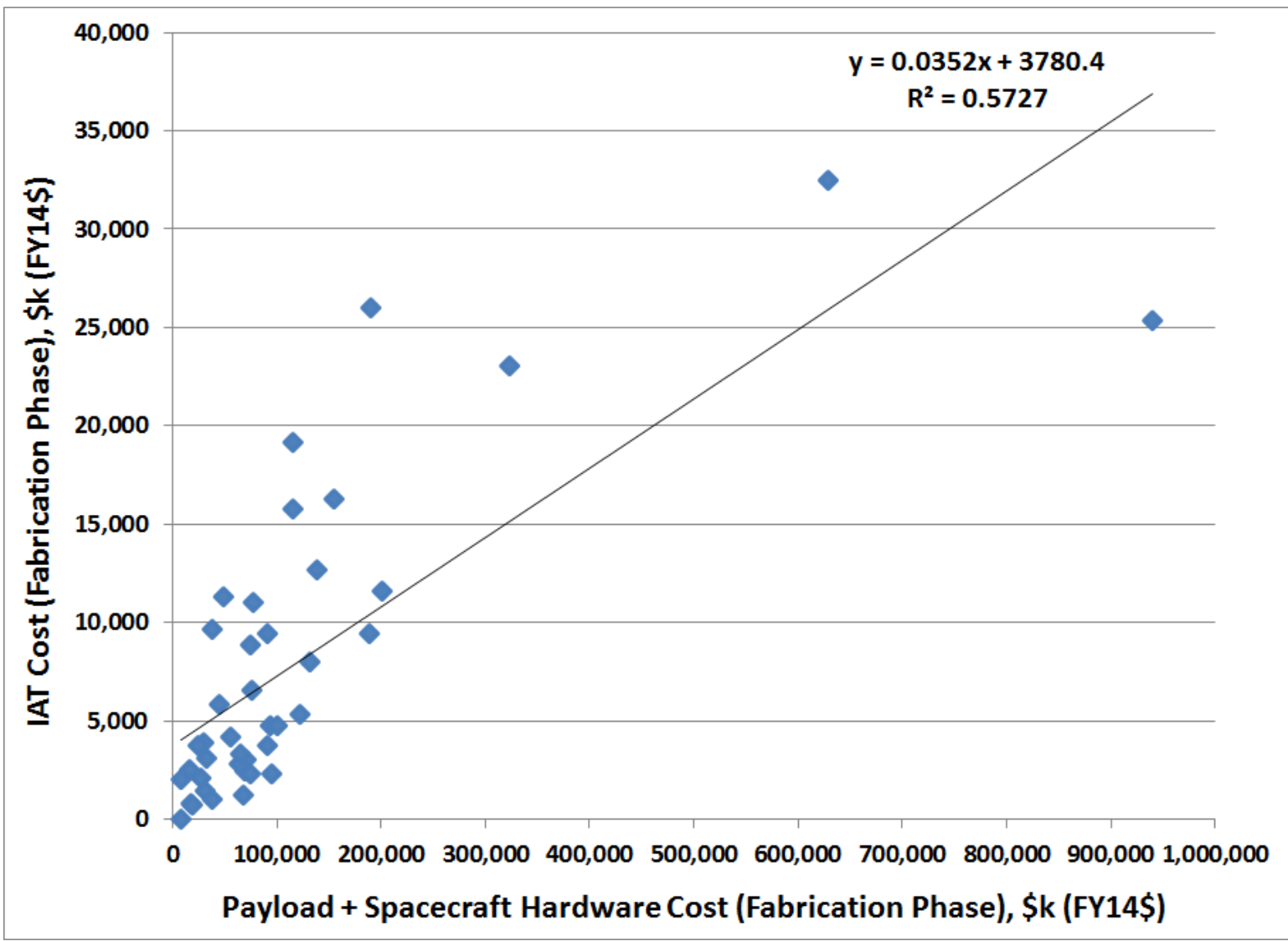




Standard Regression Approach Results – I&T for the Fabrication Phase



Engineering
Cost
Office

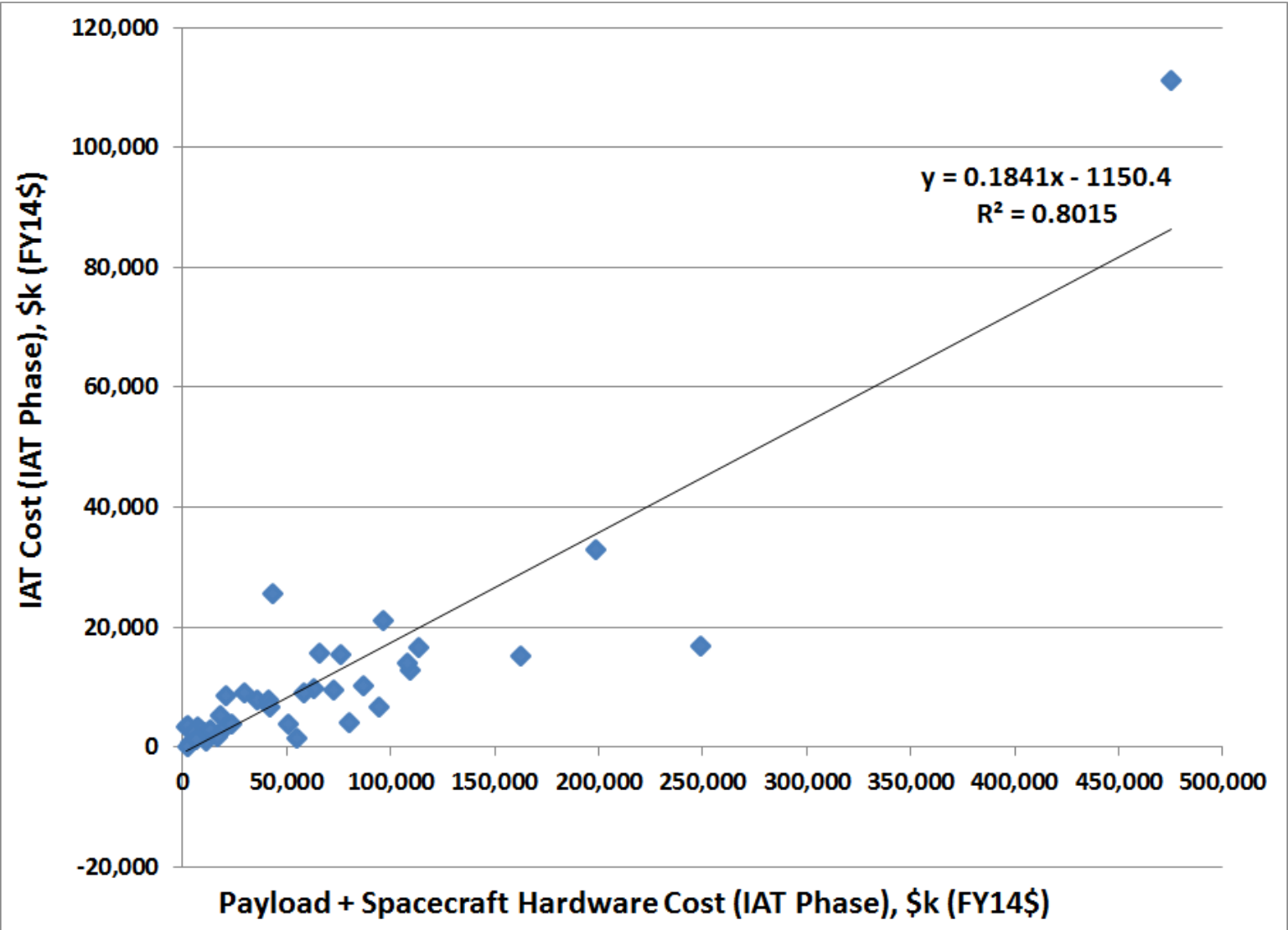




Standard Regression Approach Results – I&T for the I&T Phase



Engineering
Cost
Office

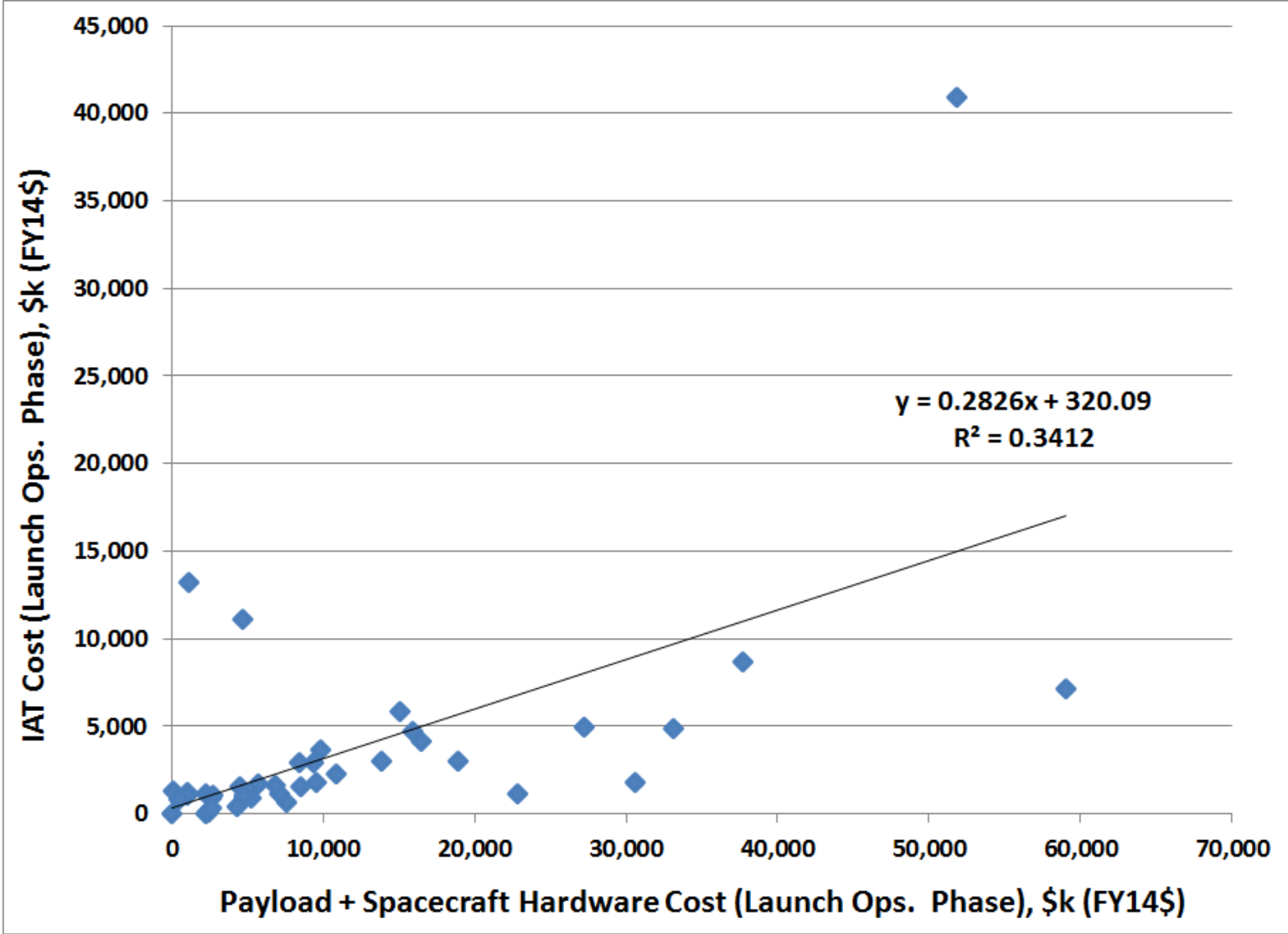




Standard Regression Approach Results – I&T for the LOCO Phase



Engineering
Cost
Office





BACKUP



**Engineering
Cost
Office**

STATISTICAL APPROACH DETAILS – PCA Approach





Data Set Correlation Matrix



Engineering
Cost
Office

	LN \$k-FY14 Per Month	LN DIRECTED or AO	LN MISSION RISK CLASS	LN MISSION DESTINATION	LN FLIGHT SYSTEM TYPE	LN LEAD ORGANIZATION	LN FLIGHT SYSTEM ORGANIZATION	LN PAYLOAD ORG.	LN FLIGHT SYSTEM LEAD ORG. EXPERIENCE	LN PAYLOAD LEAD ORG. EXPERIENCE	LN FLIGHT SYSTEM MASS	LN FLIGHT SYSTEM POWER	LN FLIGHT SYSTEM HERITAGE & TRL	LN PARTS RATING	LN # OF KEY SPACECRAFT CONTRACTORS	LN PAYLOAD MASS	LN PAYLOAD POWER	LN # OF PAYLOAD ELEMENTS	LN # OF KEY PAYLOAD CONTRACTORS	LN IN-HOUSE SCOPE	LN INTERNATIONAL PARTICIPATION (HW)
LN \$k-FY14 Per Month	1																				
LN DIRECTED or AO	0.25	1																			
LN MISSION RISK CLASS	-0.6	-0.5	1																		
LN MISSION DESTINATION	0.55	-0	-0.2	1																	
LN FLIGHT SYSTEM TYPE	0.16	0.07	0.09	0.37	1																
LN LEAD ORGANIZATION	0.66	0.29	-0.4	0.22	0.13	1															
LN FLIGHT SYSTEM ORGANIZATION	0.34	-0	0.06	0.09	0.12	0.27	1														
LN PAYLOAD ORG.	0.39	0.14	-0.2	-0	0.05	0.52	0.39	1													
LN FLIGHT SYSTEM LEAD ORG. EXPERIENCE	0.21	0.14	-0.1	0	-0.2	0.3	0.31	0.12	1												
LN PAYLOAD LEAD ORG. EXPERIENCE	0.63	0.2	-0.5	0.34	-0.1	0.26	0.33	0.15	0.49	1											
LN FLIGHT SYSTEM MASS	0.74	0.38	-0.5	0.48	0.11	0.55	0.07	0.3	0.07	0.48	1										
LN FLIGHT SYSTEM POWER	0.6	0.21	-0.4	0.06	-0.1	0.57	0.08	0.3	0.06	0.32	0.72	1									
LN FLIGHT SYSTEM HERITAGE & TRL	0.02	-0.1	0.01	-0	-0.2	0.02	0.3	0.02	0.44	0.38	-0.1	-0.2	1								
LN PARTS RATING	0.76	0.19	-0.6	0.54	0.22	0.52	-0	0.34	0.03	0.55	0.71	0.58	-0.2	1							
LN # OF KEY SPACECRAFT CONTRACTORS	0.43	0.04	-0.1	0.46	0.27	0.16	0.18	0.3	-0.2	0.06	0.38	0.19	-0.1	0.33	1						
LN PAYLOAD MASS	0.27	0.3	-0.3	-0.1	-0.1	0.44	0.11	0	0.43	0.2	0.28	0.31	0.14	0.19	-0.4	1					
LN PAYLOAD POWER	0.39	0.42	-0.4	-0.1	-0.1	0.46	0.2	0.28	0.36	0.41	0.54	0.55	0.12	0.34	-0	0.7	1				
LN # OF PAYLOAD ELEMENTS	0.08	0.06	-0.1	0.31	0.12	-0.4	-0.3	-0.2	-0.2	0.15	0.18	-0.2	-0.2	0.12	0.23	-0.4	-0.2	1			
LN # OF KEY PAYLOAD CONTRACTORS	-0.1	-0.1	-0.1	0.12	-0.1	-0.2	-0.3	-0.2	-0.2	-0.2	-0	-0.1	-0.3	-0.1	-0.1	-0.1	-0.3	0.58	1		
LN IN-HOUSE SCOPE	0.34	0.18	-0.3	0.33	0.16	0.1	-0.3	0.03	0.03	0.3	0.49	0.16	-0.1	0.47	0.34	-0.1	0.08	0.47	0.08	1	
LN INTERNATIONAL PARTICIPATION (HW)	-0	0.15	-0.1	-0.2	-0.1	0.13	-0	0.12	0.23	-0.1	-0.1	0.09	-0.2	-0.1	0.03	-0	0.03	-0.2	-0.3	-0.1	1





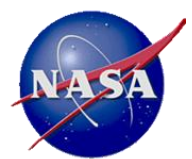
Results of the PCA



Engineering
Cost
Office

Explained variance ratio					LN DIRECTED or AO	LN MISSION RISK CLASS	LN MISSION DESTINATION	LN FLIGHT SYSTEM TYPE	LN LEAD ORGANIZATION	LN FLIGHT SYSTEM ORGANIZATION	LN PAYLOAD ORG.	LN FLIGHT SYSTEM LEAD ORG. EXPERIENCE	LN PAYLOAD LEAD ORG. EXPERIENCE	LN FLIGHT SYSTEM MASS	LN FLIGHT SYSTEM POWER	LN FLIGHT SYSTEM HERITAGE & TRL	LN PARTS RATING	LN # OF KEY SPACECRAFT CONTRACTORS	LN PAYLOAD MASS	LN PAYLOAD POWER	LN # OF PAYLOAD ELEMENTS	LN # OF KEY PAYLOAD CONTRACTORS	LN IN-HOUSE SCOPE	LN INTERNATIONAL PARTICIPATION (HW)
Component 1	40%			Component 1	-0.077	0.112	-0.049	-0.004	-0.151	-0.023	-0.089	-0.038	-0.044	-0.432	-0.472	0.008	-0.127	-0.003	-0.467	-0.542	0.064	0.044	-0.040	-0.010
Component 2	20%			Component 2	0.004	-0.104	0.255	0.057	0.034	-0.023	-0.012	-0.066	0.035	0.430	0.289	-0.112	0.176	0.099	-0.422	-0.253	0.505	0.276	0.136	0.005
Component 3	10%			Component 3	0.084	-0.050	0.057	0.000	-0.162	-0.069	-0.341	0.007	-0.002	-0.010	-0.389	-0.023	-0.033	-0.096	0.525	0.063	0.420	0.461	0.057	-0.076
Component 4	6%			Component 4	-0.115	0.010	-0.358	-0.242	-0.010	-0.048	-0.206	-0.070	-0.090	-0.226	0.589	-0.136	-0.092	-0.097	0.138	-0.301	-0.129	0.391	-0.160	0.042
Component 5	6%			Component 5	0.006	0.046	-0.487	-0.186	-0.248	-0.045	0.100	-0.030	0.005	-0.119	0.034	-0.016	-0.144	-0.019	-0.349	0.562	0.421	-0.005	-0.030	-0.007
Component 6	3%			Component 6	-0.114	-0.152	-0.091	-0.013	0.301	0.183	0.778	0.051	-0.039	-0.089	-0.137	-0.018	-0.007	-0.061	0.148	-0.104	0.154	0.304	-0.185	-0.091
Component 7	3%			Component 7	0.438	-0.274	-0.366	0.123	0.058	-0.219	0.125	-0.043	-0.096	0.052	-0.080	-0.289	-0.011	-0.072	0.107	-0.212	0.038	-0.177	0.348	0.445
Component 8	2%			Component 8	-0.050	0.195	-0.063	0.531	-0.225	-0.027	0.093	-0.425	-0.289	0.113	-0.017	-0.445	-0.049	0.004	0.002	0.078	-0.156	0.081	-0.081	-0.304
Component 9	2%			Component 9	0.231	0.039	-0.226	-0.336	-0.203	-0.078	0.126	-0.082	-0.009	0.519	-0.147	0.310	-0.207	-0.047	-0.047	-0.174	-0.284	0.140	0.143	-0.355
Component 10	2%			Component 10	-0.623	-0.034	-0.168	-0.017	-0.159	-0.180	0.074	-0.037	-0.004	0.084	0.055	0.075	0.093	-0.160	0.255	-0.234	0.258	-0.426	0.261	-0.177
Component 11	1%			Component 11	-0.179	0.625	-0.071	0.077	0.300	0.006	-0.027	0.246	-0.142	0.276	-0.052	-0.036	-0.448	0.027	0.041	-0.026	0.097	0.038	0.062	0.310
Component 12	1%			Component 12	0.281	-0.079	-0.241	0.224	0.094	0.335	-0.189	0.019	0.093	0.131	0.079	0.117	-0.174	0.044	0.112	-0.236	0.304	-0.361	-0.487	-0.209
Component 13	1%			Component 13	-0.093	-0.057	0.248	-0.460	-0.352	0.089	0.095	-0.110	-0.036	0.234	-0.124	-0.395	-0.093	-0.024	0.085	-0.053	0.005	-0.178	-0.417	0.330
Component 14	1%			Component 14	0.213	0.282	0.017	-0.396	0.387	-0.162	-0.051	0.061	0.050	-0.096	-0.059	-0.476	0.204	0.011	0.018	-0.014	0.095	-0.183	0.062	-0.454
Component 15	1%			Component 15	-0.284	-0.336	-0.146	-0.106	0.530	-0.165	-0.277	-0.422	-0.179	0.170	-0.251	0.072	-0.030	0.097	-0.119	0.133	-0.065	0.019	-0.174	0.067
Component 16	1%			Component 16	0.055	-0.004	0.102	-0.144	-0.066	0.034	0.104	-0.217	-0.190	-0.190	0.105	0.093	-0.221	0.813	0.194	-0.051	0.122	-0.074	0.197	-0.035
Component 17	0%			Component 17	0.221	0.041	0.348	0.024	0.040	-0.612	0.153	-0.149	-0.199	-0.149	0.181	0.262	-0.268	-0.278	0.021	0.001	0.166	-0.109	-0.231	-0.032
Component 18	0%			Component 18	0.055	0.368	-0.237	0.048	-0.088	-0.302	0.083	-0.047	0.026	0.095	-0.087	0.224	0.619	0.268	0.086	-0.076	0.019	0.055	-0.335	0.208
Component 19	0%			Component 19	0.104	0.315	0.058	-0.100	0.100	0.316	0.054	-0.684	0.333	-0.099	0.039	0.149	0.039	-0.257	0.075	-0.017	0.091	-0.030	0.194	0.176
Component 20	0%			Component 20	-0.111	-0.066	-0.038	0.152	0.004	-0.359	0.060	-0.028	0.802	0.026	-0.018	-0.176	-0.283	0.214	0.021	-0.012	-0.087	0.080	-0.094	-0.020

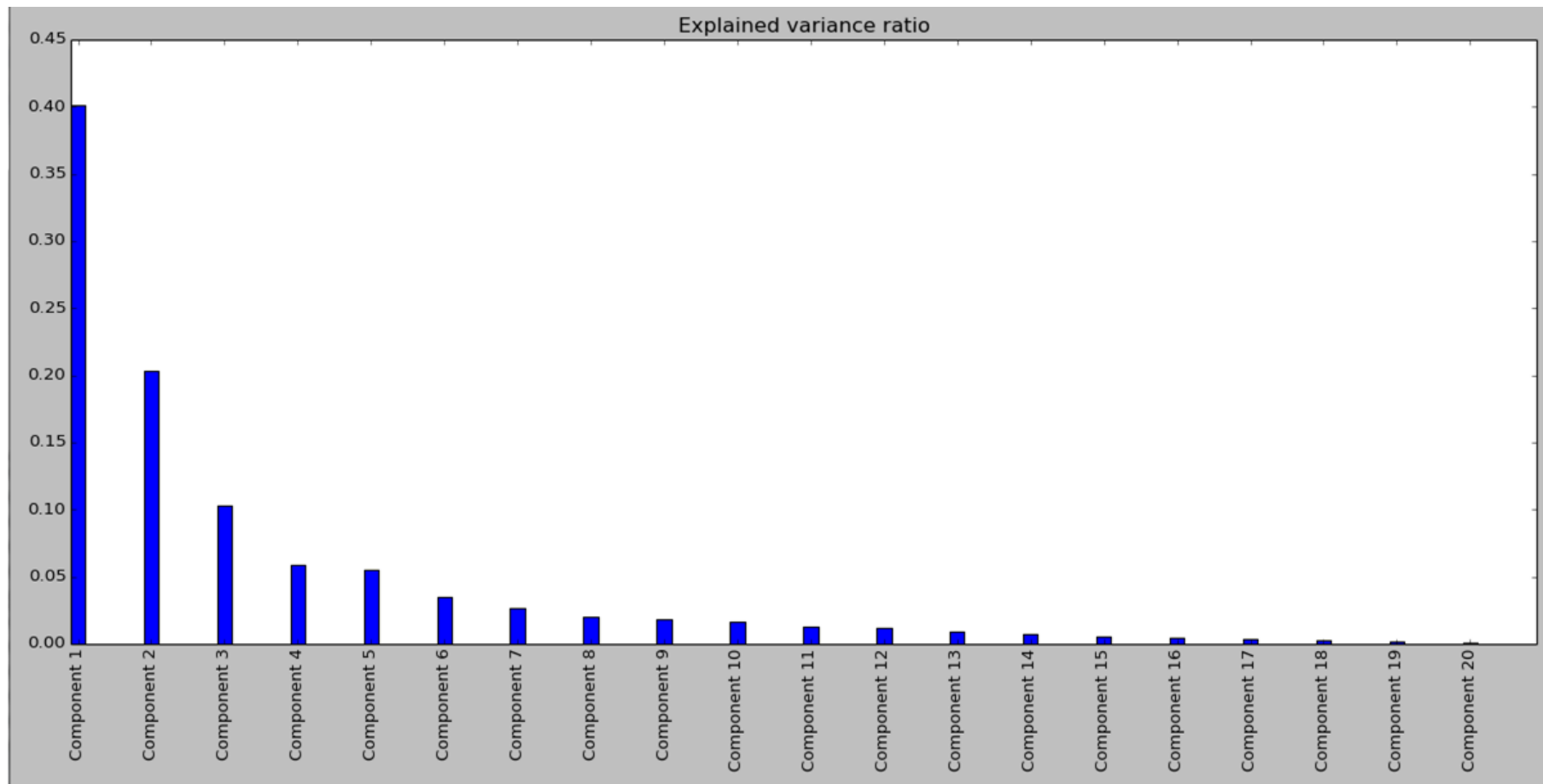




Variance Explained by PCA Components



Engineering
Cost
Office



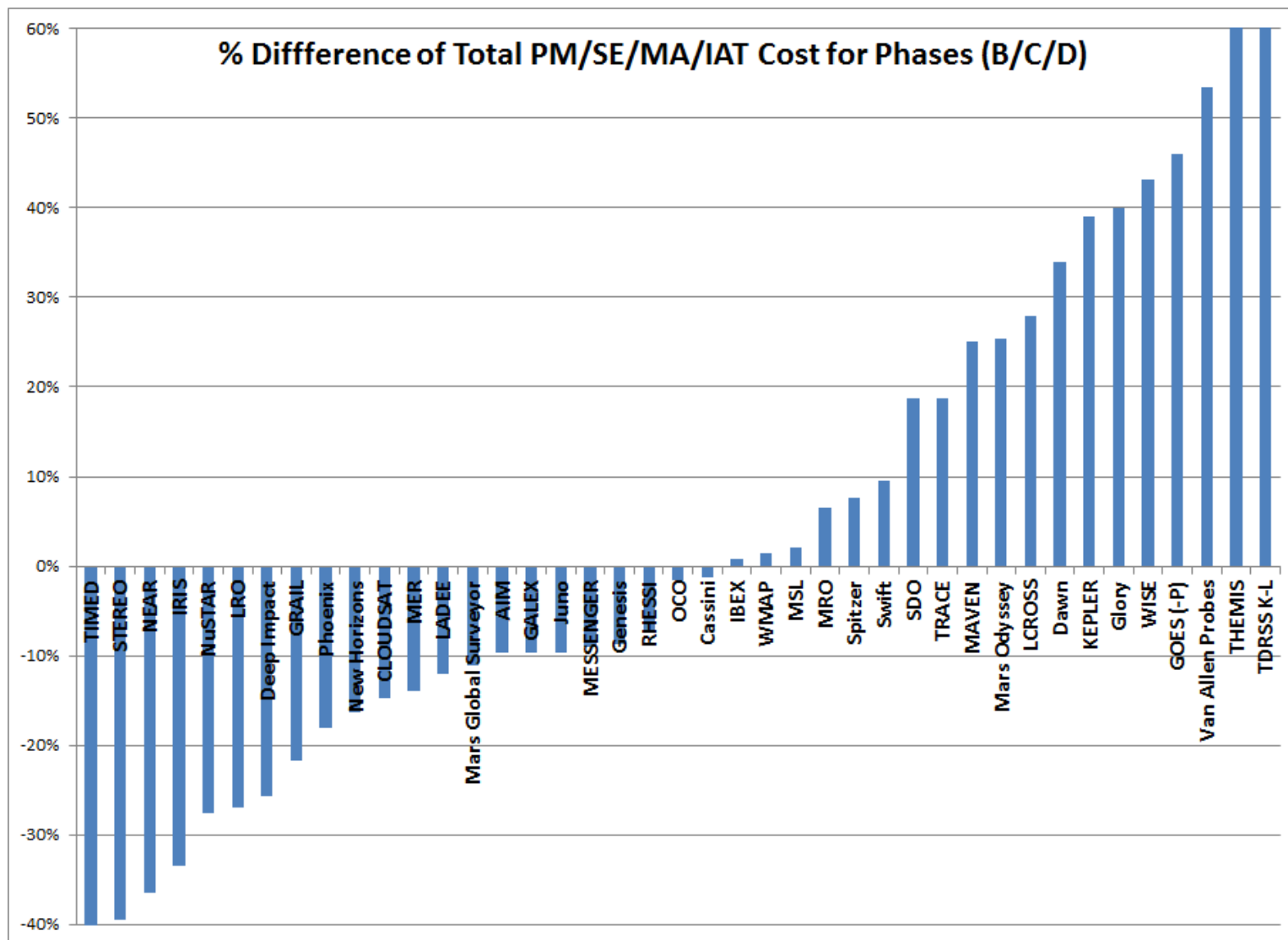


PCA Approach Results

Total PM/SE/MA/I&T for Phase B/C/D



Engineering
Cost
Office

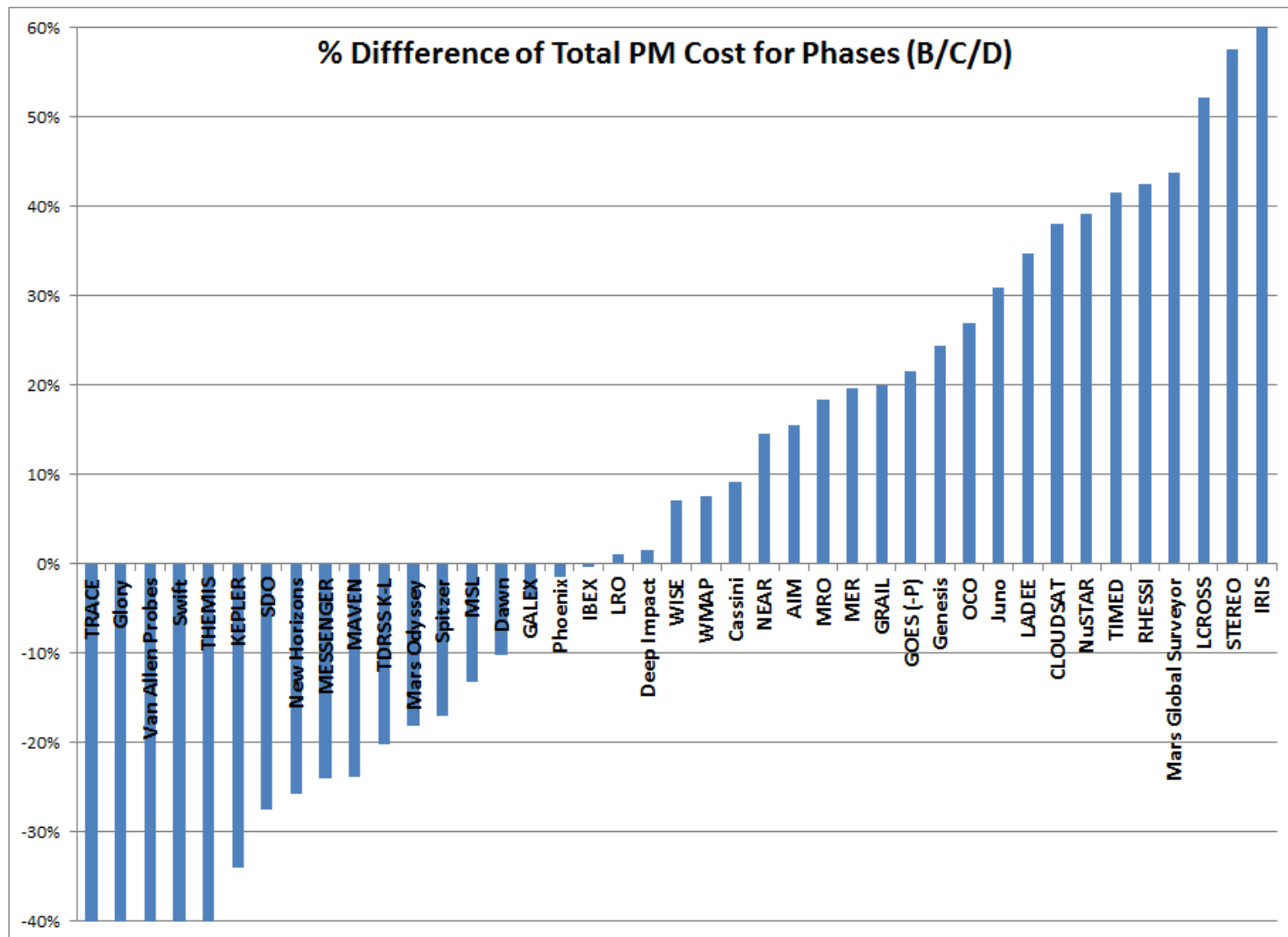




PCA Approach Results – Total PM for Phase B/C/D



Engineering
Cost
Office

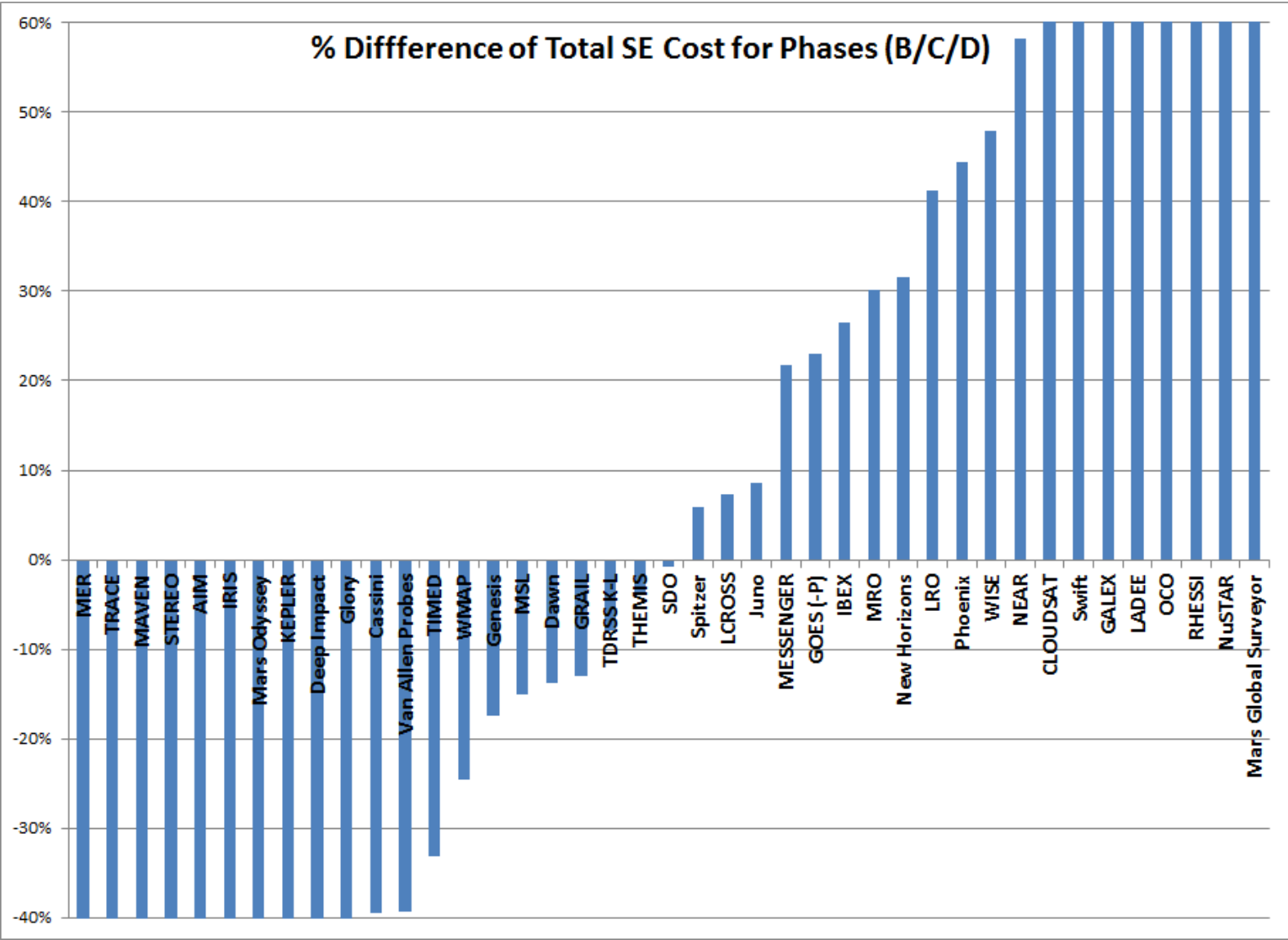




PCA Approach Results – Total SE for Phase B/C/D



Engineering
Cost
Office

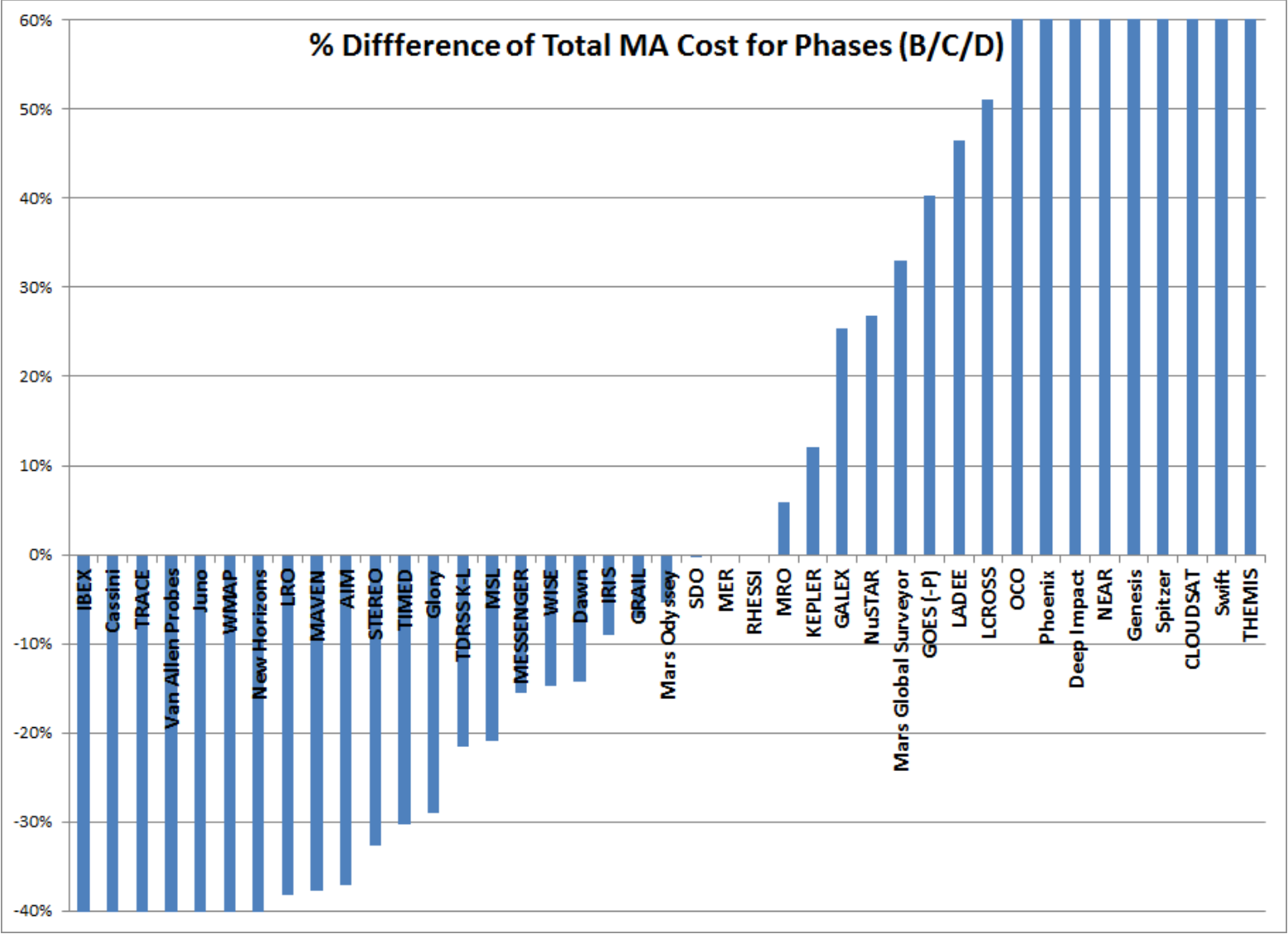




PCA Approach Results – Total MA for Phase B/C/D



Engineering
Cost
Office

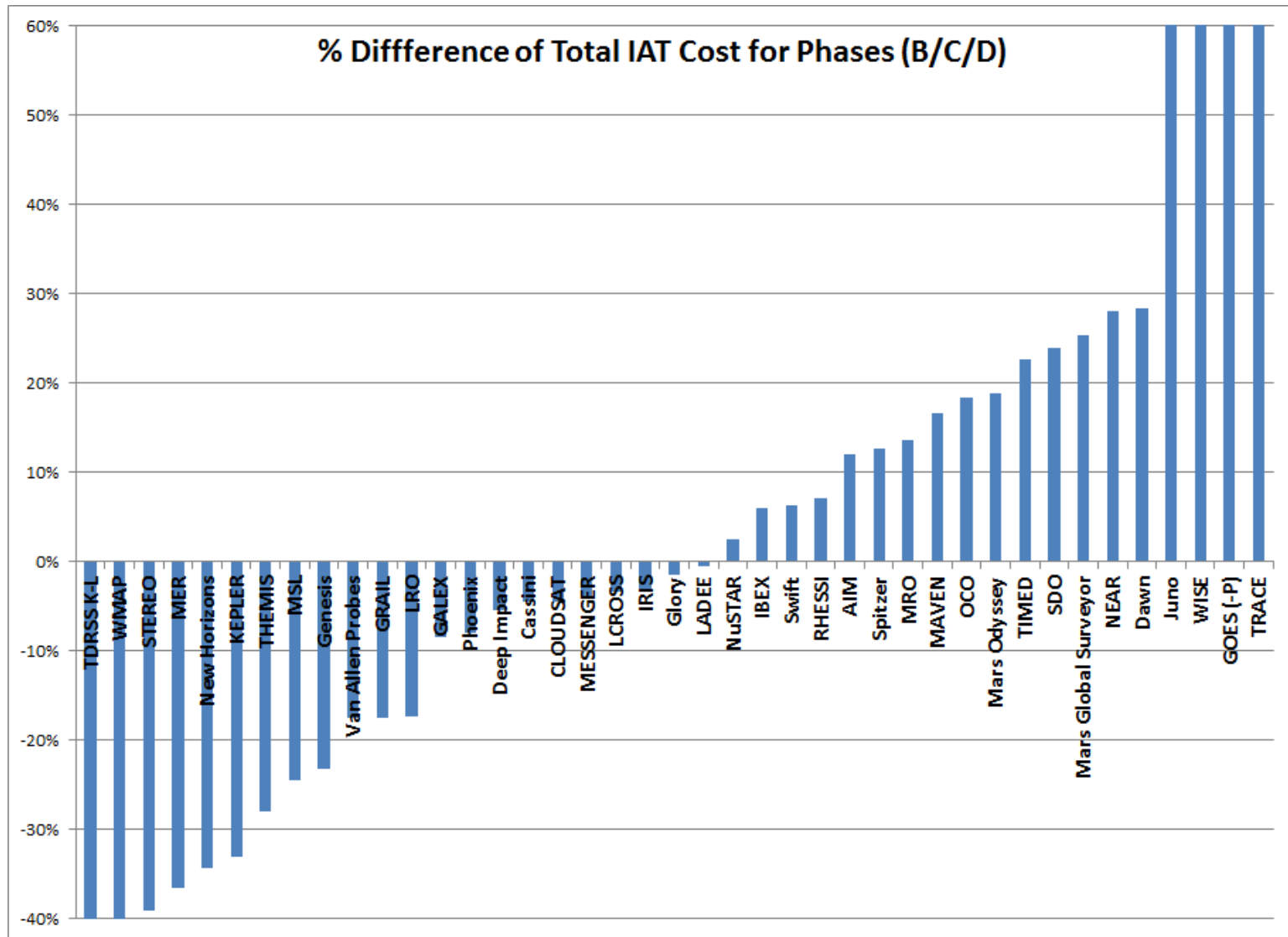




PCA Approach Results – Total I&T for Phase B/C/D



Engineering
Cost
Office

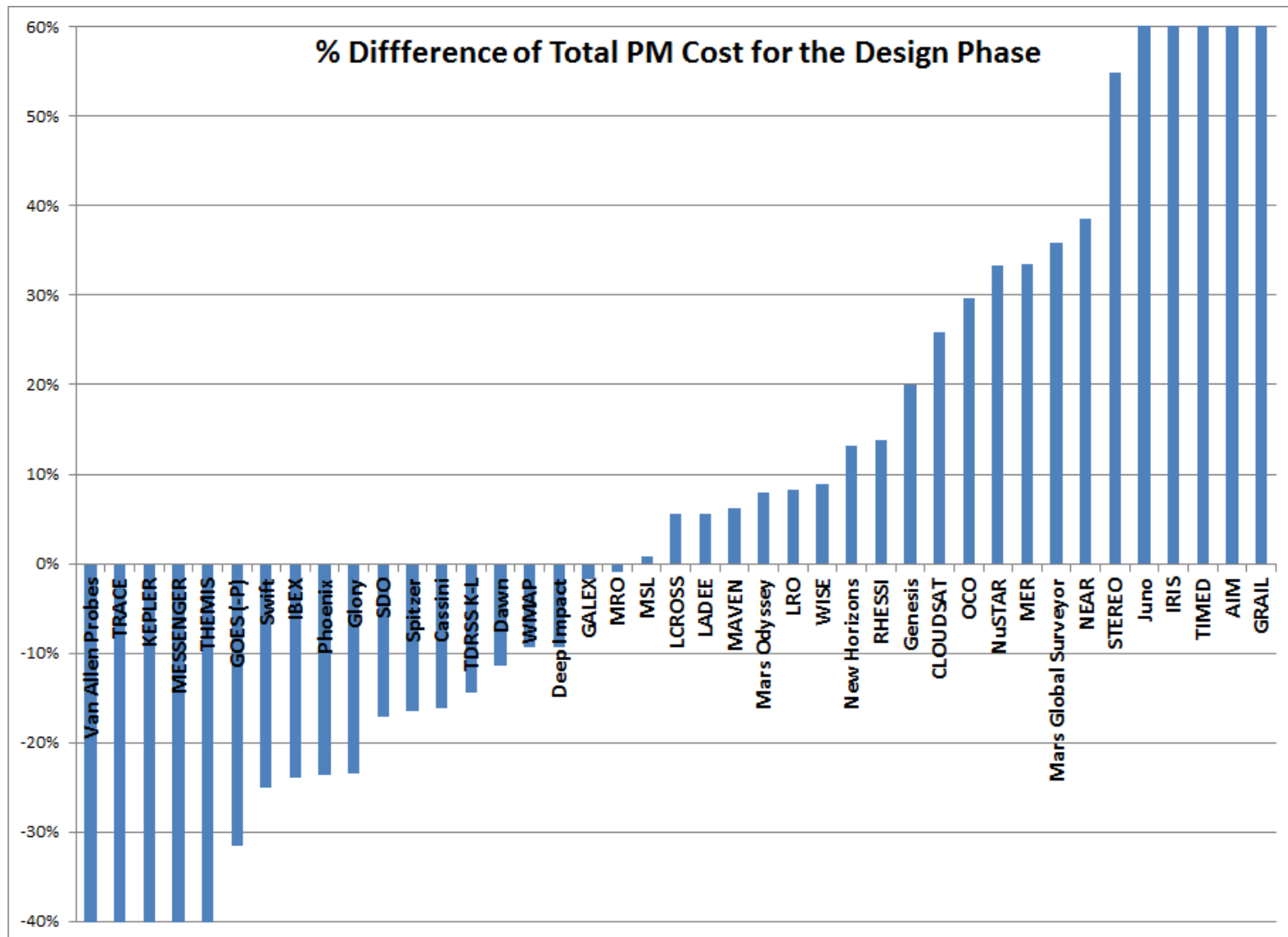




PCA Approach Results – PM for the Design Phase



Engineering
Cost
Office

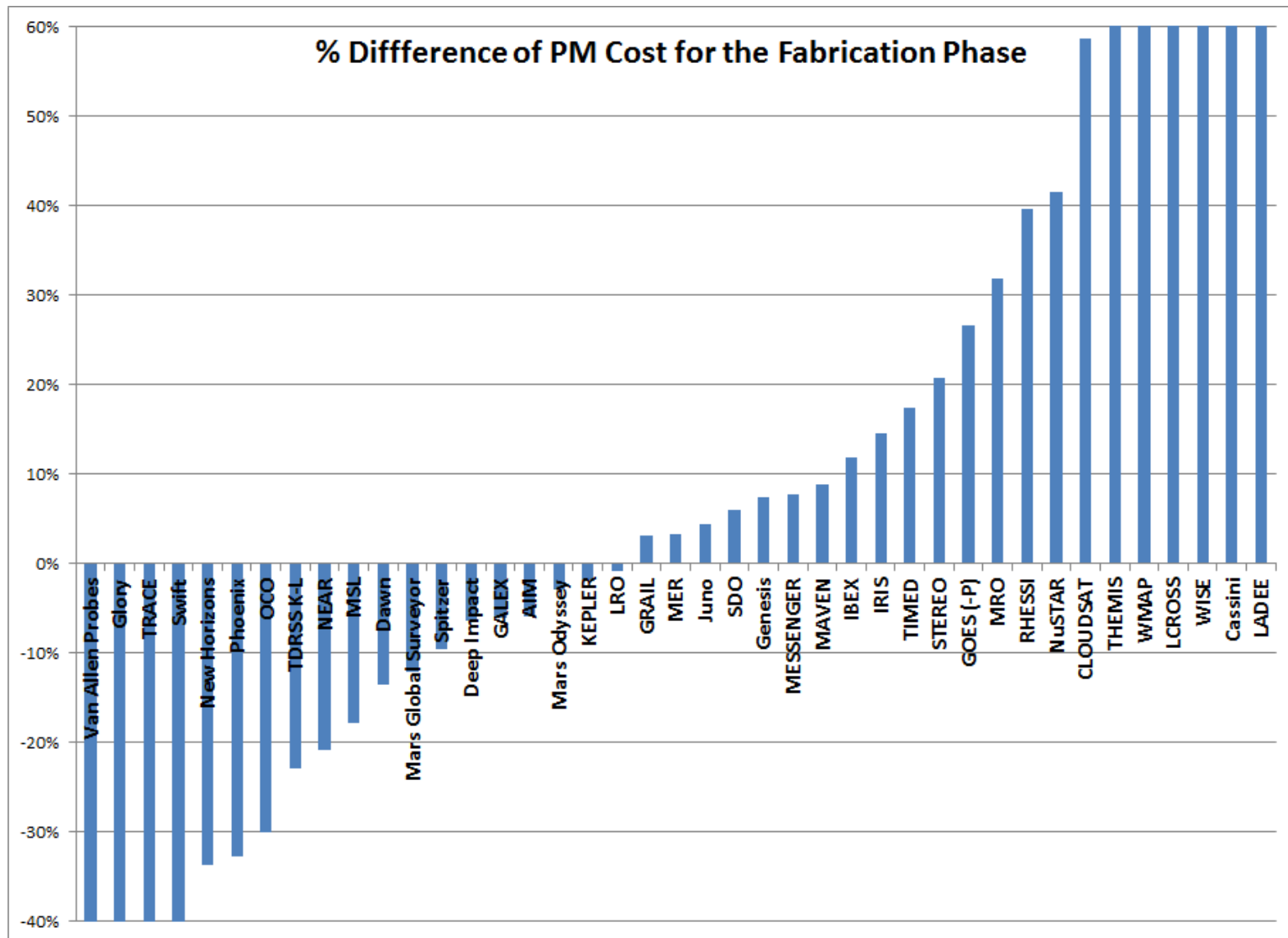


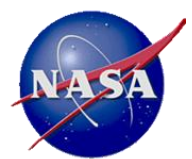


PCA Approach Results – PM for the Fabrication Phase

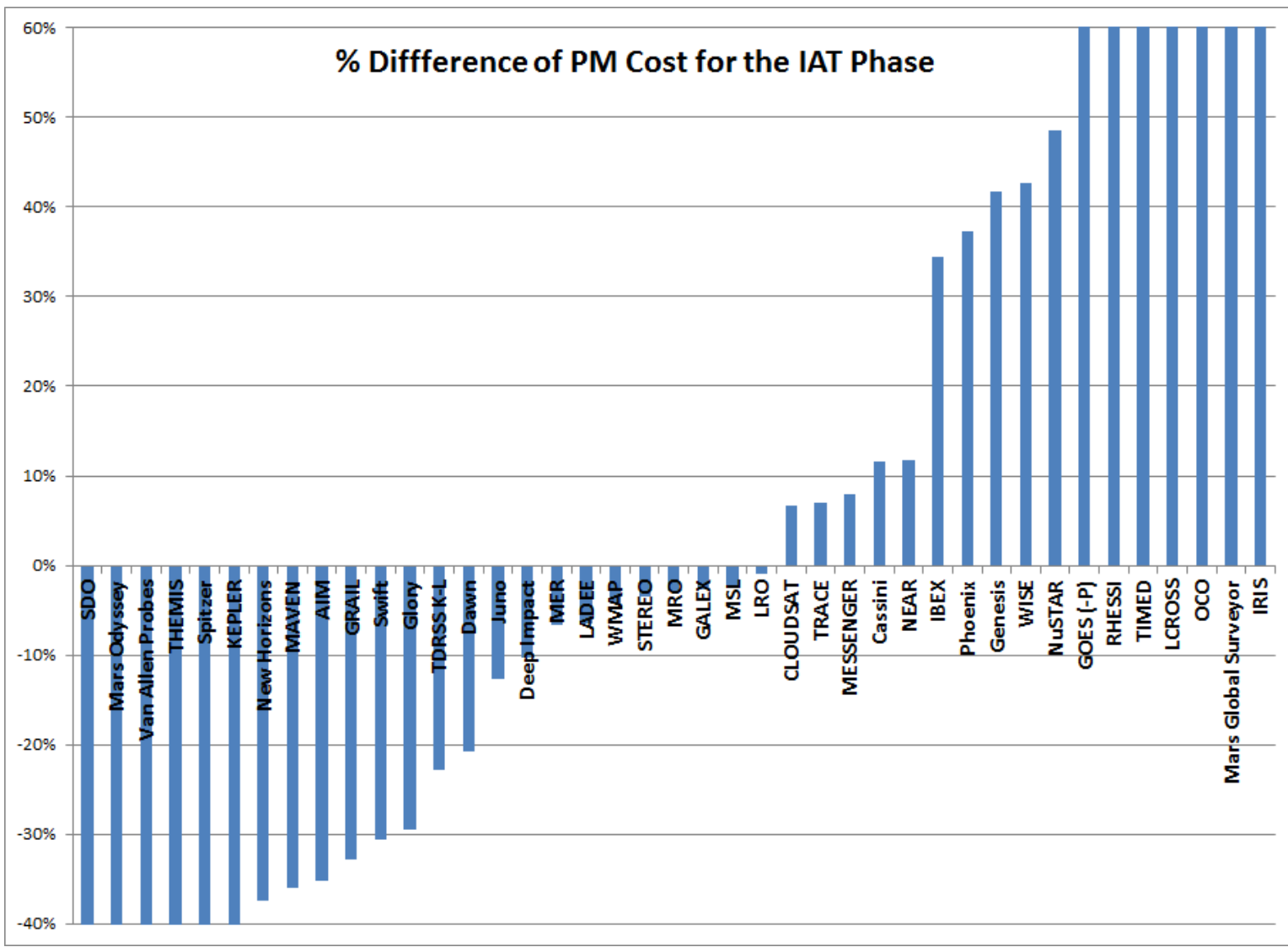


Engineering
Cost
Office



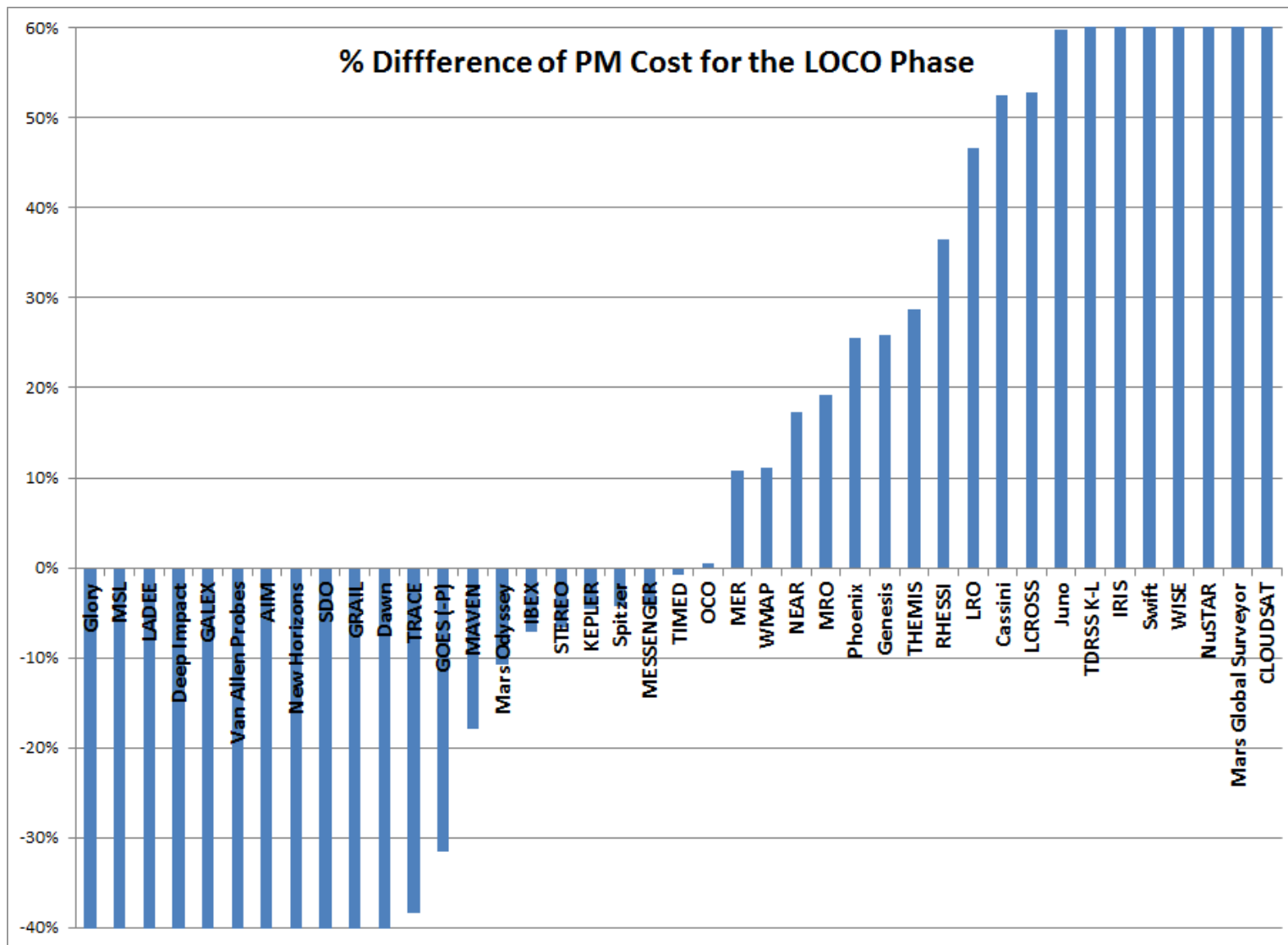


PCA Approach Results – PM for the I&T Phase



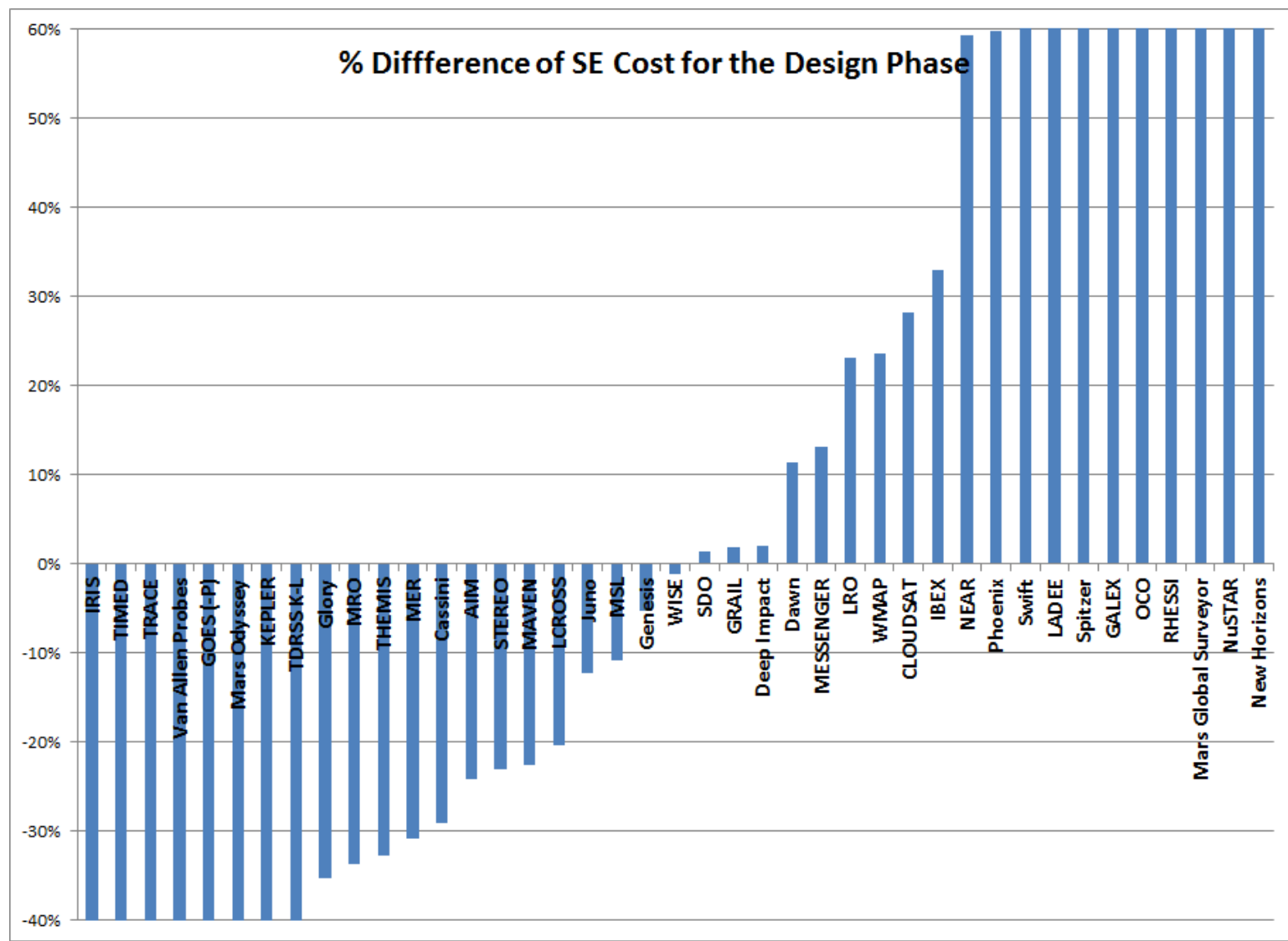


PCA Approach Results – PM for the LOCO Phase



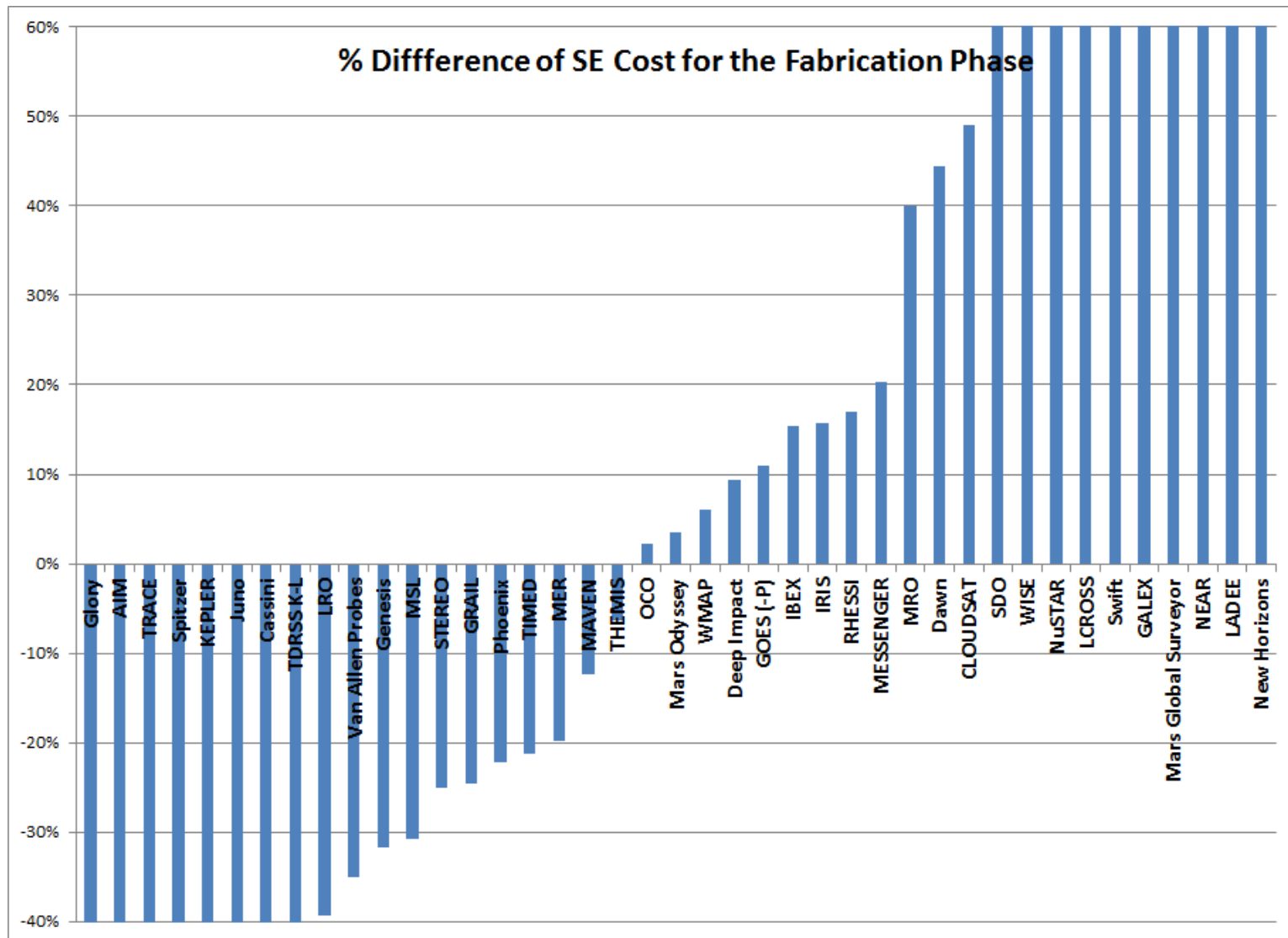


PCA Approach Results – SE for the Design Phase



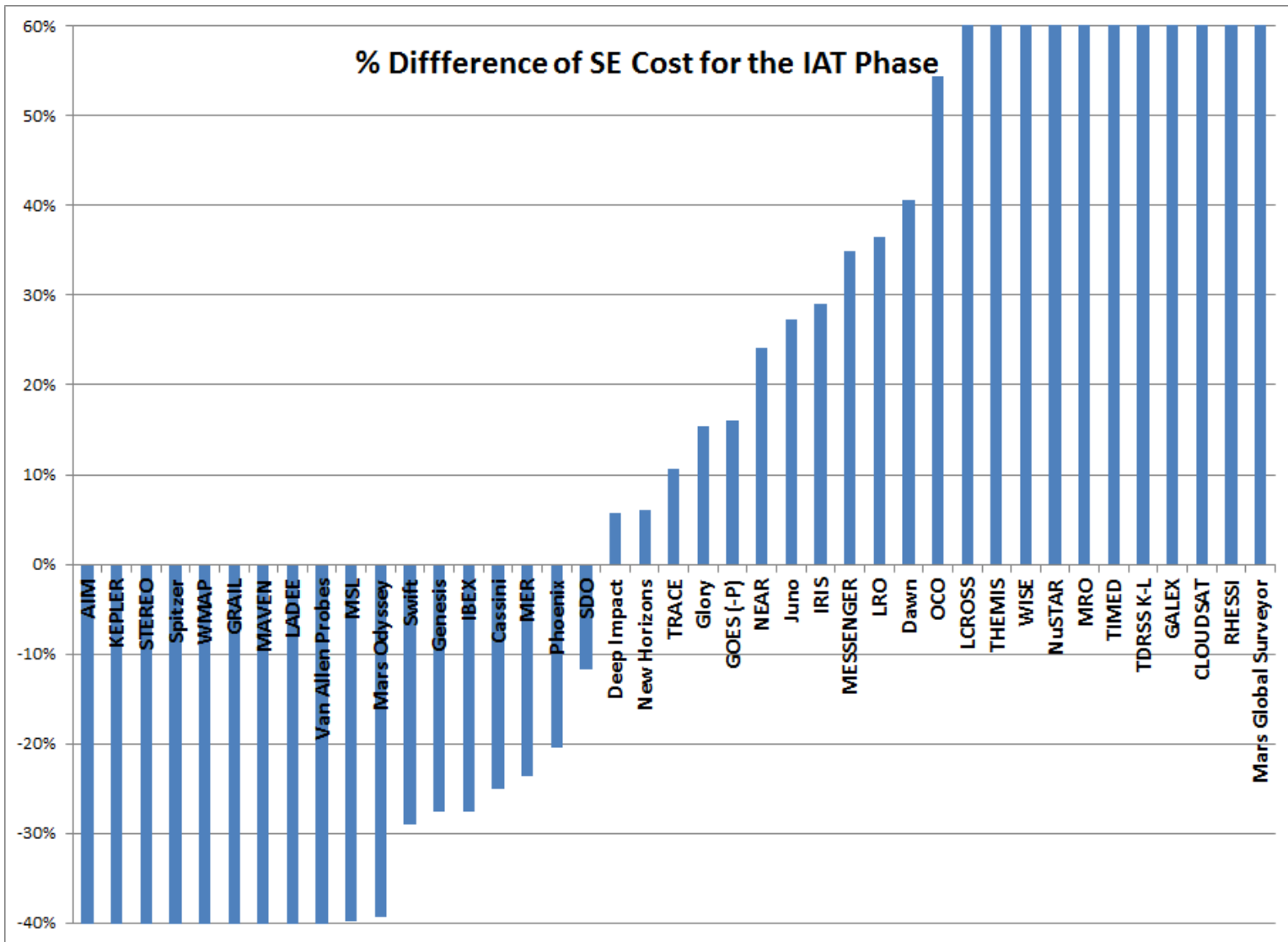


PCA Approach Results – SE for the Fabrication Phase



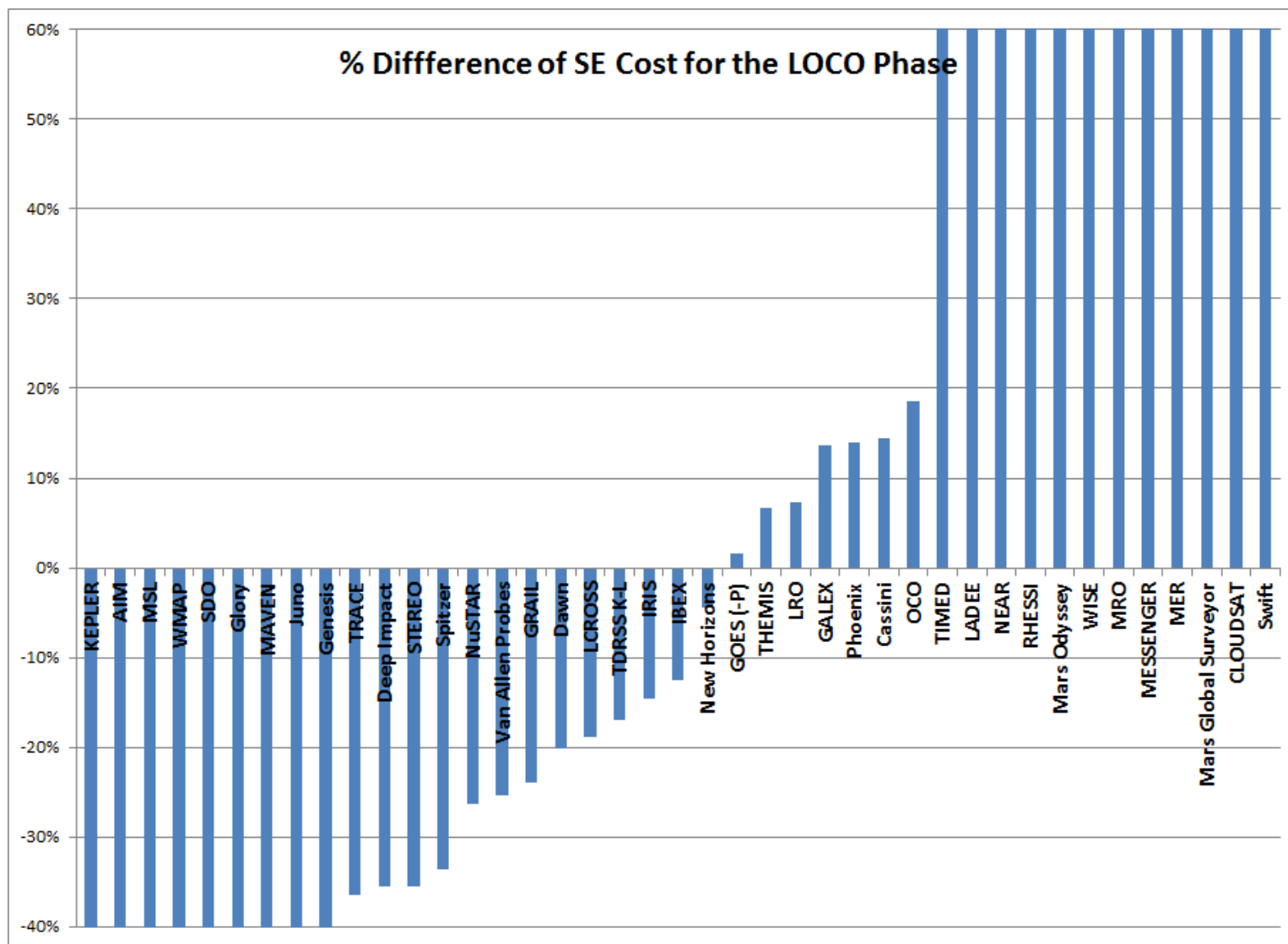


PCA Approach Results – SE for the I&T Phase





PCA Approach Results – SE for the LOCO Phase

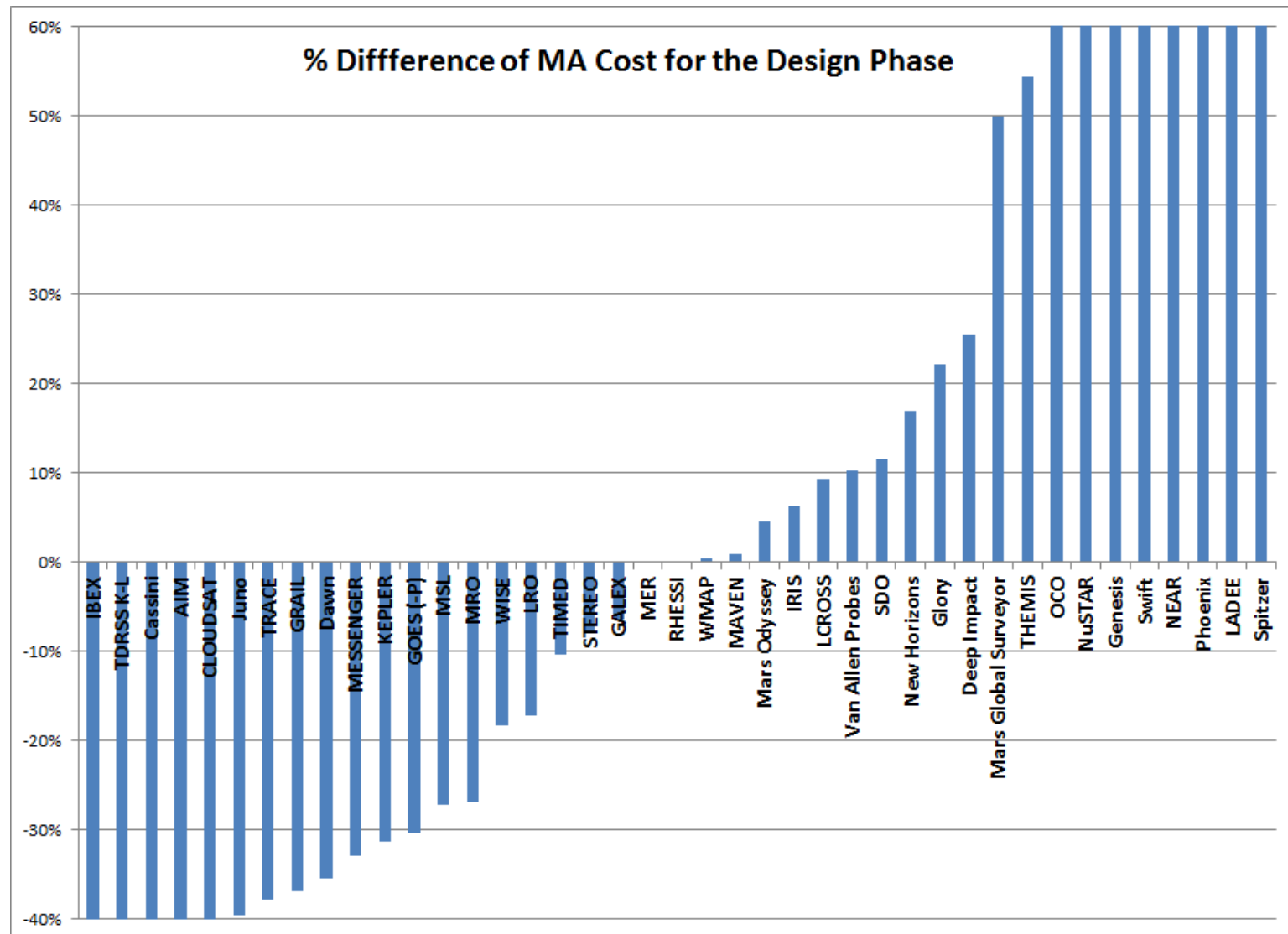




PCA Approach Results – MA for the Design Phase

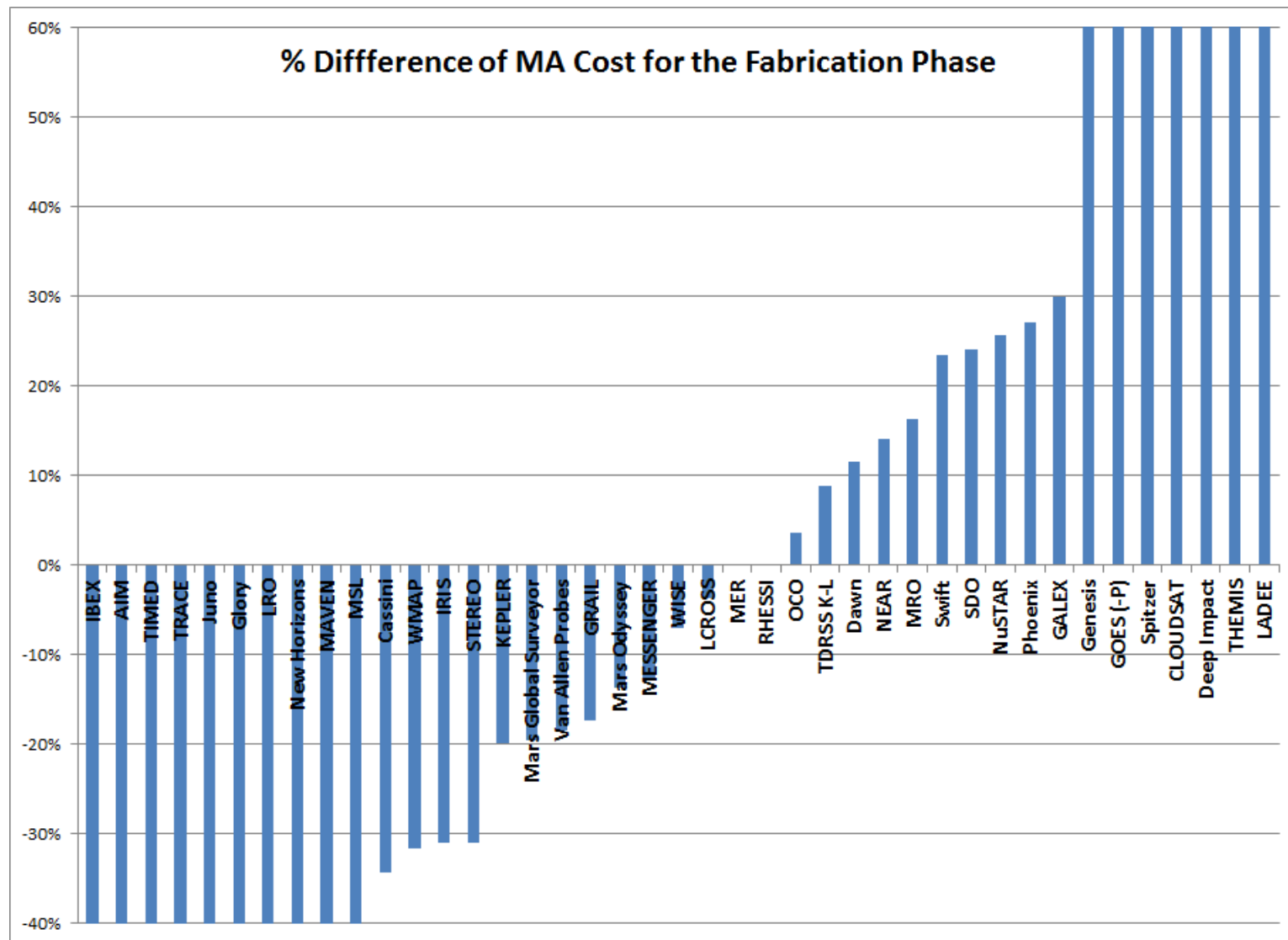


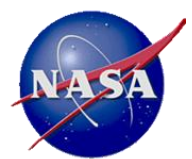
Engineering
Cost
Office



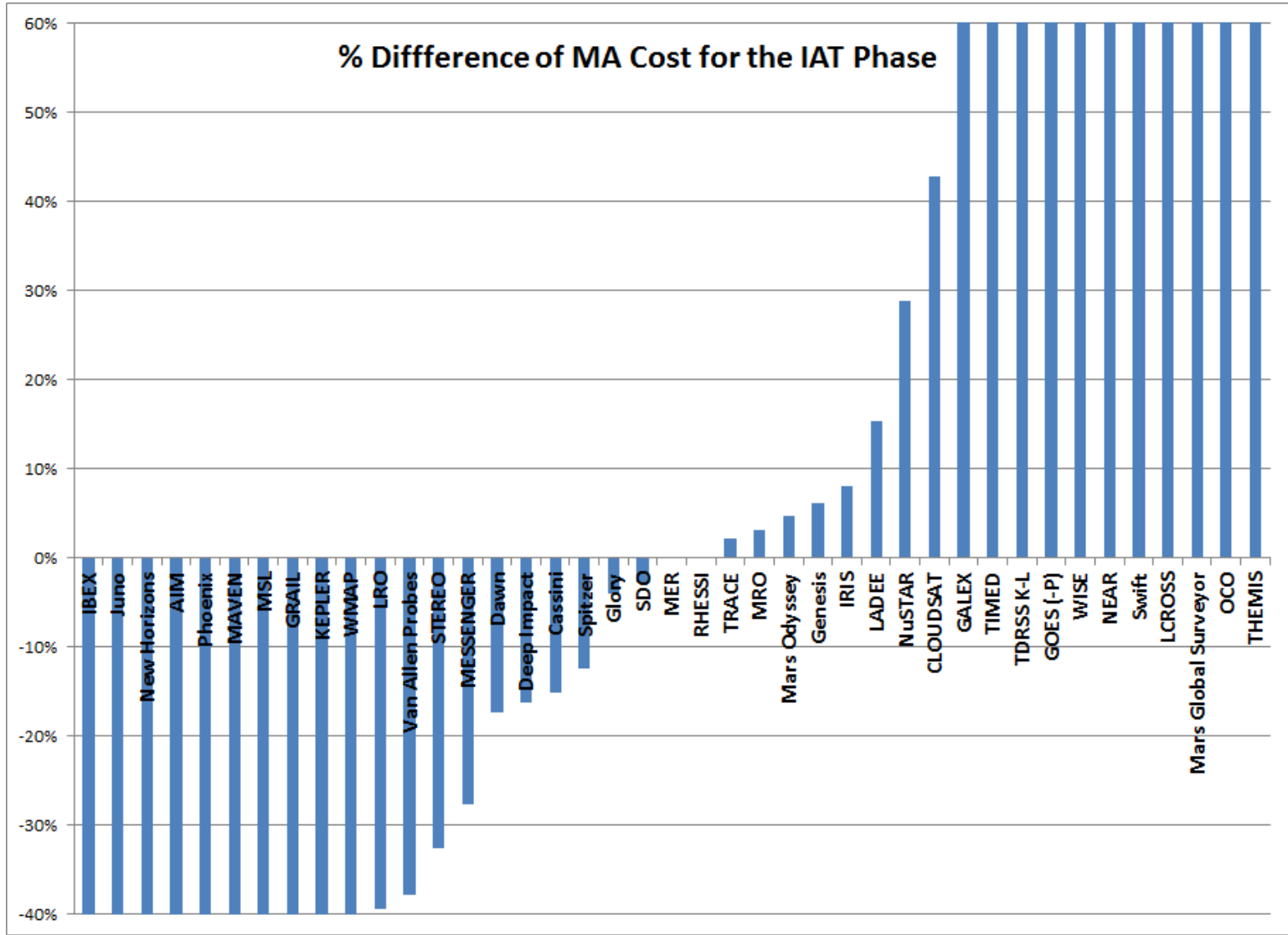


PCA Approach Results – MA for the Fabrication Phase



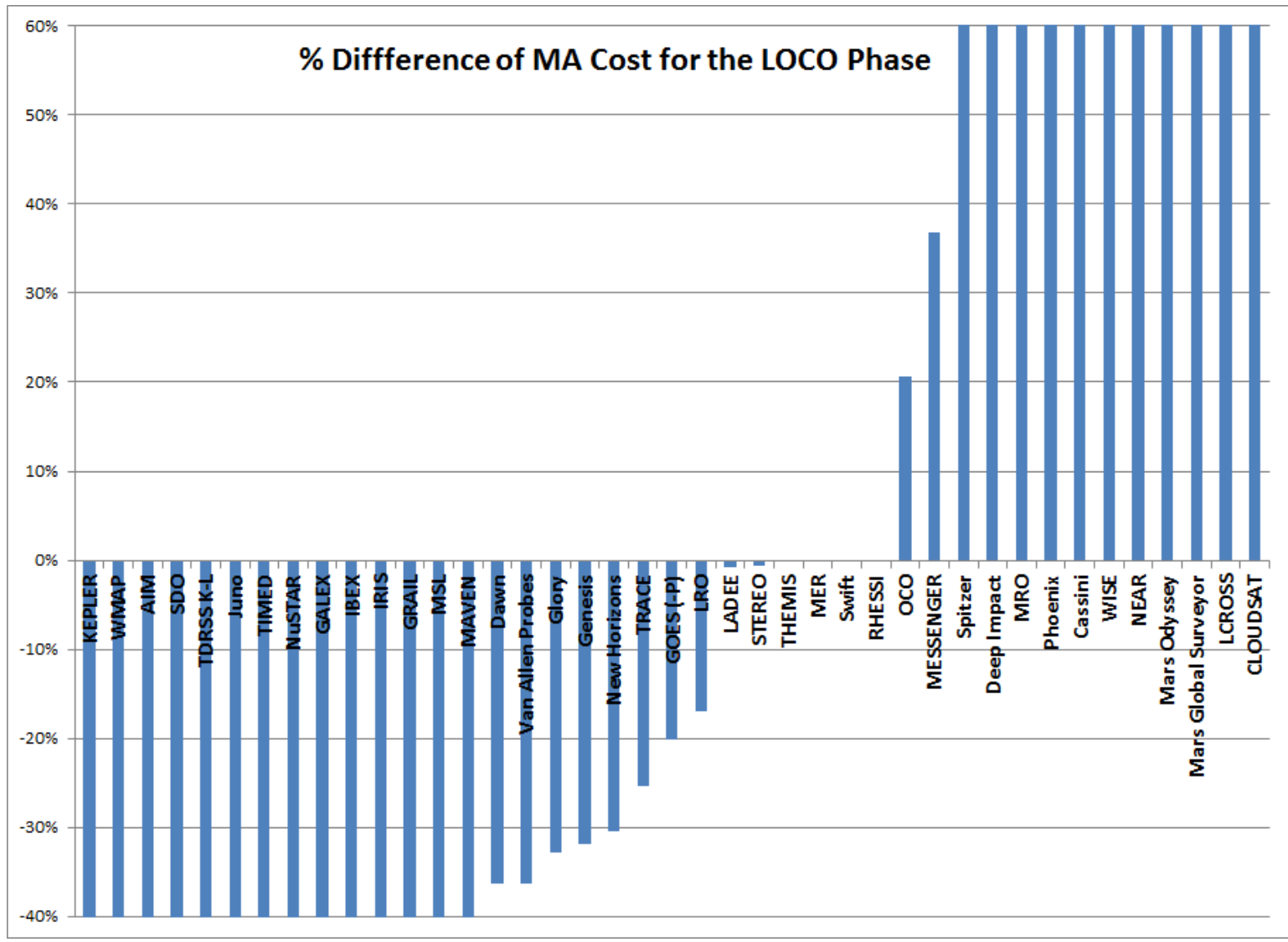


PCA Approach Results – MA for the I&T Phase



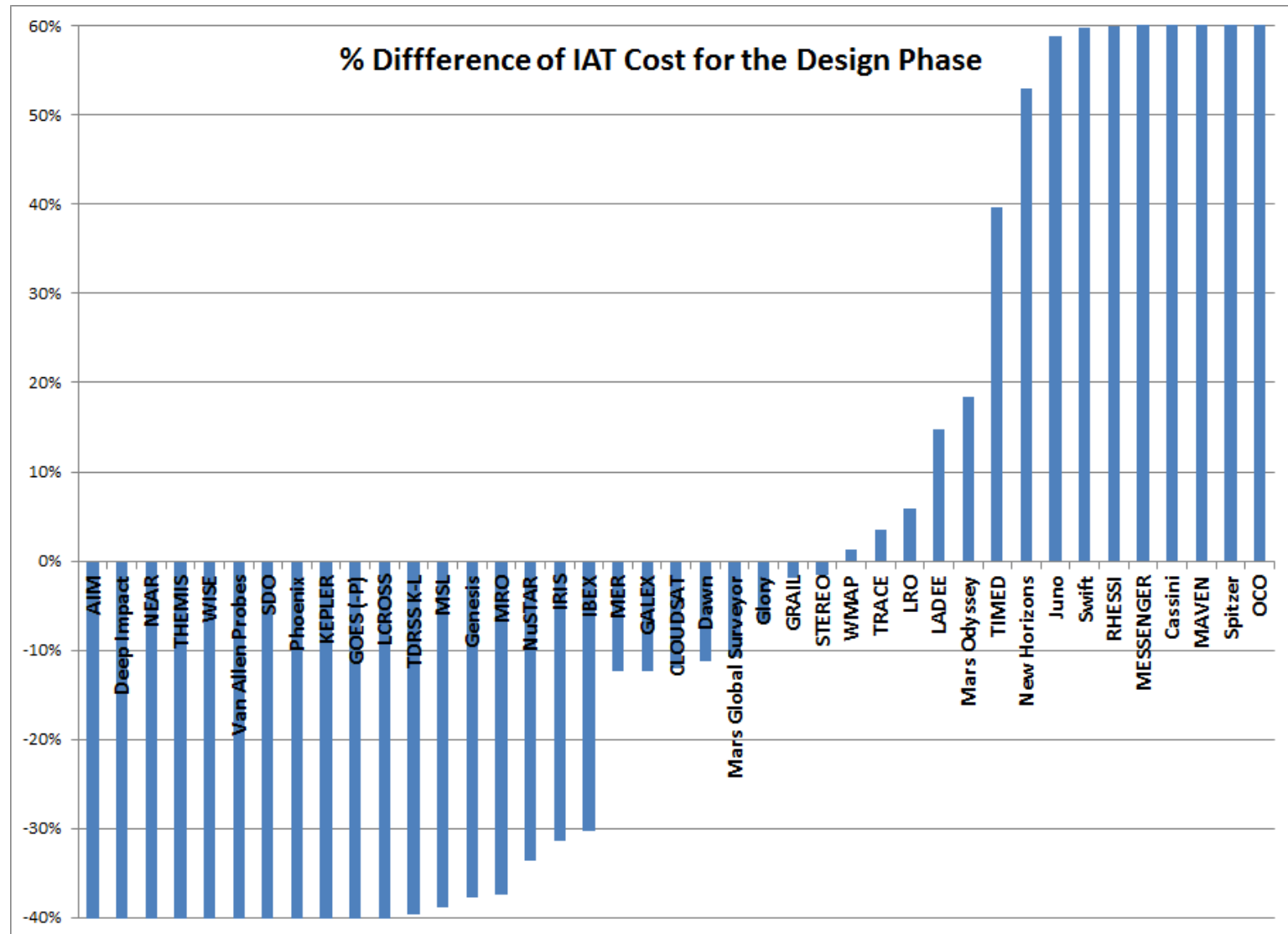


PCA Approach Results – MA for the LOCO Phase





PCA Approach Results – I&T for the Design Phase

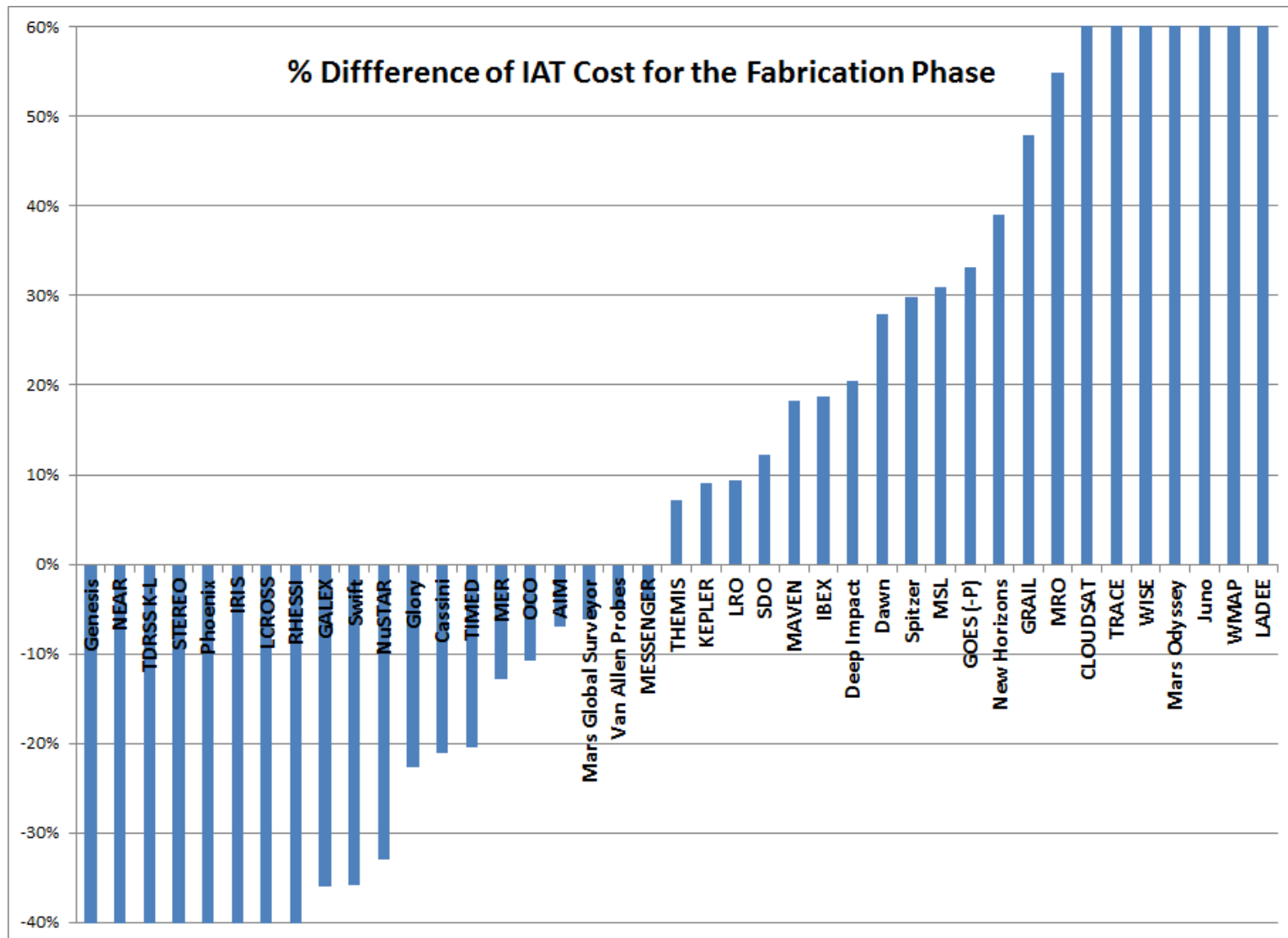




PCA Approach Results – I&T for the Fabrication Phase



Engineering
Cost
Office

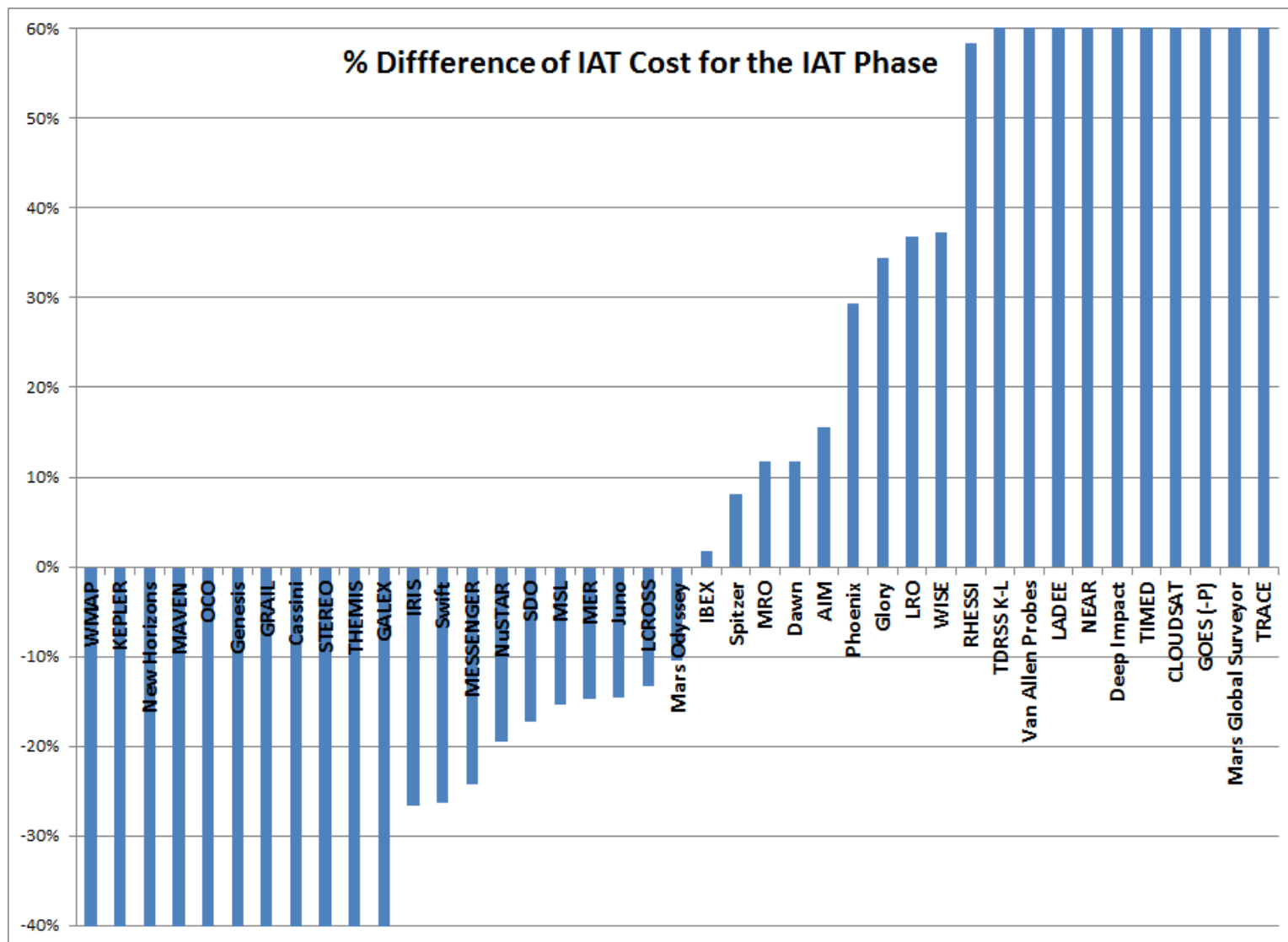




PCA Approach Results – I&T for the I&T Phase

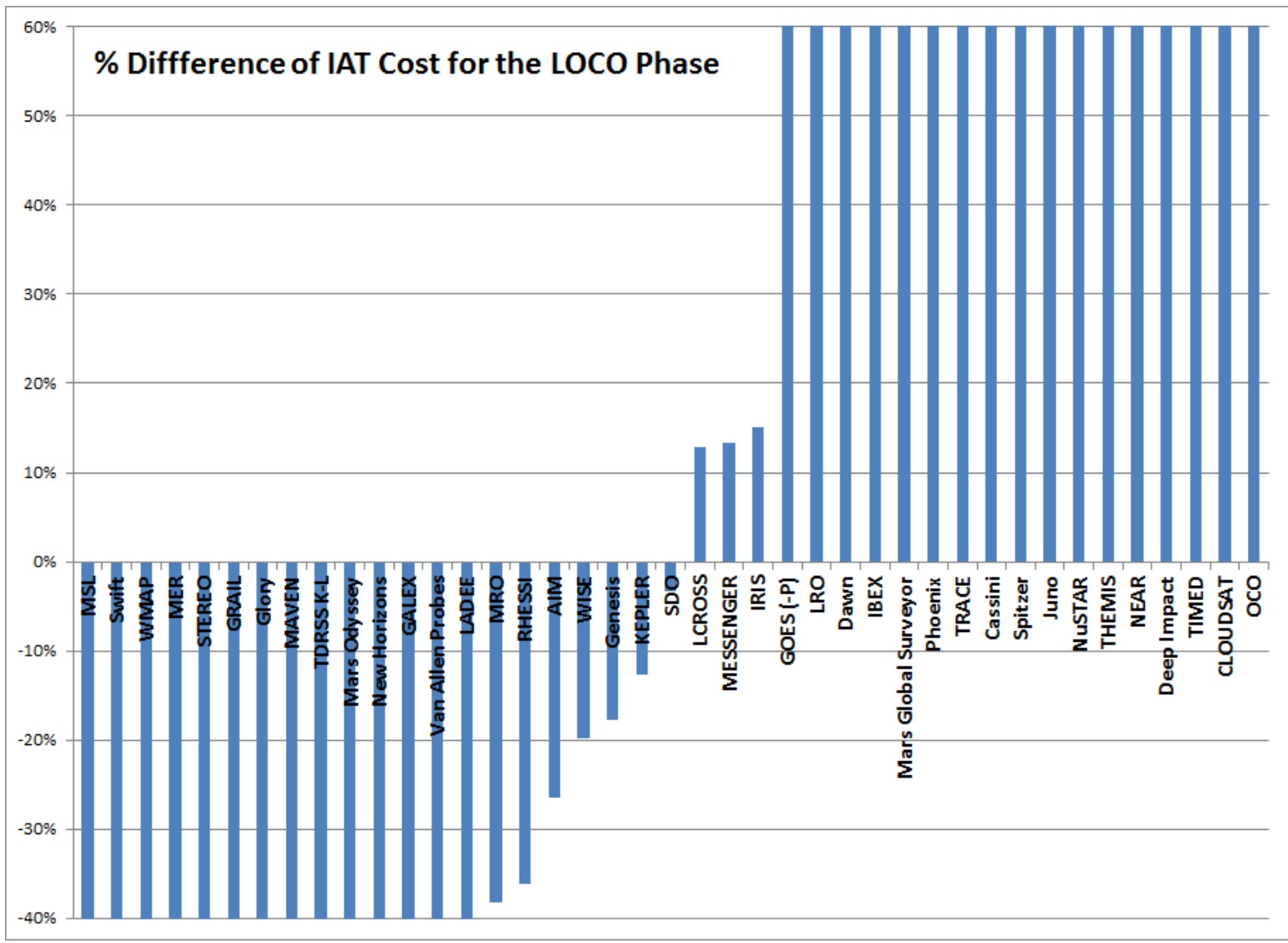


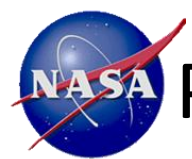
Engineering
Cost
Office





PCA Approach Results – I&T for the LOCO Phase





PCA Support Function CERs – Level 1 /2

Parameters	Total PM	Total SE	Total MA	Total IAT	Total PM/SE/MA/IAT
Parameter 1	FLIGHT SYSTEM TYPE	MISSION RISK CLASS	LEAD ORGANIZATION	DIRECTED or AO	MISSION RISK CLASS
Parameter 2	LEAD ORGANIZATION	LEAD ORGANIZATION	PARTS RATING	MISSION DESTINATION	MISSION DESTINATION
Parameter 3	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM ORGANIZATION	-	FLIGHT SYSTEM ORGANIZATION	LEAD ORGANIZATION
Parameter 4	FLIGHT SYSTEM HERITAGE & TRL	# OF PAYLOAD ELEMENTS	-	FLIGHT SYSTEM POWER	FLIGHT SYSTEM ORGANIZATION
Parameter 5	PARTS RATING	-	-	PAYLOAD MASS	FLIGHT SYSTEM POWER
Parameter 6	-	-	-	# OF PAYLOAD ELEMENTS	# OF PAYLOAD ELEMENTS
Parameter 7	-	-	-	# OF KEY PAYLOAD CONTRACTORS	-
Factors	Total PM	Total SE	Total MA	Total IAT	Total PM/SE/MA/IAT
α	-0.4491	-0.6289	1.0714	-0.4174	-0.3374
β	0.5073	1.6227	1.4507	0.5089	0.3264
γ	1.391	0.7725	-	1.0959	0.6317
δ	-0.4	0.3192	-	0.3182	0.8013
ϵ	0.5882	-	-	0.2264	0.1818
ζ	-	-	-	0.5159	0.1647
η	-	-	-	-0.2824	-
Constant	3.8485	3.1888	1.8207	0.83	4.2309

- Equations take the form of:
 - $\text{LN}(\$/\text{mo}) = \alpha \cdot \text{LN}(\text{Parameter 1}) + \beta \cdot \text{LN}(\text{Parameter 2}) + \dots + \text{Constant}$
- For example, the equation for Total MA would be:
 - $\text{Total MA Cost } (\$/\text{mo FY14}) = e^{1.0714 \cdot \text{LN}(\text{LEAD ORG.}) + 1.4507 \cdot \text{LN}(\text{PARTS RATING}) + 1.8207}$





PCA Support Function CERs – Level 3 (PM)



Parameters	PM - Design	PM - Fabrication	PM - IAT	PM - LOCO	Total PM
Parameter 1	FLIGHT SYSTEM TYPE	MISSION RISK CLASS	DIRECTED or AO	DIRECTED or AO	FLIGHT SYSTEM TYPE
Parameter 2	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM TYPE	LEAD ORGANIZATION	LEAD ORGANIZATION	LEAD ORGANIZATION
Parameter 3	FLIGHT SYSTEM HERITAGE & TRL	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM ORGANIZATION
Parameter 4	PARTS RATING	FLIGHT SYSTEM POWER	# OF KEY PAYLOAD CONTRACTORS	PAYLOAD ORG.	FLIGHT SYSTEM HERITAGE & TRL
Parameter 5	# OF KEY PAYLOAD CONTRACTORS	PARTS RATING	-	# OF KEY PAYLOAD CONTRACTORS	PARTS RATING
Parameter 6	IN-HOUSE SCOPE	PAYLOAD MASS	-	IN-HOUSE SCOPE	-
Parameter 7	-	PAYLOAD POWER	-	-	-
Parameter 8	-	IN-HOUSE SCOPE	-	-	-
Factors	PM - Design	PM - Fabrication	PM - IAT	PM - LOCO	Total PM
α	-0.8136	0.7634	-0.5253	-0.779	-0.4491
β	2.1938	-0.5482	0.7998	1.1021	0.5073
γ	-0.6258	2.3989	1.3229	1.5468	1.391
δ	0.7004	0.2107	0.4001	-0.6091	-0.4
ϵ	-0.3077	1.1472	-	0.4181	0.5882
ζ	0.5733	0.2758	-	0.8902	-
η	-	-0.2687	-	-	-
θ	-	0.6619	-	-	-
Constant	3.6442	-0.7276	3.1145	1.907	3.8485

- Equations take the form of:
 - $\text{LN}(\$/\text{mo}) = \alpha \cdot \text{LN}(\text{Parameter 1}) + \beta \cdot \text{LN}(\text{Parameter 2}) + \dots + \text{Constant}$

Parameters	SE - Design	SE - Fabrication	SE - IAT	SE - LOCO	Total SE
Parameter 1	MISSION RISK CLASS	MISSION DESTINATION	LEAD ORGANIZATION	MISSION RISK CLASS	MISSION RISK CLASS
Parameter 2	LEAD ORGANIZATION	LEAD ORGANIZATION	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM TYPE	LEAD ORGANIZATION
Parameter 3	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM POWER	FLIGHT SYSTEM MASS	LEAD ORGANIZATION	FLIGHT SYSTEM ORGANIZATION
Parameter 4	PAYLOAD MASS	FLIGHT SYSTEM HERITAGE & TRL	PAYLOAD POWER	PAYLOAD ORG.	# OF PAYLOAD ELEMENTS
Parameter 5	-	PAYLOAD POWER	-	PAYLOAD MASS	-
Parameter 6	-	# OF PAYLOAD ELEMENTS	-	-	-
Parameter 7	-	# OF KEY PAYLOAD CONTRACTORS	-	-	-
Parameter 8	-	-	-	-	-
Factors	SE - Design	SE - Fabrication	SE - IAT	SE - LOCO	Total SE
α	-1.324	0.6258	1.2542	-1.8003	-0.6289
β	1.1828	1.1785	1.2551	1.3716	1.6227
γ	1.0758	0.6153	0.7568	2.7664	0.7725
δ	-0.2956	0.9679	-0.4485	-1.1355	0.3192
ϵ	-	-0.3596	-	-0.8098	-
ζ	-	0.476	-	-	-
η	-	-0.398	-	-	-
θ	-	-	-	-	-
Constant	5.4663	0.4832	0.3218	7.0029	3.1888

- Equations take the form of:
 - $LN(\$ / mo) = \alpha \cdot LN(\text{Parameter 1}) + \beta \cdot LN(\text{Parameter 2}) + + \text{Constant}$



PCA Support Function CERs – Level 3 (MA)



Engineering
Cost
Office

Parameters	MA - Design	MA - Fabrication	MA - IAT	MA - LOCO	Total MA
Parameter 1	MISSION RISK CLASS	LEAD ORGANIZATION	LEAD ORGANIZATION	MISSION RISK CLASS	LEAD ORGANIZATION
Parameter 2	MISSION DESTINATION	PARTS RATING	PAYLOAD ORG.	FLIGHT SYSTEM TYPE	PARTS RATING
Parameter 3	LEAD ORGANIZATION	-	FLIGHT SYSTEM POWER	LEAD ORGANIZATION	-
Parameter 4	PAYLOAD MASS	-	-	PAYLOAD ORG.	-
Parameter 5	IN-HOUSE SCOPE	-	-	PAYLOAD MASS	-
Parameter 6	-	-	-	-	-
Parameter 7	-	-	-	-	-
Parameter 8	-	-	-	-	-

Factors	MA - Design	MA - Fabrication	MA - IAT	MA - LOCO	Total MA
α	-0.9512	0.7966	1.7141	-1.8436	1.0714
β	0.3816	1.9518	-0.5915	1.5967	1.4507
γ	1.3512	-	0.4153	2.6268	-
δ	-0.2082	-	-	-1.4256	-
ϵ	0.6263	-	-	-0.9669	-
ζ	1.265	-	-	-	-
η	0.531	-	-	-	-
θ	-0.319	-	-	-	-
Constant	4.4335	1.6211	1.4491	7.5113	1.8207

- Equations take the form of:
 - $\text{LN}(\$/\text{mo}) = \alpha \cdot \text{LN}(\text{Parameter 1}) + \beta \cdot \text{LN}(\text{Parameter 2}) + \dots + \text{Constant}$





PCA Support Function CERs – Level 3 (I&T)



Parameters	IAT - Design	IAT - Fabrication	IAT - IAT	IAT - LOCO	Total IAT
Parameter 1	FLIGHT SYSTEM TYPE	MISSION RISK CLASS	FLIGHT SYSTEM TYPE	DIRECTED or AO	DIRECTED or AO
Parameter 2	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM ORGANIZATION	FLIGHT SYSTEM MASS	MISSION RISK CLASS	MISSION DESTINATION
Parameter 3	PARTS RATING	FLIGHT SYSTEM POWER	# OF PAYLOAD ELEMENTS	FLIGHT SYSTEM TYPE	FLIGHT SYSTEM ORGANIZATION
Parameter 4	-	PARTS RATING	IN-HOUSE SCOPE	# OF KEY PAYLOAD CONTRACTORS	FLIGHT SYSTEM POWER
Parameter 5	-	PAYLOAD MASS	-	-	PAYLOAD MASS
Parameter 6	-	PAYLOAD POWER	-	-	# OF PAYLOAD ELEMENTS
Parameter 7	-	# OF PAYLOAD ELEMENTS	-	-	# OF KEY PAYLOAD CONTRACTORS
Parameter 8	-	# OF KEY PAYLOAD CONTRACTORS	-	-	-

Factors	IAT - Design	IAT - Fabrication	IAT - IAT	IAT - LOCO	Total IAT
α	-1.1293	1.6112	0.9494	-2.0733	-0.4174
β	1.605	1.914	0.5801	-2.0785	0.5089
γ	1.8072	0.5159	0.4834	1.4943	1.0959
δ	-	1.4667	-1.047	1.1506	0.3182
ϵ	-	0.6233	-	-	0.2264
ζ	-	-0.4699	-	-	0.5159
η	-	0.9141	-	-	-0.2824
θ	-	-0.6018	-	-	-
Constant	1.1389	-4.3453	2.0337	5.2601	0.83

- Equations take the form of:
 - $\text{LN}(\$/\text{mo}) = \alpha \cdot \text{LN}(\text{Parameter 1}) + \beta \cdot \text{LN}(\text{Parameter 2}) + \dots + \text{Constant}$